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LIST OF ACRONYMS AND ABBREVIATIONS

ACQ	Acquisition
ADJ	Adjust
AOH	Apollo Operations Handbook
AOS	Acquisition of Signal
ATT	Attitude
AWY	Away
BEF	Blunt End Forward
B/U	Backup
CALIB	Calibration
CB	Circuit Breaker
CDH	Constant Delta Height
CDR	Commander
CM	Command Module
CMC	Command Module Computer
CMD	Command Module Commander's Position
CMP	Command Module Pilot
COAS	Crew Optical Alignment Sight .
CSI	Concentric Sequence Initiation
CSM	Command and Service Module
CT	Cease Tracking
DAP	Digital Autopilot
DH	Delta Height
DOI	Descent Orbit Insertion
DPS	Descent Propulsion System
DSKY	Display and Keyboard
DV	Delta Velocity
DWN	Down
EMS	Entry Monitor System
ET	Event Timer
FDAI	Flight Director Attitude Indicator
FPS	Feet Per Second

FWD	Forward
GDC	Gryo Display Coupler
GET	Ground Elapsed Time
GETI	Ground Elapsed Time of Ignition
GMBL	Gimbal
GND	Ground (Mission Control)
GPI	Gimbal Position Indicator
HA	Apogee Altitude
HGA	High-Gain Antenna
HOR	Horizon
HORIZ	Horizontal
HP	Perigee Altitude
IMU	Inertial Measurement Unit
INS	Insertion
IT	Initiate Tracking
LEB	Lower Equipment Bay
LGC	LM Guidance Computer
LLMK	Lunar Landmark
LM	Lunar Module
LMK	Landmark
LMP	Lunar Module Pilot
LOI	Lunar Orbit Injection
LOS	Loss of Signal
LV	Launch Vehicle
MNVR	Maneuver
MCC	Midcourse Correction
MCC-H	Mission Control Center - Houston
MDC	Main Display Console
MGA	Middle Gimbal Angle
MSFN	Manned Space Flight Network
MTCS	Move to Command Seat
MTLEB	Move to Lower Equipment Bay
MTVC	Manual Thrust Vector Control
NAV	Navigation

OPT	Optics
ORDEAL	Orbital Rate Drive Earth and Lunar
OSS	Optical Subsystem
PAD	Data Voiced to Crew From Ground
PB	Pushbutton
PC	Plane Change
PDI	Powered Descent Initiation
PGNCS	Primary Guidance, Navigation, and Control System
PHS	Phasing
PIPA	Pulse Integrating Pendulous Accelerometers
PLM	LM Pitch Angle
PRO	Proceed
PROG	Program
PROP	Propellant
R	Range
RCS	Reaction Control System
RDOT	Range Rate
REFSMMAT	Reference Stable Member Matrix
RHC	Rotation Hand Controller
RR	Rendezvous Radar
S	Shaft
SC	Spacecraft
SCS	Stabilization and Control System
SCT	Scanning Telescope
SECS	Sequence Events Control System
SEF	Small End Forward
SEP	Separation
S-IVB	Saturn S-IVB Stage
SM	Service Module
SPS	Service Propulsion System
S/U	Setup
SXT	Sextant
SYNC	Synchronize
TEI	Transearth Injection
TEMCC	Transearth Midcourse Correction

TFI	Time From Ignition
THC	Translation Hand Controller
THETA	Angle Between SC +X Axis and Local Horizontal
TIGN	Time of Ignition
TLI	Translunar Injection
TLM	Telemetry
TLMCC	Translunar Midcourse Correction
TPF	Terminal Phase Finalization
TPI	Terminal Phase Initiation
TRUN	Trunnion
TVC	Thrust Vector Control
VG	Velocity to be Gained
VHF	Very High Frequency
(XX:XX)	Indicates GET From Liftoff in Hours:Minutes
(XXX:XX:XX)	Indicates GET From Liftoff in Hours:Minutes:Seconds
(XXX,XXX/XXX,XXX)	(Ordeal/Inertial) Angles (Roll, Pitch, Yaw)
(XX,XX,XX)	Local Vertical DV's

Tracking Stations

ANG	Antigua Near Space Support Station
BDA	Bermuda Near Space Support Station
CRO	Carnarvon Near Space Support Station
CYI	Canary Near Space Support Station
GYM	Guaymas Near Space Support Station
HSK	Honeysuckle Deep Space Support Station
HTV	Huntsville Near Space Support Station
MAD	Madrid Deep Space Support Station
MER	Mercury Near Space Support Ship
MIL	MILA Near Space Support Station
RED	Redstone Near Space Support Ship
TEX	Corpus Christi Near Space Support Station
VAN	Vanguard Near Space Support Ship

1.0 PURPOSE

This document contains the nominal PGNCS crew procedures for the CSM-106 spacecraft which will be the target vehicle for the LM-4 active lunar orbit rendezvous. The procedures were developed in accordance with the Detailed Test Objectives P20.78 and P20.77 defined in Reference 10.1.

The purpose of the CSM Rendezvous Procedures document is to provide a single source of procedures information for use in flight planning, in crew training, and in preparing onboard data.

This is a control document, subject to review by all elements of the Apollo Program and to approval by the Procedures Configuration Control Board. Comments should be directed to Mr. Duane K. Mosel, Flight Procedures Branch, Flight Crew Support Division, Extension 5340 or to Mr. Stephen G. Paddock, Jr., Apollo Flight Crew Operations Group, Houston Operations, McDonnell Douglas Astronautics Company, Extension 6101.

2.0 INTRODUCTION

The CSM-106/LM-4 lunar orbit rendezvous exercise will begin during the twelfth revolution with undocking at 98:10:00 and end at approximately 105:52 with post rendezvous station keeping. The CSM procedures during this period are divided into eight segments of major activities which are discussed in detail in Section 3.0.

The nominal CSM-106/LM-4 mission profile is contained in Figure (2-1). This figure shows the locations in time and relative positions in space of the most significant nominal mission events. Trajectory data used to generate the mission profile and timeline for procedures development was obtained from References 10.2, 10.15, and 10.18. The rendezvous navigation update schedule assumed in the procedures was obtained from References 10.16 and 10.20. The schedule indicates tracking periods and assumes a one per minute SXT and VHF mark frequency. The minimum number of marks required during a tracking period has been defined as five at the beginning and five at the end of the period. However, the minimum number of marks is not recommended since simulation experience has shown that the planned number of marks is not always achieved due to systems monitoring requirements or target visibility problems. In addition, the general rules for SXT/VHF marking as defined in Reference 10.3 should be followed.

A history of the CSM body attitudes during the rendezvous is presented in Figure (2-2) through (2-5). Each figure illustrates the body attitudes with respect to the Moon, Sun, and Earth

and indicates FDAI roll, pitch, and yaw gimbal angles and the ORDEAL pitch angle for significant events during each lunar orbit. The orbital position of the CSM at each event is assumed and no attempt is made to show the LM orbital position other than an indication of it being above (below) and behind (ahead) the CSM.

Sections 4.0, 5.0, and 6.0 contain the onboard nominal rendezvous checklist; a summary timeline and summary checklist for the nominal mission, and the procedures ground rules, detailed nominal mission procedures, and CSM attitude summary, respectively. Section 7.0 contains a description of the five abort and rescue cases for which onboard rescue checklists have been developed. These cases are:

- 1) PDI Abort - LM Active/CSM Backup
- 2) Partial LM Phasing (DV greater than or Equal to 40 FPS) - CSM Active
- 3) Partial LM Phasing (DV less than 40 FPS) - CSM Active
- 4) Zero LM Insertion - CSM Active
- 5) Partial LM Insertion - CSM Active

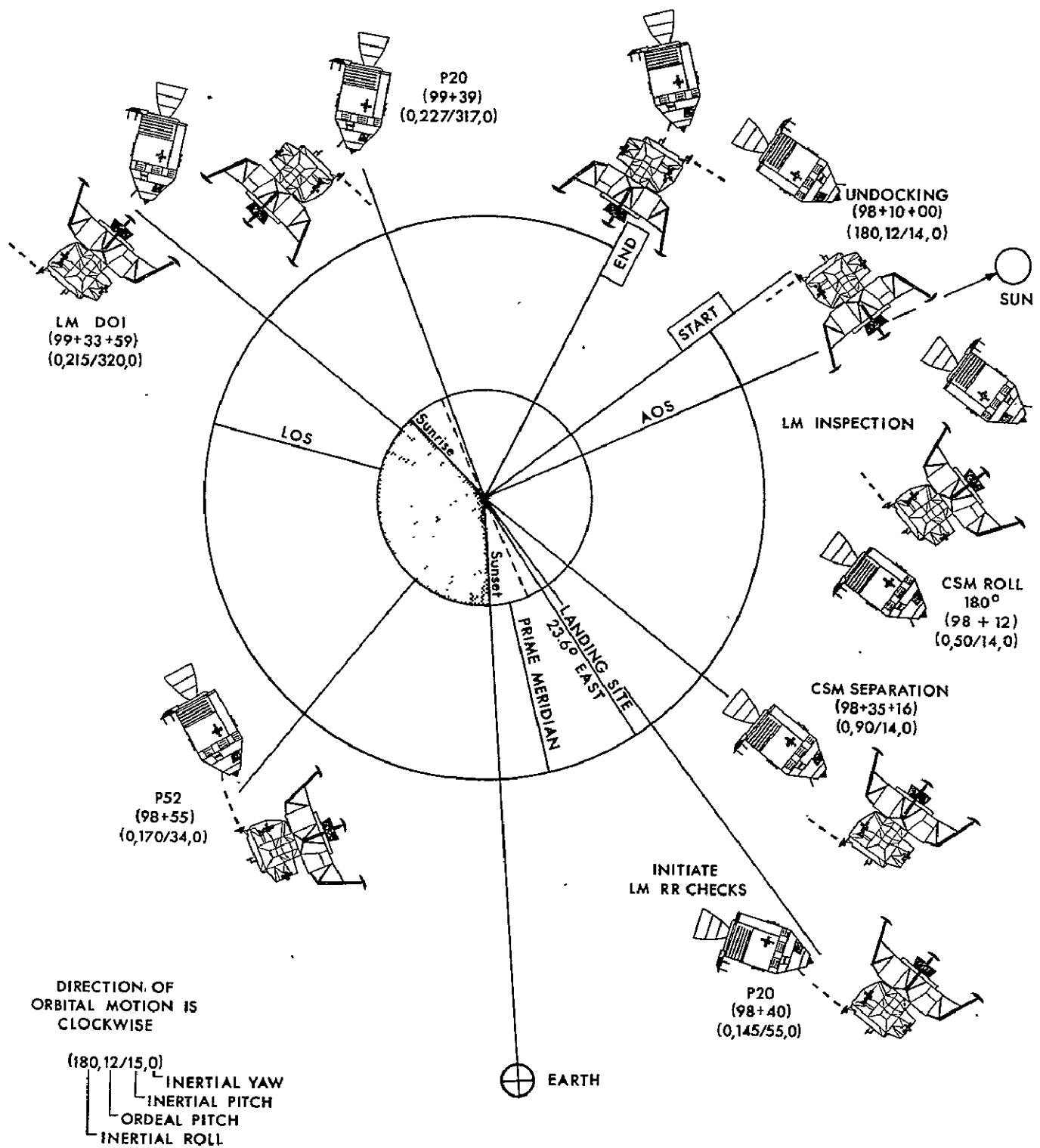
Section 8.0 contains the one-page onboard rescue checklist and a relative motion plot and pad page for each of the five LM rescue cases. A LM rescue procedures summary is contained in Section 9.0 including rescue procedures ground rules, summary timelines, and CSM attitude and navigation summaries for each rescue case.

All nominal and rescue procedures and onboard data were generated assuming a Mission F liftoff on 18 May 1969, at 16:49:00 GMI.

MISSION F

FIGURE 2-2

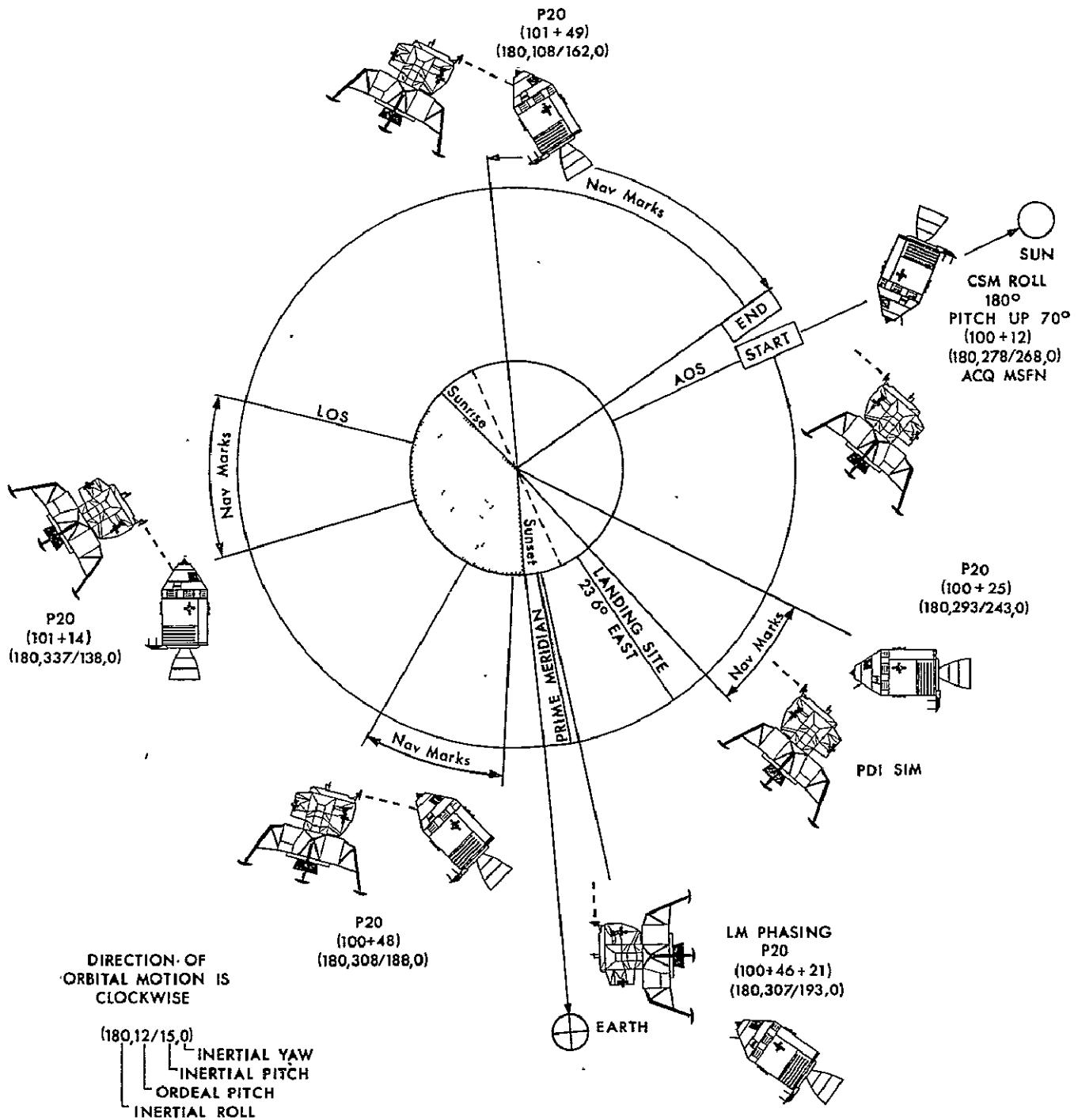
LUNAR ORBIT RENDEZVOUS
ATTITUDE TIME HISTORY
FOR THE CSM



MISSION F

FIGURE 2-3

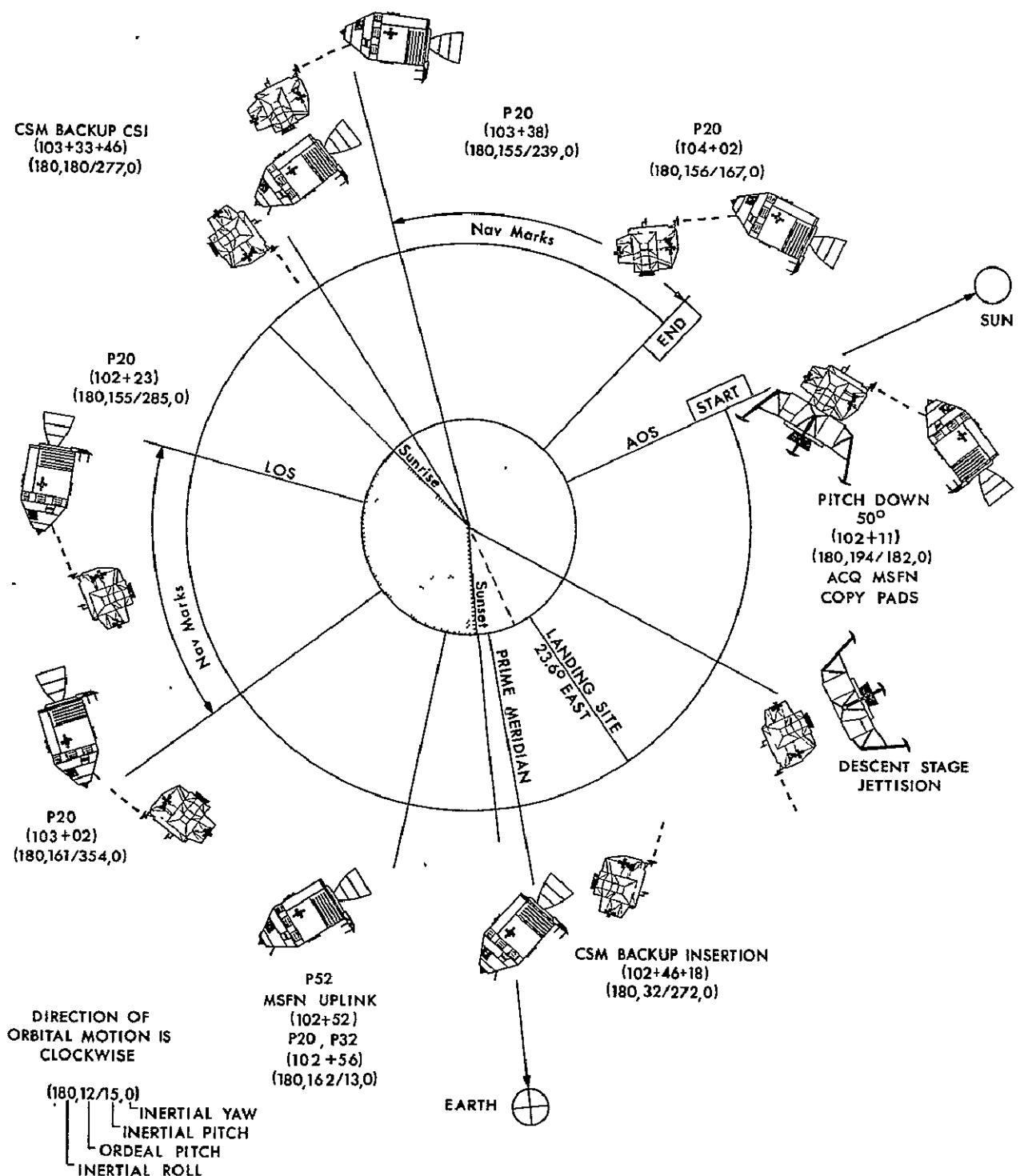
LUNAR ORBIT RENDEZVOUS
ATTITUDE TIME HISTORY
FOR THE CSM



MISSION F

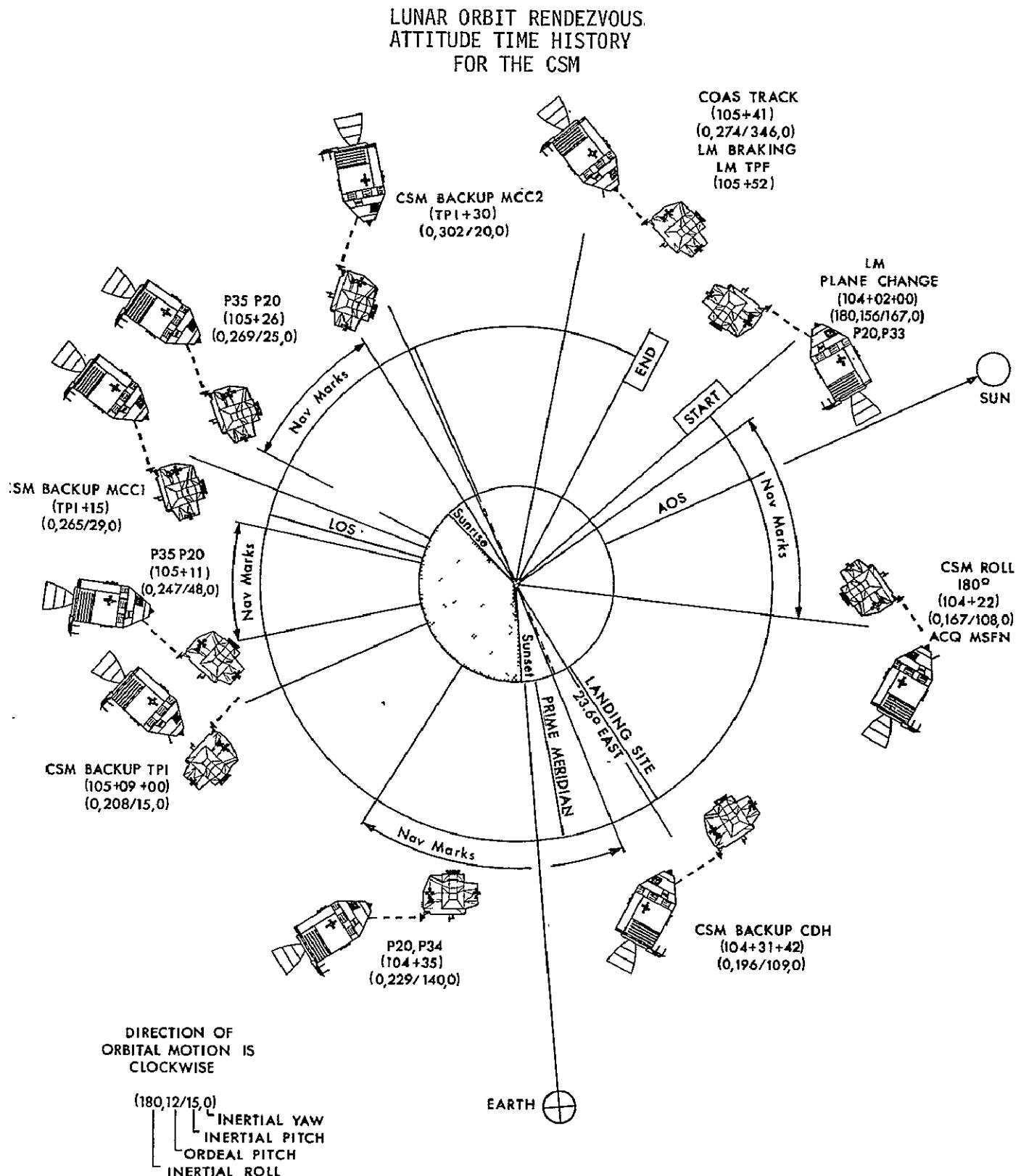
FIGURE 2-4

LUNAR ORBIT RENDEZVOUS
ATTITUDE TIME HISTORY
FOR THE CSM



MISSION F

FIGURE 2-5



3.0 DISCUSSION OF NOMINAL MAJOR EVENTS

3.1 Undocking and Separation

The preparation for undocking includes obtaining a MCC-H uplink of the CSM vector and copying the Separation maneuver pad at 97:05. Thirty minutes prior to undocking an automatic maneuver to the undocking attitude is performed using extended verb 49. This attitude is identical inertially to the Separation burn attitude except for the roll angle which is 180 degrees and a 14-degree yaw which is required during a pre-undocking LM AGS calibration. Prior to undocking, the CSM is yawed to zero degrees, a RR transponder check is made, the GDC is aligned to the IMU, the FDAO ORDEAL is verified, and the DAP is reloaded to reflect a CSM only configuration. At 98:10 undocking is performed after which the CSM will station keep in SCS control at about 40 feet. The CMP will then inspect and photograph the LM landing gear and descent engine bell as the LM does a 360-degree yaw maneuver. Prior to the inspection the CSM will manually roll 180 degrees to allow acquisition of the MSFN with the HGA for color TV. At 98:20 LM maneuver P76 pads will be copied for the DOI, Phasing, and PDI abort burns. Prior to the CSM-active separation burn at 98:35:16, the CMP will load the External DV Program, P30, with a minus 2.5 FPS VG_z (local vertical). The RCS Thrust Program, P41, will be called and an automatic maneuver will be made to the burn attitude which is shown in Figure (2-2). (This should be a very small maneuver since the undocking attitude was the inertial separation burn attitude, except for roll.) The burn will be accomplished

by thrusting aft in the CSM minus X direction (i.e. radially down) and increasing the DSKY VG_x (body) from a plus 2.5 FPS to 5 FPS. The HGA can be utilized for communication during the separation burn.

3.2 Descent Orbit Insertion

Following the CSM Separation burn, the Rendezvous Navigation Program, P20, will be called and the CSM maneuvered automatically 40 degrees to the preferred track axis for LM radar, VHF range, and optics checks. An IMU realign to REFSMMAT will be performed after sunset at 98:55 using the IMU Realign Program, P52. At approximately 99:05 CSM and LM state vectors will be uplinked by MCC-H. Since the CSM does not backup the DOI burn, its only function will be to monitor the LM burn and hold an attitude which provides radar transponder coverage. This will be done in Program P20 with SXT tracking along the preferred track axis. The Target DV Program, P76, will be loaded with the LM DOI burn parameters and these data incorporated when the burn is confirmed by the LM. LM DOI occurs at 99:33:59.

The attitude of the CSM from the CSM Separation burn through LOS prior to the LM DOI burn is favorable for HGA communications.

3.3 Phasing

Following the DOI burn, Program P20 will be called and the CSM attitude will be trimmed to the preferred track axis if required. The CMP will confirm that the rendezvous navigation program is tracking the LM correctly. At 100:07 the CSM will be manually rolled 180 degrees and pitched up 70 degrees to allow acquisition

with the HGA. At 100:25 Program P20 will be called and the CSM will be maneuvered 25 degrees to the preferred track axis. A period of SXT/VHF marking will occur from 100:27 to 100:37. The CSM will not back up the LM Phasing burn so it will be tracking with the preferred track axis during the burn. LM Phasing occurs at 100:46:21. After confirmation of the nominal LM Phasing burn, the CMP will incorporate the LM Phasing burn parameters into Program P76. In the event the LM cannot perform the Phasing burn or performs a partial burn, the CSM will do nothing immediately but will setup to do the prescribed rescue burn at approximately 101:33. The CSM has HGA coverage continuously from AOS through the Phasing burn.

3.4 Insertion

Following the LM Phasing burn, Program P20 will be called and tracking with the preferred track axis will again be established. A period of SXT/VHF marking occurs from 100:51 to 101:24. This period is interrupted for 13 minutes during which time the LM does a P52 (tracker light not visible). Only VHF marks are taken during this period. At approximately 101:10 maneuver pads for the LM Insertion burn and the CSM Backup Insertion burn are transmitted and copied. Another period of SXT/VHF navigation updates occurs from 101:49 to 102:09. These update periods include VHF marks only if the range is less than the 327 N.M. VHF range measurement limit. At 102:11 AOS occurs and the CSM will be pitched down 50 degrees and the MSFN acquired with the HGA. The MCC-H will uplink the CSM state vector and the CSM Backup Insertion burn pad will be updated at this time if required. The CSM Backup Insertion burn parameters, targeted three minutes after the LM ignition time, will be loaded into Program P30 at 102:30 and Program P40 will be called at 102:32. The maneuver to the burn

attitude will require a pitch down of 110 degrees. Following the attitude maneuver, the SPS will be configured for the backup burn and the GDC aligned to the IMU. The LM Insertion occurs at 102:43:18. After confirmation is received that the LM has completed the burn, Program P76 will be called and the LM burn data incorporated into the CMC LM state vector. The CSM attitudes as shown in Figure (2-3) and (2-4) are compatible with HGA usage during the phase.

3.5 Concentric Sequence Initiation

Immediately following the Insertion burn, a P52 IMU realign to REFSMMAT will be performed and the MCC-H will uplink the LM state vector to the CMC. The LM vector will be that computed by the LM after Insertion and relayed to the ground.

Program P20 will then be called and an automatic pitch maneuver of 101 degrees will be made to the preferred track attitude. At 102:59 the CSI Targeting Program, P32, is called and the CSM Backup CSI targeting parameters are loaded. The CSM CSI burn will be targeted for 103:33:46 which is identical to the LM TIGN. A period of SXT/VHF marking will then take place from 103:02 to 103:07 followed by a period of VHF only marks to 103:25. Concurrently with the VHF only tracking, the CMP will obtain the LM and CSM out-of-plane velocities from extended verb 90 for inclusion in the LM and CSM CSI targeting programs.

At approximately 103:28 the LM will voice over its CSI solution for P76. Program P40 will then be called and an automatic pitch maneuver of 11 degrees will be made to the Backup CSI burn attitude. The LM CSI burn occurs at 103:33:46. After verification of the nominal LM CSI burn, the CMP will incorporate the LM burn parameters in Program P76.

The CSM attitude, as specified in Figure (2-4) is compatible with HGA coverage from Insertion until the beginning of the track period at about 103:02 at which time the attitude may be unfavorable. The tracking attitude later in the period does afford HGA coverage until LOS at 103:23.

3.6 Constant Delta Height and Plane Change

At 103:38 approximately four minutes after the LM CSI burn, Program P20 will be called and an automatic maneuver of 38 degrees will be made to the preferred track axis. A period of SXT/VHF markings occurs from 103:41 to 104:02. At 103:44 after three SXT/VHF marks, the WR matrix (2000,2) will be loaded. Extended verb 90 will be called at 103:55 and the targeting parameters for the LM Plane Change maneuver will be computed and voiced to the LM. The CSM onboard state vectors are used instead of the LM state vectors, because the CSM knowledge of the out-of-plane positions with SXT tracking is more accurate than the LM knowledge of out-of-plane positions with radar tracking. Following the LM Plane Change burn at 104:02, the target DV parameters are incorporated into the LM state vector. At 104:05 the CDH targeting Program P33 will be called and the CSM Backup CDH burn will be targeted for 104:31:42 which is the LM TIGN. Since the CSM nominally does not backup the LM Plane Change burn, a trim maneuver to the preferred track axis should not be required when Program P33 is called. SXT and VHF marks will be taken from 104:08 to 104:20, at which time tracking is terminated and the CSM will voice to the LM an extended verb 90 out-of-plane solution. At 104:22 the CSM will be manually maneuvered 180 degrees in roll so that the MSFN can be acquired with the HGA. Approximately five minutes before the LM CDH burn the CMP copies the LM CDH pad for later loading of Program P76. The RCS Thrust Program, P41, is called at 104:29 and the burn

attitude maneuver is bypassed if the nominally zero burn is small. Following the LM CDH burn at 104:31:42, Program P76 is called and the LM CDH burn incorporated into the CMC LM state vector.

The CSM attitudes are favorable for HGA communications from approximately 104:22 through the CDH burn as shown in Figure (2-5).

3.7 Terminal Phase Initiation

After completion of the LM CDH burn, Program P20 will be called which will request an automatic maneuver of 49 degrees to the preferred tracking attitude. Program P34 will be called and the CSM Backup TPI burn Data will be loaded with the elevation angle option. SXT and VHF marks are scheduled for a period of 18 minutes starting at 104:38. It is probable that sun interference in the SXT will limit the total number of SXT marks to 11 taken in darkness.

After moving to the command seat, the CMP will verify the ORDEAL FDAI. He will then recall P34 and, using the "TIGN" option with the LM computed TPI TIGN, compute the CSM TPI Backup burn parameters. Program P40 will be called and an automatic maneuver of 53 degrees will be made to the TPI burn attitude. The TPI burn will nominally be performed at 105:09:00 with a CSM to LM elevation angle of 208.3 degrees. After the LM has completed the burn the CMP will incorporate the LM target DV in Program P76. As seen in Figure (2-5), the CSM has HGA coverage from acquisition right after AOS through the TPI burn.

3.8 TPI To Station Keeping

After the TPI burn, Program P20 will be called and the CSM will be automatically maneuvered 33 degrees to the preferred track attitude. The CMP will move to the LEB during the maneuver,

call the MCC Targeting Program, P35, and take SXT and VHF marks for eight minutes starting at 105:13. After obtaining the MCC1 solution in Program P35 he will compare it with the LM solution and call Program P41 in preparation for the backup burn. The CSM will not maneuver from the preferred tracking attitude. The MCC1 burn will nominally be performed by the LM at 105:24:00 (TPI plus 15 minutes). After MCC1 the CMP will incorporate the LM MCC1 Target DV in Program P76.

Following MCC1, Program P35 will be called and an automatic trim to the preferred tracking attitude will be made if required. SXT and VHF marks will be taken for a period of 9 minutes terminating at 105:35. The MCC2 solution will be compared with the LM TPI solution, after which Program P41 will be called. The CSM will remain at the preferred tracking attitude while the LM performs MCC2 at 105:39:00 (TPI plus 30 minutes). After MCC2 the CMP will incorporate the LM Target DV in Program P76.

The CSM will then be maneuvered automatically 34 degrees to the COAS tracking attitude using extended verb 89 while the CMP moves back to the command seat. The Thrust Monitor Program, P47, will be called at a range of 1.25 nautical miles and VHF ranging data and V83 will be used to monitor the LM line-of-sight control and braking. Should the LM experience difficulty, the CSM will perform the line-of-sight control and braking. The braking gates are specified in the checklist of Section 4.0. TPF nominally occurs at 105:52:00.

4.0 NOMINAL ONBOARD RENDEZVOUS CHECKLIST

The nominal CSM onboard rendezvous checklist is presented in this Section. The rendezvous checklist was formulated to be compatible with crew operations in simulators and in flight. Therefore, the checklist reflects procedures in an extremely abbreviated form. The narrative presented in Section 3.0 provides a word description of the checklist events. The rendezvous checklist procedures have in part been verified on a man-in-the-loop simulator. Additional simulations will occur before launch and the rendezvous checklist updated accordingly to produce a verified checklist.

The nominal onboard rendezvous checklist includes procedures for performing all CSM PGNCS activities required during the LM active rendezvous. All activities required for the operation and/or monitoring of systems other than the PGNCS will be included in the rendezvous checklist by the appropriate systems personnel.

EXTERNAL DV PADS

CSM SEP PAD

33	00	:	000	:	0	.
81	+	0000.0	+	0000.0	-	0002.5
22	XXX		XXX		XXX	

CSM BACKUP
INSERTION PAD
INITIAL

47	+	.	+	00000.		
48		
33	00	:	000	:	0	.
81
22	XXX		XXX		XXX	
ΔV_C	X	.				
11	00	:	000	:	0	.
37	00	:	000	:	0	.
N						

CSM BACKUP
INSERTION PAD
UPDATE

47	+	.	+	00000.		
48		
33	00	:	000	:	0	.
81
22	XXX		XXX		XXX	
ΔV_C	X	.				
11	00	:	000	:	0	.
37	00	:	000	:	0	.
N						

NOMINAL LM IGNITION TIMES

CSI 11	00	:	000	:	0	.
PC 33	00	:	000	:	0	.
TPI 37	00	:	000	:	0	.

CSM RENDEZVOUS
RESCUE PADS

CSI ONE	00	:	000	:	0	.
81
N						

RESCUE TWO PAD

47	+	.	+	00000.		
48		
33	00	:	000	:	0	.
81
22	XXX		XXX		XXX	
ΔV_C	X	.				
11	00	:	000	:	0	.
37	00	:	000	:	0	.
N						

CSI TWO

11	00	:	000	:	0	.
81
N	.					

CDH

13	00	:	000	:	0	.
81

CANNED RESCUE TWO PADS FOR:

1. LM PDI ABORT
2. PARTIAL PHASING (0 - 40)
3. PARTIAL PHASING (40 - NOM)
4. ZERO INSERTION
5. PARTIAL INSERTION

ARE INCLUDED ON RESCUE CHECKLISTS

TPI

37	00	:	000	:	0	.
81
59
LOS BT	XX	:	XX	:	XX	:

97:00 CONTINGENCY EVA PREP

POO; MCC-H UPLINK RLS
AND CSM VECTOR

V66
COPY: CSM SEP PADS
DONN HELMET AND GLOVES
CSM/LM COMM CHECK
VHF AM A - SIMPLEX
VHF RCV ONLY -,B DATA

97:10 MIN DB FOR LM RCS COLD FIRE CHECKS

M
S
F
N MAX DB FOR LM RCS HOT FIRE CHECKS

97:20 VERIFY AUTO RCS SEL-C4(-PITCH-X)-OFF
VERIFY AUTO RCS SEL-B3(+YAW-X)-OFF FOR LM
RR SELF TEST
VERIFY RNDZ XPNDR-OFF

L
O
S

97:30

18⁴ 97:30 CONFIGURE CAMERA:
CM/SEQ/18/CEX-BRKT (RH WIN)
MIR (f11,250,00) 6FPS,15 MIN

DAP (21112) (+37768) (-00052)
V46 (11111) (+30847) (+00059)

GND 47 + . + .
UPDATE 48

V62
V49; AUTO TO UNDOCKING ATTITUDE (180, 282/14,14)
(INERTIAL SEP ATT EXCEPT ROLL AND 14° YAW)

UNDOCK GET 98: 10: 00
LOAD 22 180 14 14

SC CONT-CMC
CMC MODE-FREE (AS REQ FOR AGS CALIB)
CMC MODE-AUTO (AFTER 32 SEC)
VERIFY MAX DB FOR AGS CALIB
SYSTEMS CHECKS AND SWITCH VER

97:50 YAW 14° LEFT AFTER LM AGS
CALIB (180,312/14,0)

RR XPNDR CHECKS:
RNDZ XPNDR ACTIVATION & SELF TEST
cb RNDZ XPNDR FLT BUS - close (verify)
RNDZ XPNDR - HTR for 24 min
(1 min if self test only)
RNDZ XPNDR - PWR
SYS TEST (1h) - XPNDR
SYS TEST (rh) - A (RRT XMTR OUT PWR)
SYS TEST ind ->1 vdc
SYS TEST (rh) - B (RRT AGC SIG)
RNDZ XPNDR - TEST (hold)
SYS TEST ind ->1 vdc

98:00

98:00 RNDZ XPNDR - OPERATE
 SYS TEST ind - 0 - 4.5 vdc
 SYS TEST (rh) - C (RRT FREQ LOCK)
 SYS TEST ind - <8 vdc unlocked, >4 vdc
 SYS TEST (rh) - B locked

(R=R=0)(IF NOT; V66)
 GDC ALIGN
 ORDEAL (V83)

SC CONT-SCS; DAP (11102) ; V46

240° ROLL OPTIMUM FOR SHADE AND MSFN!

98:10 UNDOCKING (98:10:00) (180, 12/14, 0)

-25 AUTO RCS SEL - B3; C4-MNA
 DV CG-CSM
 RR XPNDR-PWR(VERIFY)
 ROLL LEFT 180°(2°/SEC) (0,50/14,0)
 V64; ACQ MSFN
 PHOTOS; COLOR TV
 INSPECT LM

A
 O
 S

(LM 360° YAW)

UNDOCK INSPECT

98:20 COPY PADS

-15 -71.1 0 -.3

84	.	.	.	DOI P76
33	;	;	;	(20)

-12 170.4 0 -95.6

84		..	.	PHASING
33	;	;	;	P76

-45 96.2 0 0

84	.	.	.	PDI ABORT
33	;	;	;	P76

351

98:30 DAP (11102) ; V46; SC CONT-CMC

19

98:30

-5

-3

0

98:40

M
S
F
N

98:50

99:00

P30 (LOAD VGZ=-2.5)
* * * * *

P41 (TRIM)

RCS

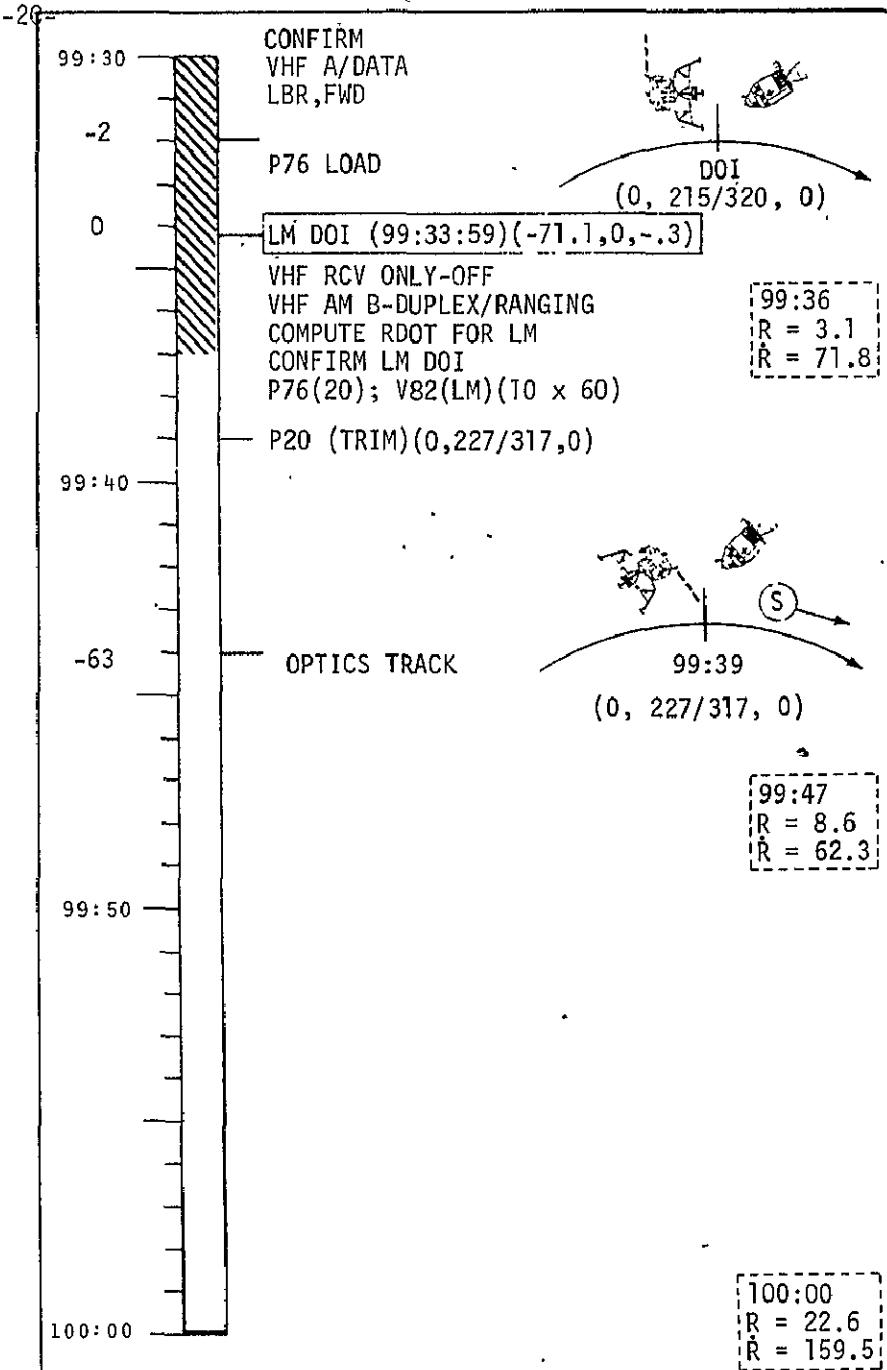
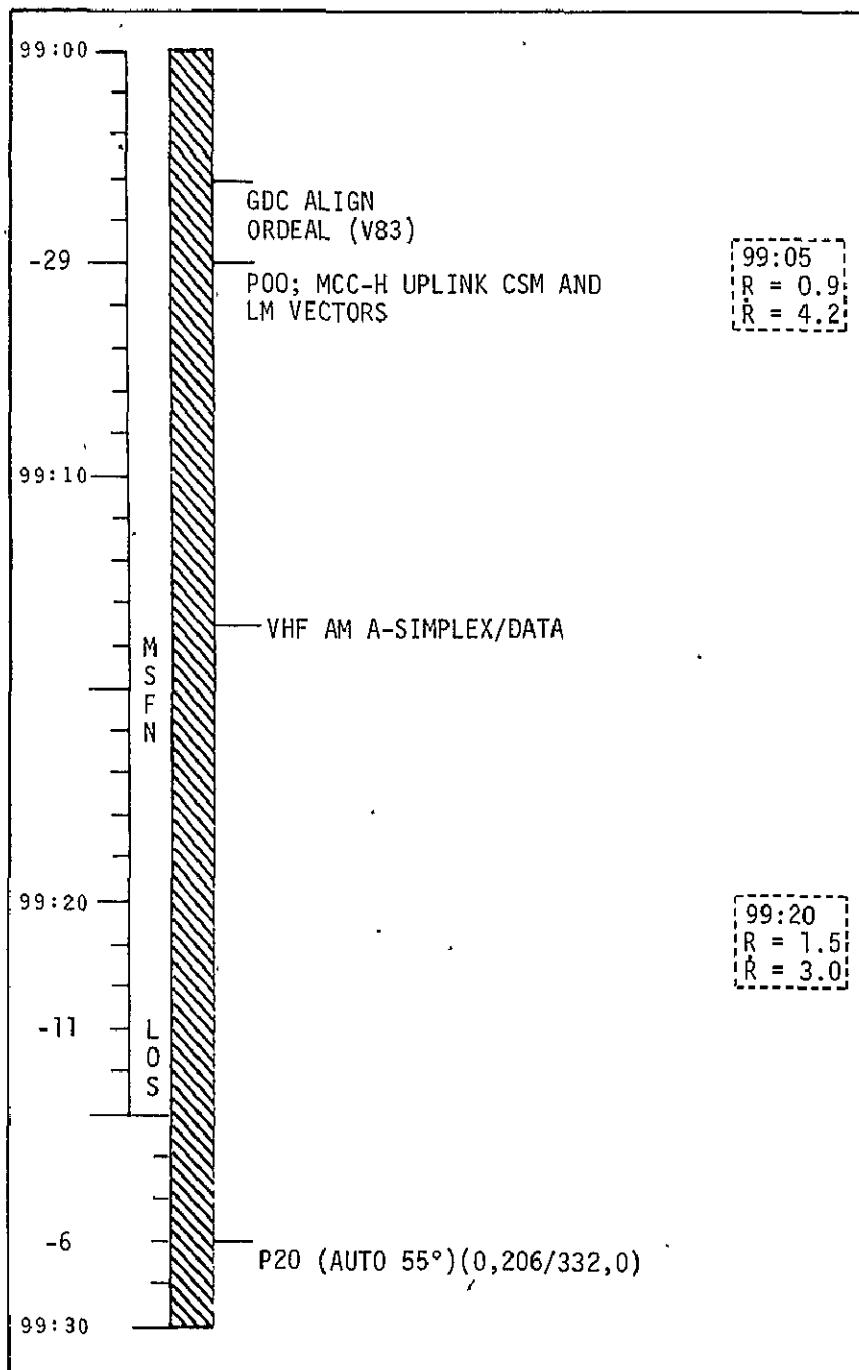
CSM SEP (98:35:16) (0, 0, -2.5)
(THRUST AFT) (0, 90/14, 0)
(BURN VGX 2.5 → 5.0)

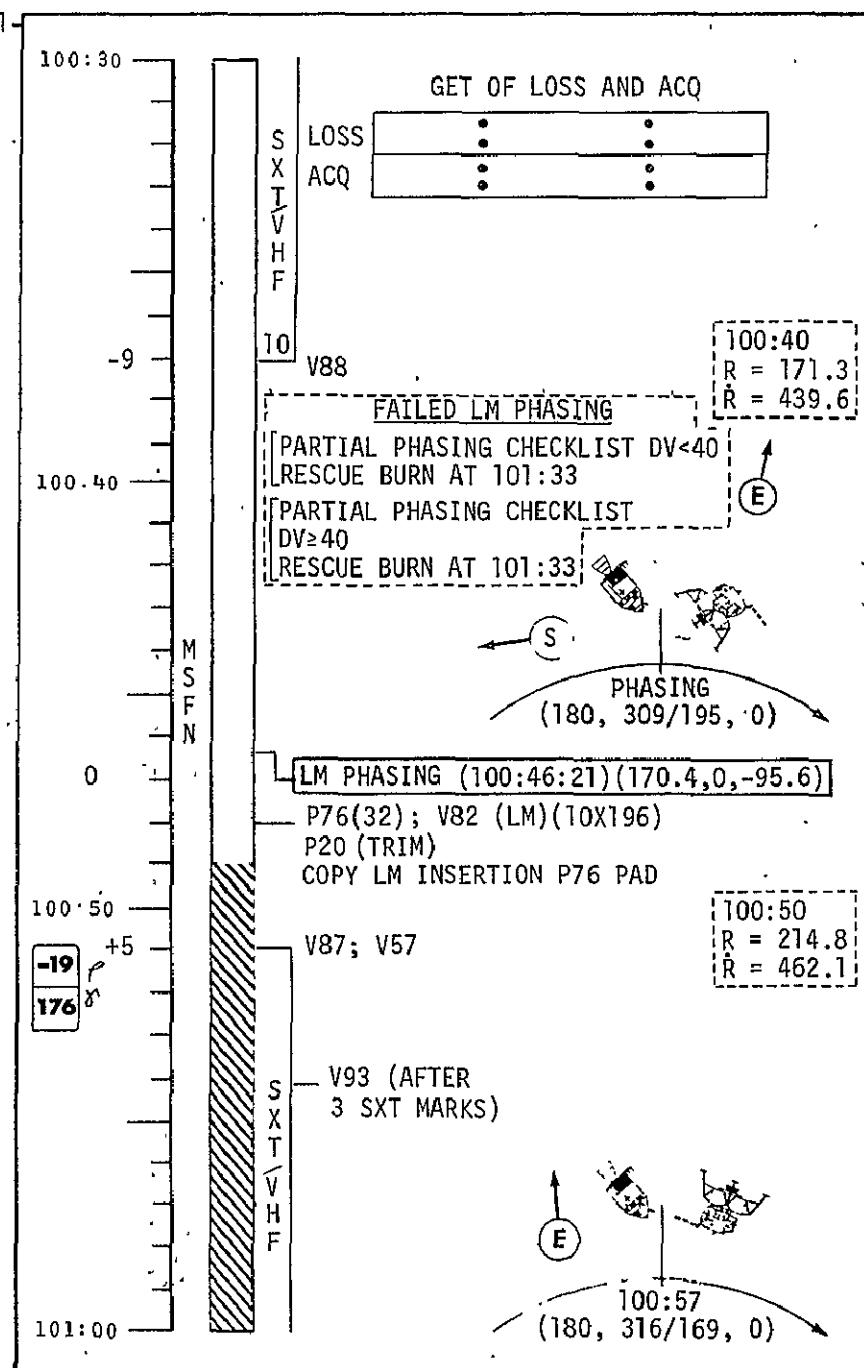
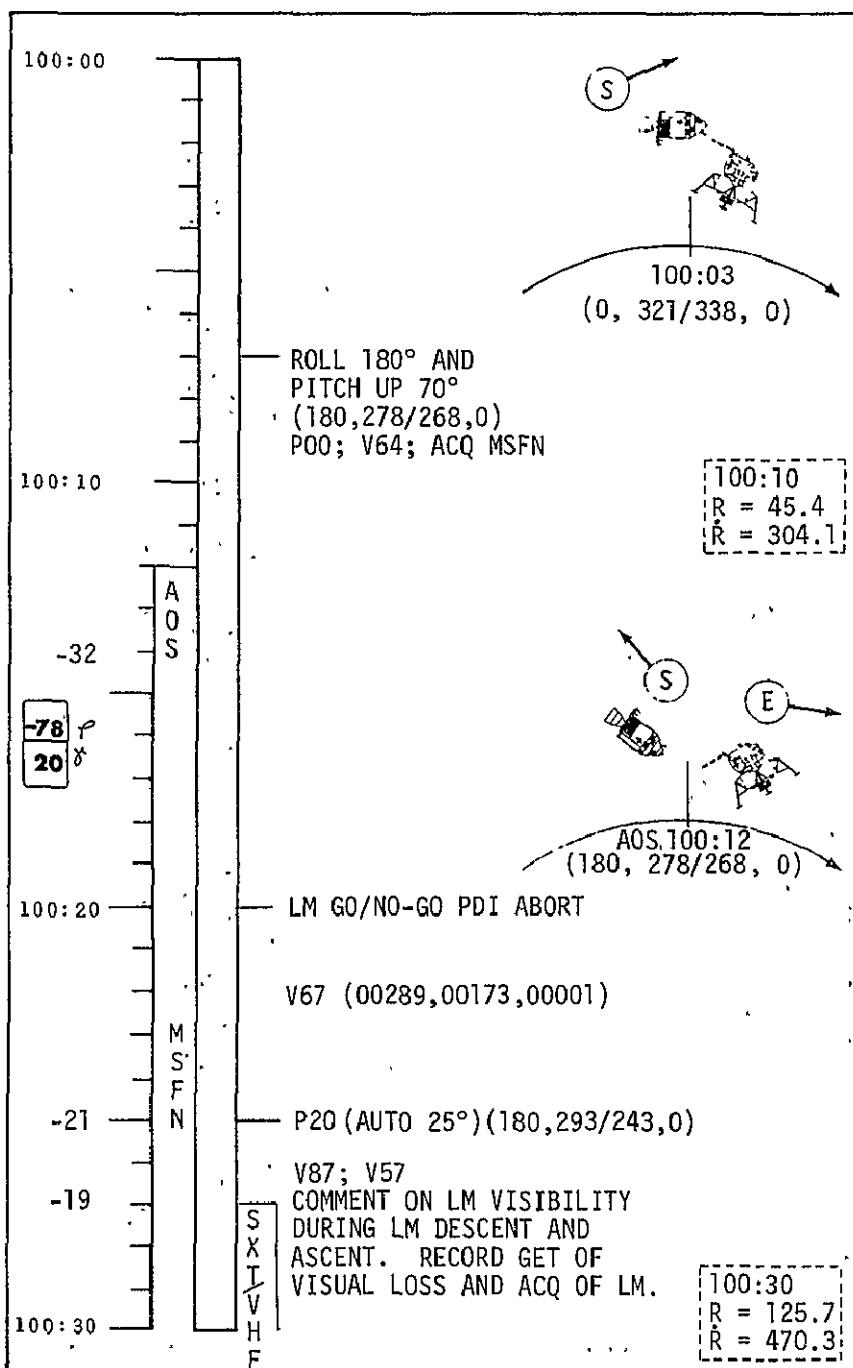
P20 (AUTO 40°) (0, 145/55, 0)
VHF ANT-RT
VHF AM B - DUPLEX/RANGING
LM RR CHECKS
EMS VHF AND V83 RANGE
OPTICS CHECKS

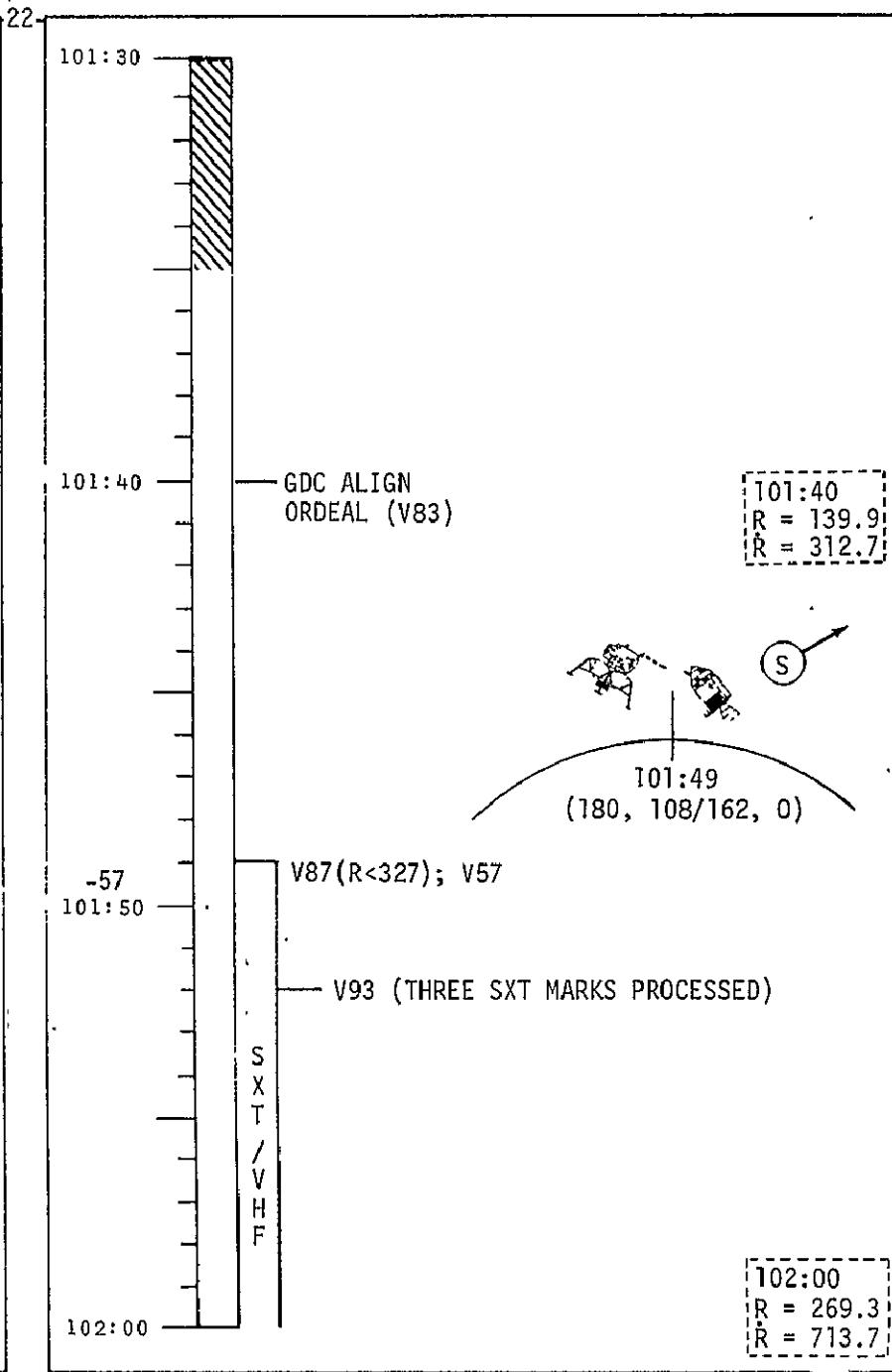
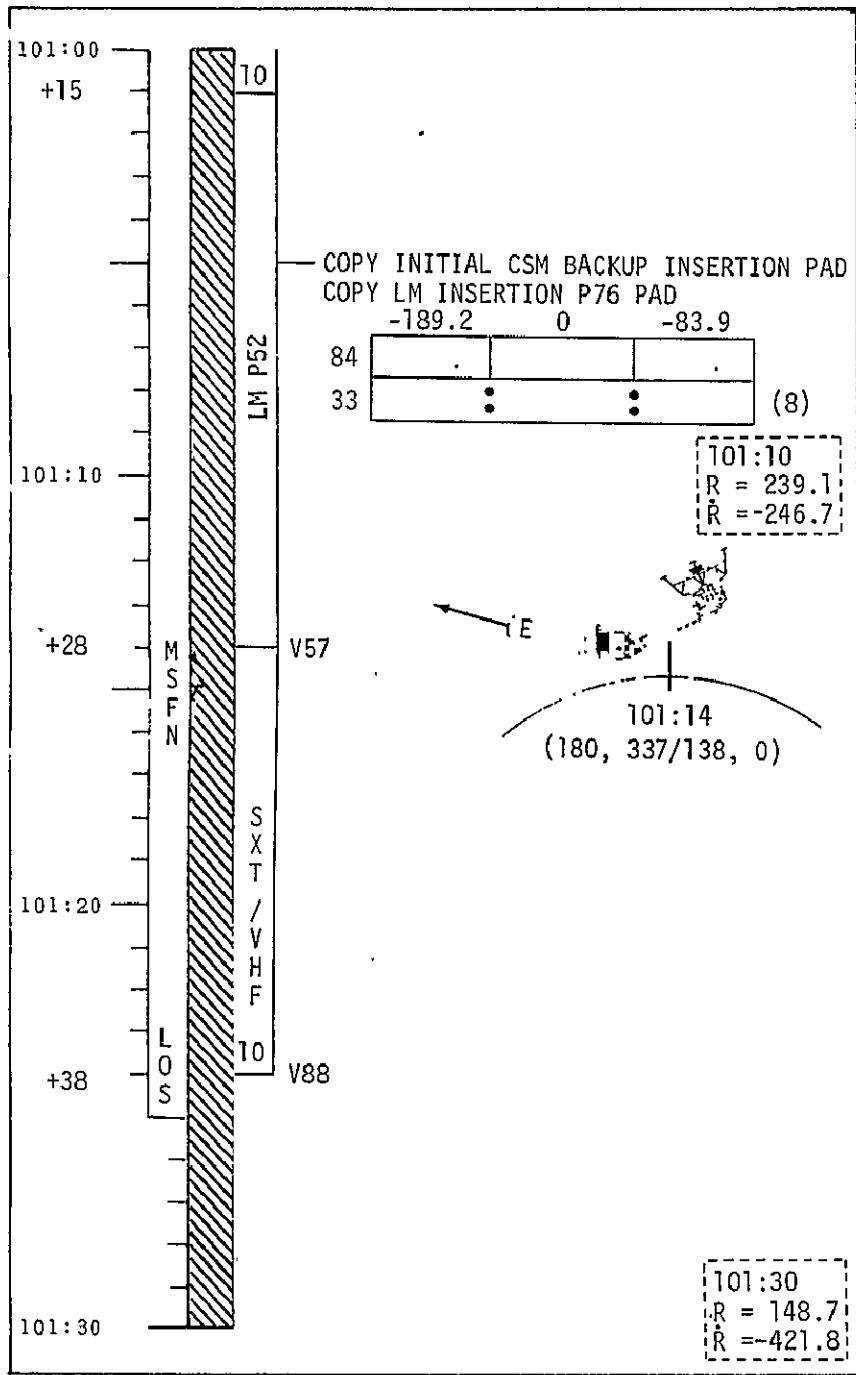
98:45
R = 0.2
R = 2.8

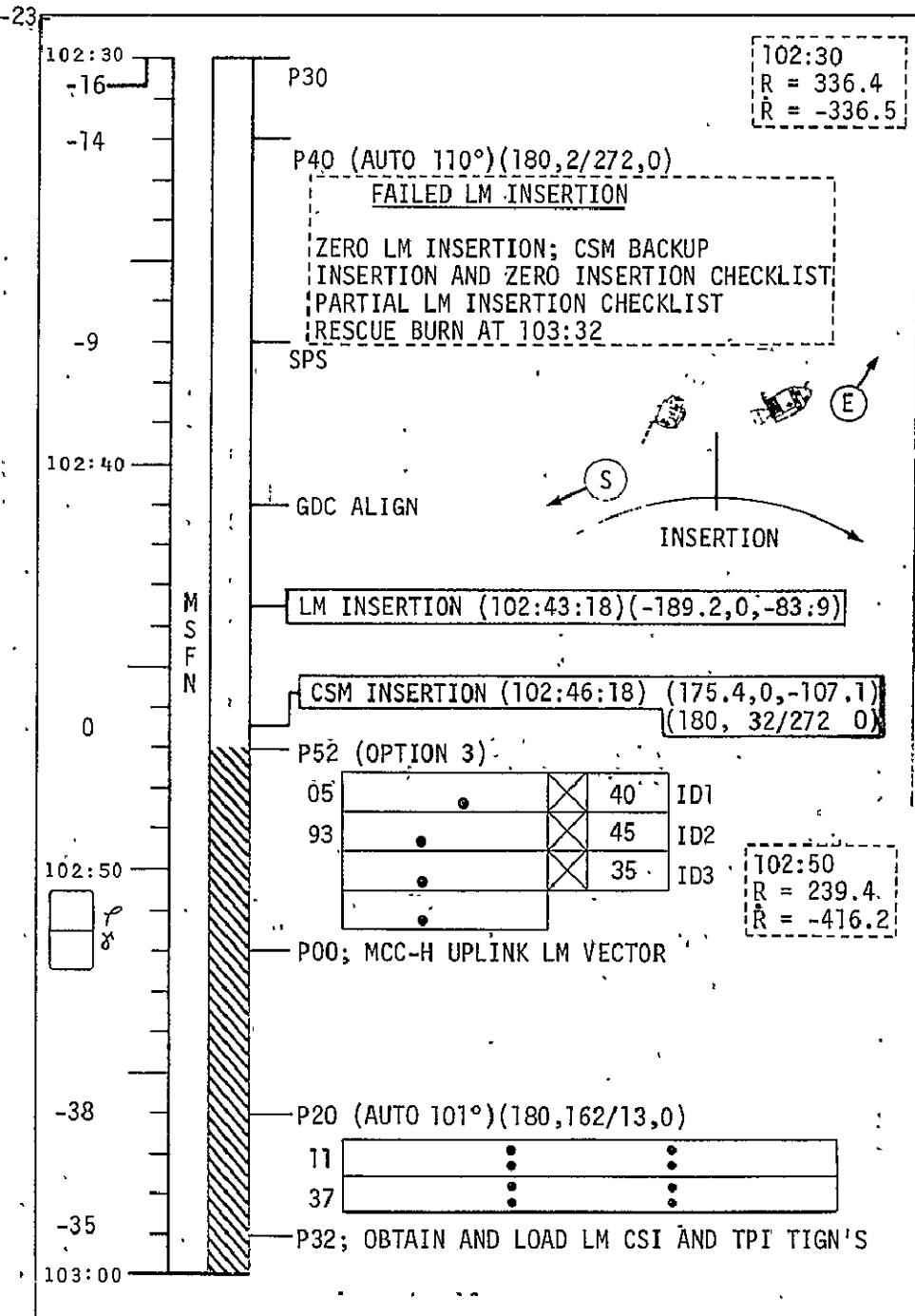
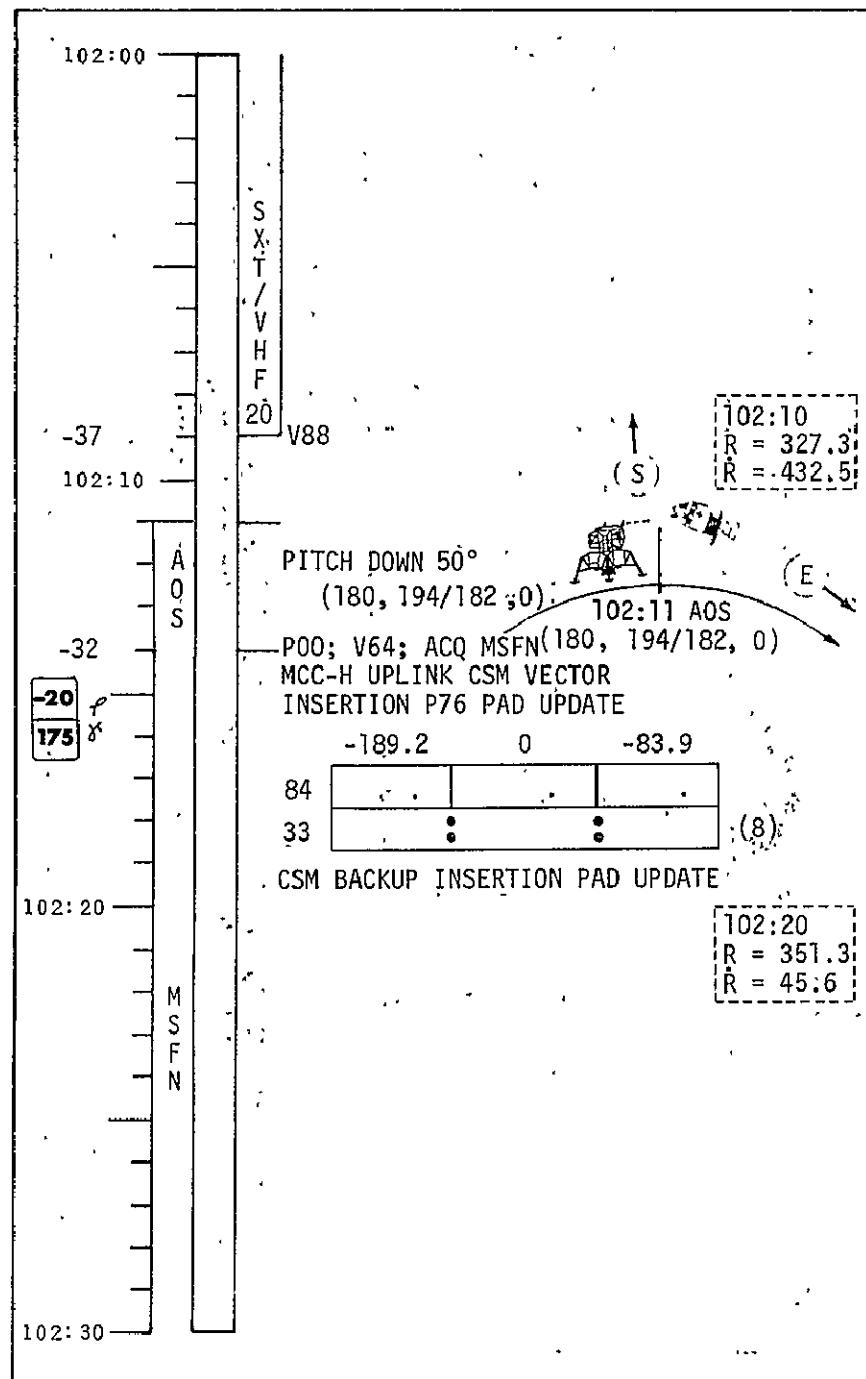
P52 (OPTION 3)

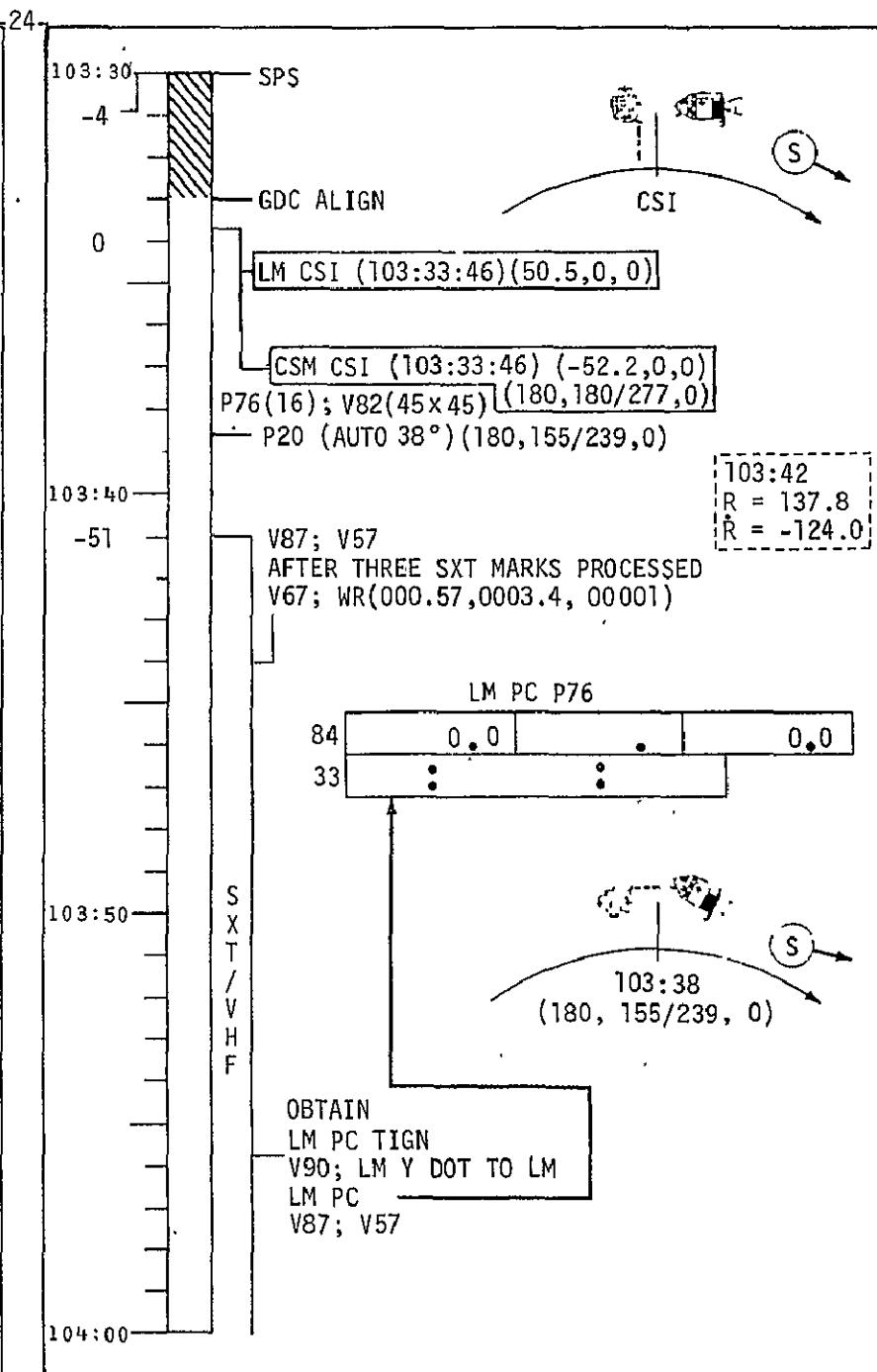
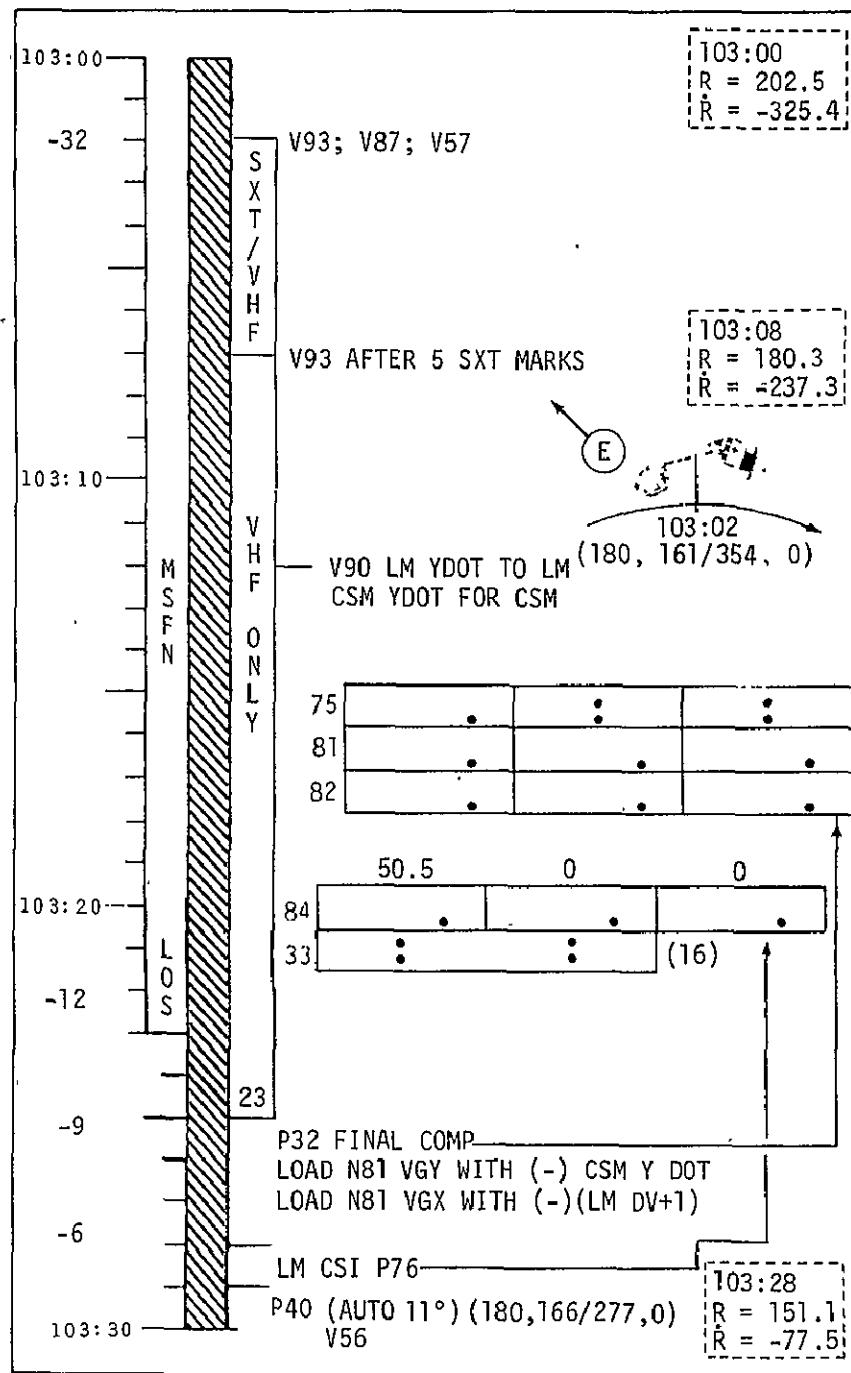
05	.	40	ID1
93	.	42	ID2
	.	35	ID3
	.		

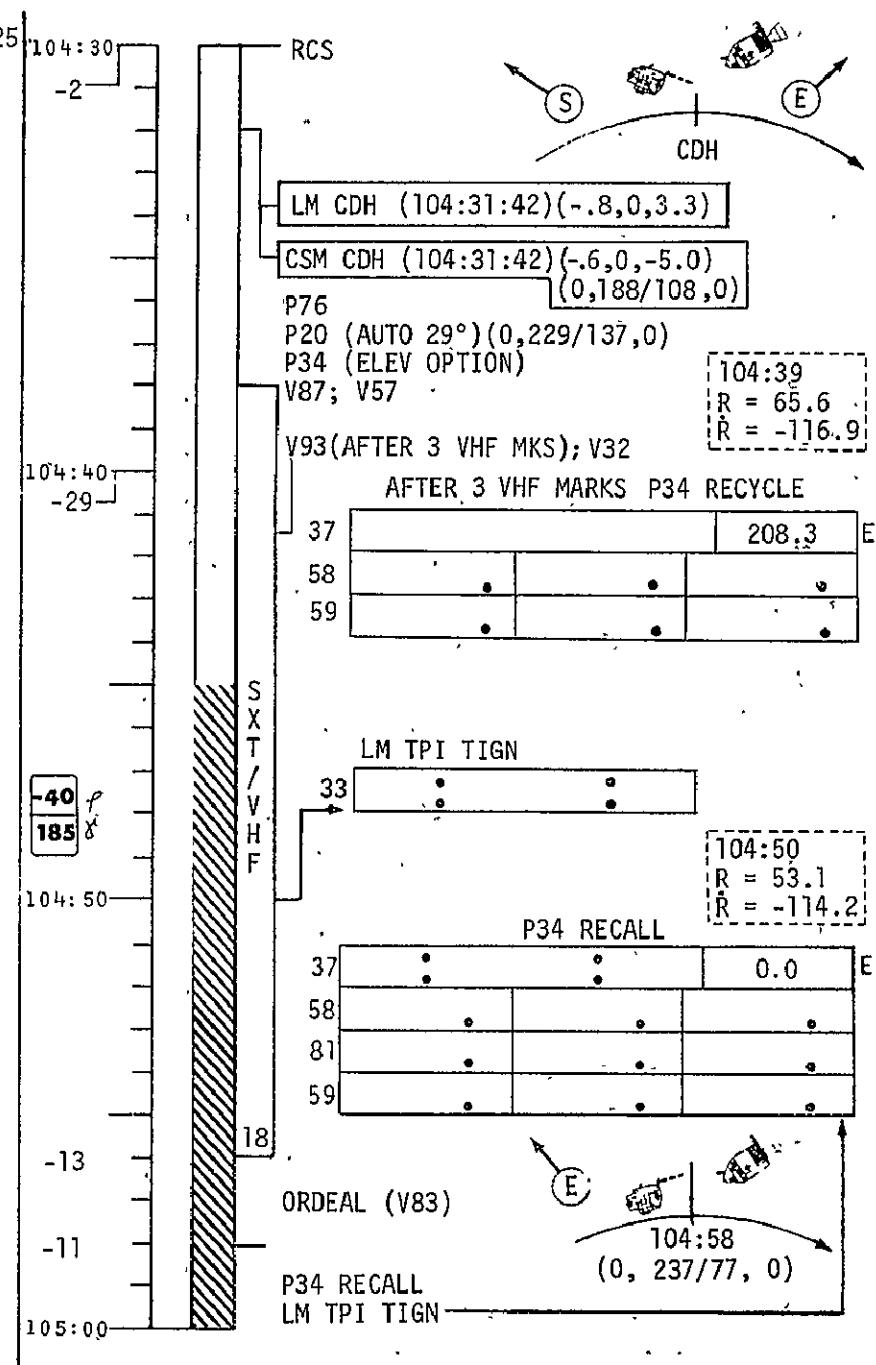
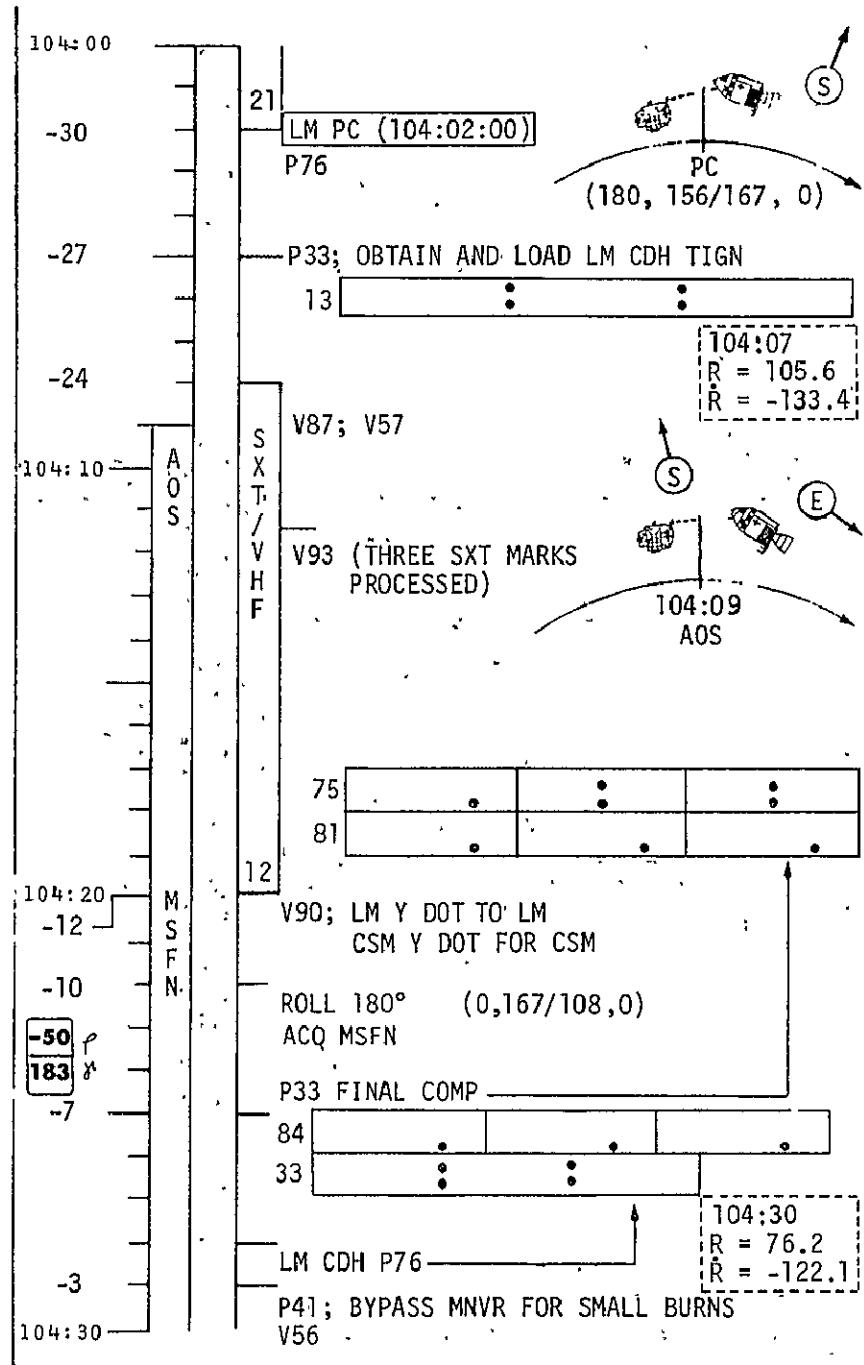


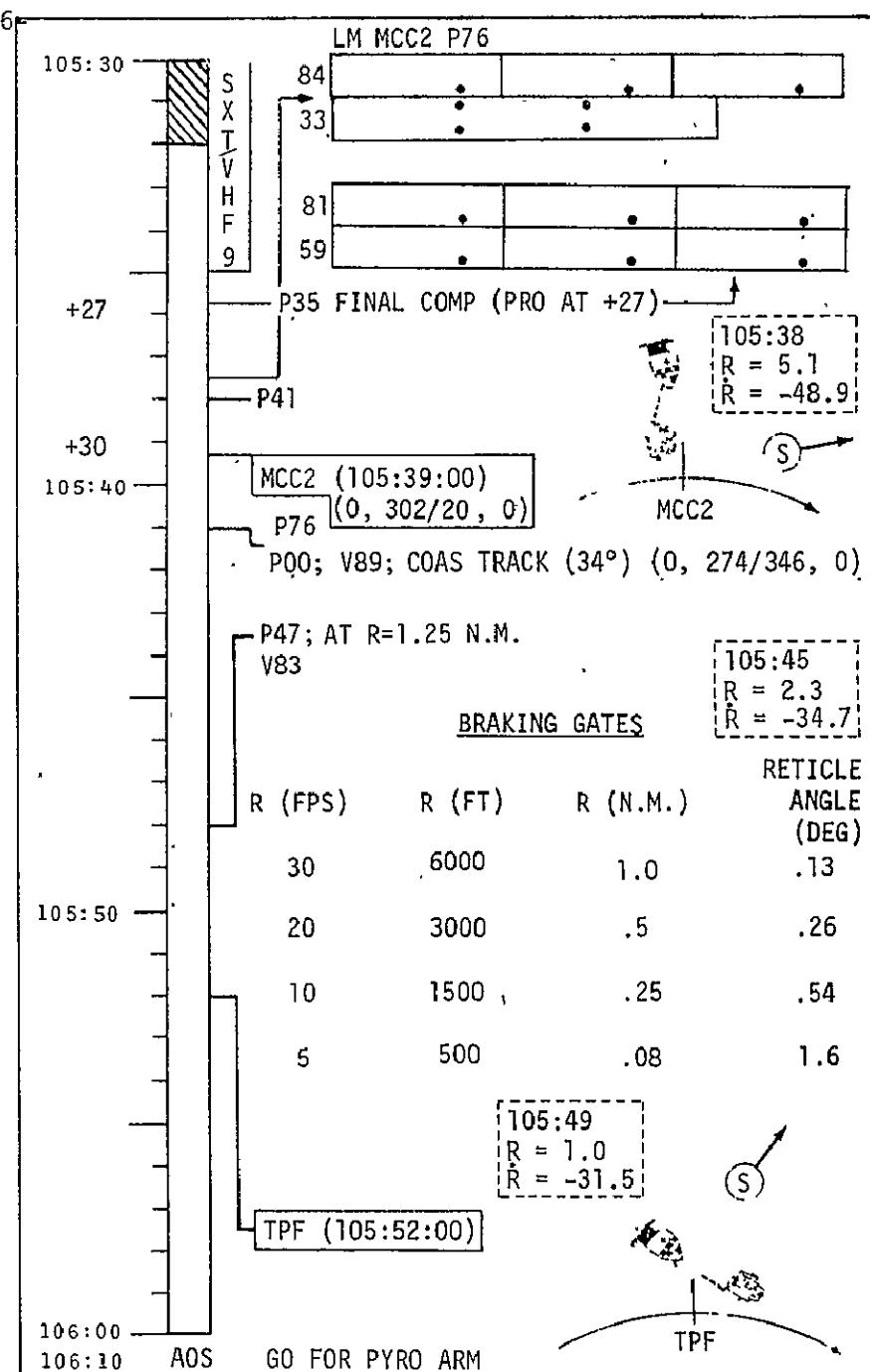
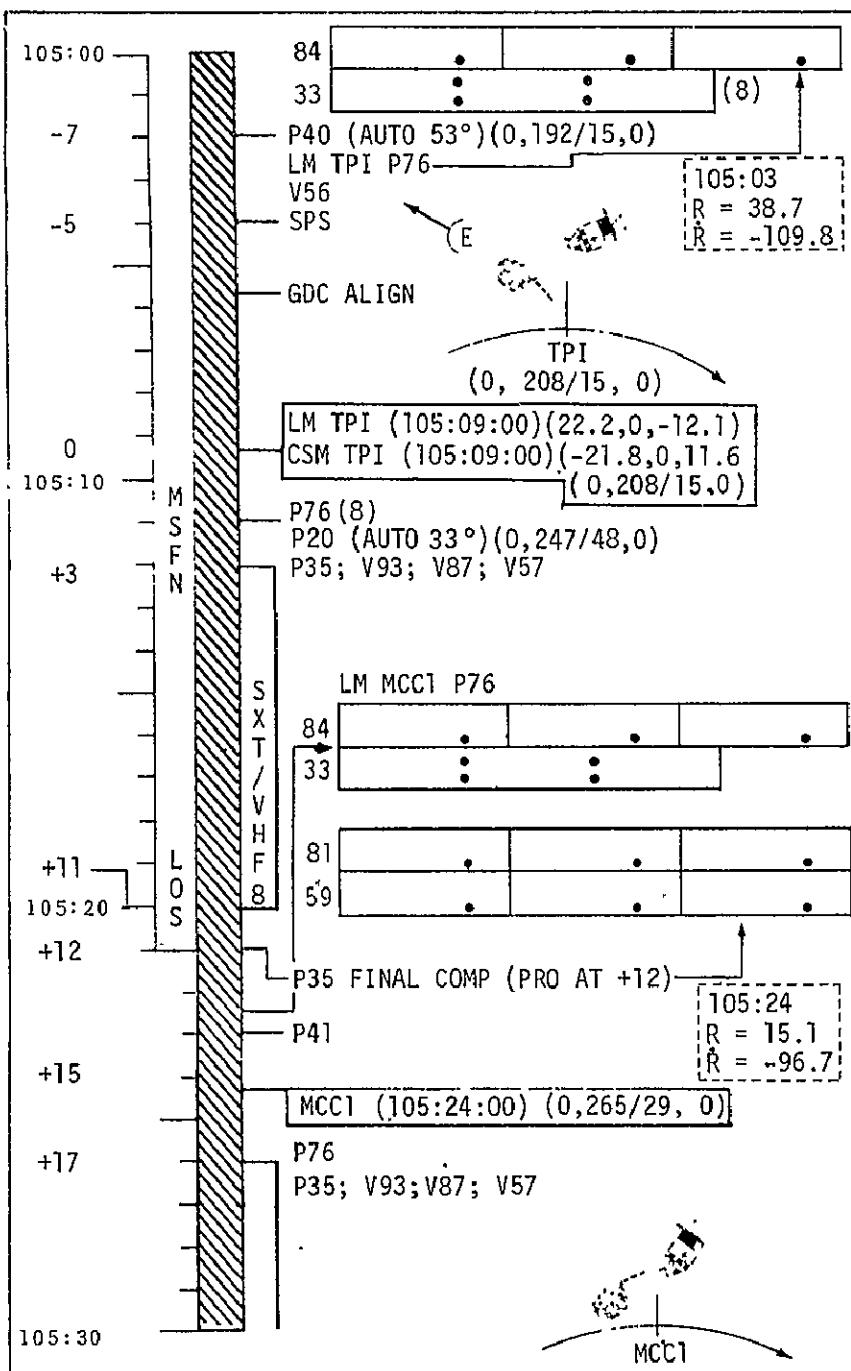












5.0 NOMINAL MAJOR PROCEDURES SUMMARY

The following sections present a summary timeline and checklist of major CSM activities during the nominal Mission F LM active rendezvous. The timeline in Section 5.1 will aid in interpreting the data included in the onboard rendezvous checklist and serves as a time correlated table of contents for the nominal detailed procedures which follow in Section 6.0. In addition, Section 5.2 includes a one-page summary checklist for the nominal mission in a format similar to that utilized for the LM rescue onboard data.

5.1 SUMMARY TIMELINE

-28-

GET	EVENT	PROG	GET	EVENT	PROG
(97+03)	MOVE TO COMMAND SEAT		(97+40)	V49, LOAD N22(180,14,14) MNVR TO UNDOCKING ATTITUDE (INERTIAL SEP ATT EXCEPT ROLL AND 14 DEG YAW) (180,282/14,14)	
(97+05)	MCC-H UPLINK (RLS AS REQUIRED (P00) AND CSM STATE VECTOR), V66 COPY CSM SEP PADS DON HELMET AND GLOVES CSM/LM COMM CHECK		(97+46)	CMC MODE-FREE (AS REQD FOR LM AGS CALIB) CMC MODE-AUTO (AFTER 32 SEC) PERFORM SYSTEMS CHECKS AND SWITCH VERIFICATION	
(97+14)	VERIFY DAP WIDE DB FOR LM RCS CHECKS		(97+50)	YAW 14 DEG LEFT AFTER LM AGS CALIB (180,312/14,0) MOVE TO LEB	
(97+20)	VERIFY AUTO RCS SEL -C4 (-PITCH-X) -OFF -B3 (+YAW-X) -OFF FOR LM RDR SELF TEST VERIFY RNDZ XPNDR=OFF		(97+52)	RNDZ XPNDR CHECKS MOVE TO COMMAND SEAT	
(97+27)	LOS		(98+05)	GDC ALIGN TO IMU	
(97+30)	CONFIGURE CAMERA		(98+06)	VERIFY ORDEAL (V83)	
(97+35)	LOAD DAP (21112), (V46) (11111)		(98+07)	RELOAD DAP (11102), (V46) (11111) SC CONT-SCS	
(97+38)	SUNUP	-25	(98+10+00)	***** UNDOCKING (180,12/14,0) ***** STATION KEEP AT 40 FT	
	VHF AM A-SIMPLEX				

GET	EVENT	PROG	-29-	GET	EVENT	PROG
	DV CG-CSM			(98+39)	RENDEZVOUS NAV PROGRAM	
	RNDZ XPNDR PWR-PWR(VERIFY)				AUTO MNVR TO SXT TRACK	(P20)
	ROLL LEFT 180 DEG AT 2 DEG/SEC				(40 DEG)	
(98+13)	(0,50/14,0)				(0,142/54,0)	
	AOS				VHF ANT-RT	
	ACQUIRE MSFN (V64)	(P00)			VHF AM B-DUPLEX/RANGING	
	ACTIVATE COLOR TV				MONITOR LM RR CHECKS	
	TAKE PHOTOS				COMPARE EMS VHF AND V83 RANGE	
-15	INSPECT LM DURING LM 360 DEG YAW				MOVE TO LEB	
(98+20)	LM STATION KEEP		(98+50)		PERFORM OPTICS CHECKS	
(98+23)	COPY PADS		(98+55)		SUNDOWN	
(98+29)	RELOAD DAP(11102),(V46)				IMU REALIGN TO REFSMMAT	(P52)
	(01111)				(OPTION 3)	
	SC CONT-CMC				COPY GYRO TORQUE ANGLES	
-05	(98+30)	EXTERNAL DV TARGETING	(P30)			
	(SEPARATION)				MOVE TO CMD SEAT	
-03	(98+32)	RCS THRUST PROGRAM	(P41)		GDC ALIGN TO IMU	
	AUTO MNVR TO BURN ATTITUDE				VERIFY ORDEAL(V83)	
	(TRIM)					
	RCS THRUST SETUP	(P41)			MCC-H UPLINK(CSM AND	(P00)
0	(98+35+16)	*****			AND LM VECTORS)	
	SEPARATION BURN(0,0,-2.5)				VHF AM A-SIMPLEX/DATA	
	(0,90/14,0)				LOS	
	THRUST(=)X(2.5,0,0) TO (5.0,0,0)				RENDEZVOUS NAV PROGRAM	(P20)
	*****				AUTO MNVR TO SXT TRACK	
					(55 DEG)	
					(0,206/332,0)	
					MOVE TO LEB	

GET	EVENT	PROG	-30-	GET	EVENT	PROG
-02	(99+32) LOAD TARGET DV PROGRAM (DOI)	(P76)		(100+12)	AOS	
	(99+33+59) ***** LM DOI BURN(-71.1,0,-.3) *****			(100+20)	ACQUIRE MSFN (V64)	(P00)
	MOVE TO CMD SEAT	-21		(100+25)	OBTAIN GO/NO GO FOR PDI ABORT	
	VHF AM B-DUPLEX/RANGING				MOVE TO LEB	
	VHF RCV ONLY-OFF				V67 LOAD WI(002.89,0017.3,00001)	
	COMPUTE RDOT FOR LM					
	LM ORBIT PARAMETER DISPLAY(V82) (9X60)	-19			RENDEZVOUS NAV PROGRAM	(P20)
	CONFIRM LM DOI				AUTO MNVR TO SXT TRACK	
	INCORPORATE P76 DATA (DOI)	-9			(25 DEG)	
		0			(180+293/243,0)	
(99+37)	SUNUP			(100+27)	CALL MARKING ROUTINES(V87+V57)	
(99+38)	RENDEZVOUS NAV PROGRAM POSSIBLE AUTO MNVR TO SXT TRACK (TRIM) (0,227/317,0)	(P20)			TERMINATE SXT/VHF MARKS(V88)	
	MOVE TO LEB				*****	
(99+44)	INITIATE LM OPTICS TRACK			(100+47)	LM PHASING BURN(170.4,0,-95.6) (180,307/193,0)	
-39	(100+07) MOVE TO CMD SEAT ROLL 180 DEG AND PITCH UP 70 DEG AT 1 DEG/SEC (180,278/268,0)				*****	
					TARGET DV PROGRAM (PHASING)	(P76)
					LM ORBIT PARAMETER DISPLAY(V82) (10X196)	
				(100+48)	RENDEZVOUS NAV PROGRAM POSSIBLE AUTO MNVR TO SXT TRACK (TRIM) (180,308/188,0)	(P20)
				(100+49)	SUNDOWN	

GET	EVENT	PROG	-31-	GET	EVENT	PROG
+05	(100+51) CALL MARKING ROUTINES(V87,V57) (PROCESS 3 MARKS,V93)		-32	(102+14)	ACQUIRE MSFN (V64) MCC-H UPLINK(CSM VECTOR)	(P00)
	(101+01) TERMINATE SXT MARKS				COPY LM INSERTION P76 PAD UPDATE AS REQUIRED	
	(101+05) COPY INITIAL CSM BACKUP INSERTION PAD				COPY CSM BACKUP INSERTION PAD UPDATE AS REQUIRED	
	(101+10) COPY LM INSERTION P76 PAD	-16	(102+29)		VERIFY ORDEAL(V83)	
+28	(101+14) CALL MARKING ROUTINE(V57)			(102+30)	EXTERNAL DV TARGETING (INSERTION)	(P30)
+38	(101+24) TERMINATE SXT/VHF MARKS,V88 LOS	-14	(102+32)		SPS THRUST PROGRAM (INSERTION)	(P40)
	(101+35) SUNUP MOVE TO COMMAND SEAT				AUTO MNVR TO BURN ATTITUDE (110 DEG) (180,3/272,0)	
	(101+40) GDC ALIGN TO IMU VERIFY ORDEAL(V83) MOVE TO LEB	-09	(102+37)		SPS THRUST SETUP (INSERTION)	(P40)
-57	(101+49) CALL MARKING ROUTINES(V87,V57) (PROCESS 3 MARKS,V93)	-05	(102+41)		GDC ALIGN	
-37	(102+09) TERMINATE SXT/VHF MARKS,V88 MOVE TO CMD SEAT	-03	(102+43+18)		***** LM INSERTION BURN (-189.2,0,-83.9) *****	
	(102+11) AOS PITCH DOWN 50 DEG (180,174/162,0)	0	(102+46+18)		***** CSM BACKUP INSERTION BURN (175.4,0,-107.1) (180,32/272,0) *****	

MISSION F NOMINAL TIMELINE APRIL 25, 1969

GET	EVENT	PROG	-32-	GET	EVENT	PROG	
	TARGET DV PROGRAM (INSERTION)	(P76)	-09	(103+25)	CSI TARGETING FINAL COMP	(P32)	
	LM ORBIT PARAMETER DISPLAY(V82) (10X45)				COPY CSM CSI SOLUTION		
(102+47)	SUNDOWN			(103+28)	LOAD N81 VGY WITH(=) CSM YDOT		
	MOVE TO LEB				COPY LM CSI PAD		
	IMU REALIGN TO REFSMMAT (OPTION 3)	(P52)		(103+29)	SPS THRUST PROGRAM,V56	(P40)	
	COPY GYRO TORQUE ANGLES				AUTO MNVR TO BURN ATTITUDE		
(102+52)	MCC-H UPLINK(LM VECTOR)	(P00)	-04		(11 DEG)		
-41	(102+56)	RENDEZVOUS NAV PROGRAM	(P20)	-03	(180+166/277+0)		
	AUTO MNVR TO SXT TRACK(101 DEG)						
	(180,162/13,0)						
(102+59)	CSI TARGETING PROGRAM	(P32)	(P20)	0	(103+30)	SPS THRUST SETUP	(P40)
	OBTAIN LM CSI AND TPI TIGNS,						
	LOAD CSI TARGETING DATA			(103+31)	GDC ALIGN		
-32	(103+02)	REINITIALZE W MATRIX (V93)					
	CALL MARKING ROUTINE(V87,V57)			(103+33)	SUNUP		
-27	(103+07)	TERMINATE SXT MARKS					
	REINITIALIZE W MATRIX (V93)			(103+33+46)	*****		
	MOVE TO CMD SEAT				LM CSI BURN(50,5,0,0)		
-22	(103+12)	OUT OF PLANE DATA(V90)			*****		
	(CSM AND LM)				*****		
	VOICE LM Y DOT TO LM				CSM BACKUP CSI BURN		
	(103+23)	LOS			(52,2,0,0)		
					(180,180/277+0)		

					TARGET DV PROGRAM	(P76)	
					(CSI)		
					LM ORBIT PARAMETER DISPLAY(V82)		
					(45X45)		

GET	EVENT	PROG	-33-	GET	EVENT	PROG
	(103+38)				(104+08)	
	RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (39 DEG) (180,155/239,0)	(P20)			(104+09)	CALL MARKING ROUTINE(V87,V57) AOS (PROCESS THREE MARKS,V93)
-54	(103+41)		-12	(104+20)		
	CALL MARKING ROUTINES(V87,V57) (PROCESS 3 MARKS) V67 LOAD WR(000.57,0003.4,00001)				TERMINATE SXT MARKS OUT OF PLANE DATA(V90) VOICE LM YDOT TO LM MOVE TO CMD SEAT	
	(103+44)	CALL MARKING ROUTINE (V57)		(104+22)		
	(103+55)	TERMINATE SXT MARKS COPY LM PC TIGN OUT OF PLANE DATA(V90) VOICE LM YDOT TO LM COPY LM PC PAD	-07	(104+25)	ROLL 180 DEG AT 2 DEG/SEC ACQUIRE MSFN (0,167/108,0)	(P33)
	(103+57)	CALL MARKING ROUTINE(V57)		(104+28)	CDH TARGETING FINAL COMP COPY CSM CDH SOLUTION	
30	(104+02+00)	TERMINATE MARKS ***** LM PLANE CHANGE(0,0,0) ***** -02	-03	(104+29)	COPY LM CDH PAD RCS THRUST PROGRAM BYPASS BURN ATTITUDE MNVR FOR SMALL BURNS	(P41)
		LOAD TARGET DV PROGRAM (P76) INCORPORATE P76 DATA (PLANE CHANGE)		(104+30)	RCS THRUST SETUP GDC ALIGN	(P41)
	(104+05)	CDH TARGETING PROGRAM (P33) POSSIBLE AUTO MNVR TO SXT TRACK (TRIM) (180,157/158,0) OBTAIN LM CDH TIGN LOAD CDH TARGETING DATA	0	(104+31+42)	***** LM CDH BURN (-.8,0,3,3) ***** ***** ***** CSM CDH BURN (-.6,0,-5,0) (0,188/108,0) *****	

MISSION F NOMINAL TIMELINE APRIL 25, 1969

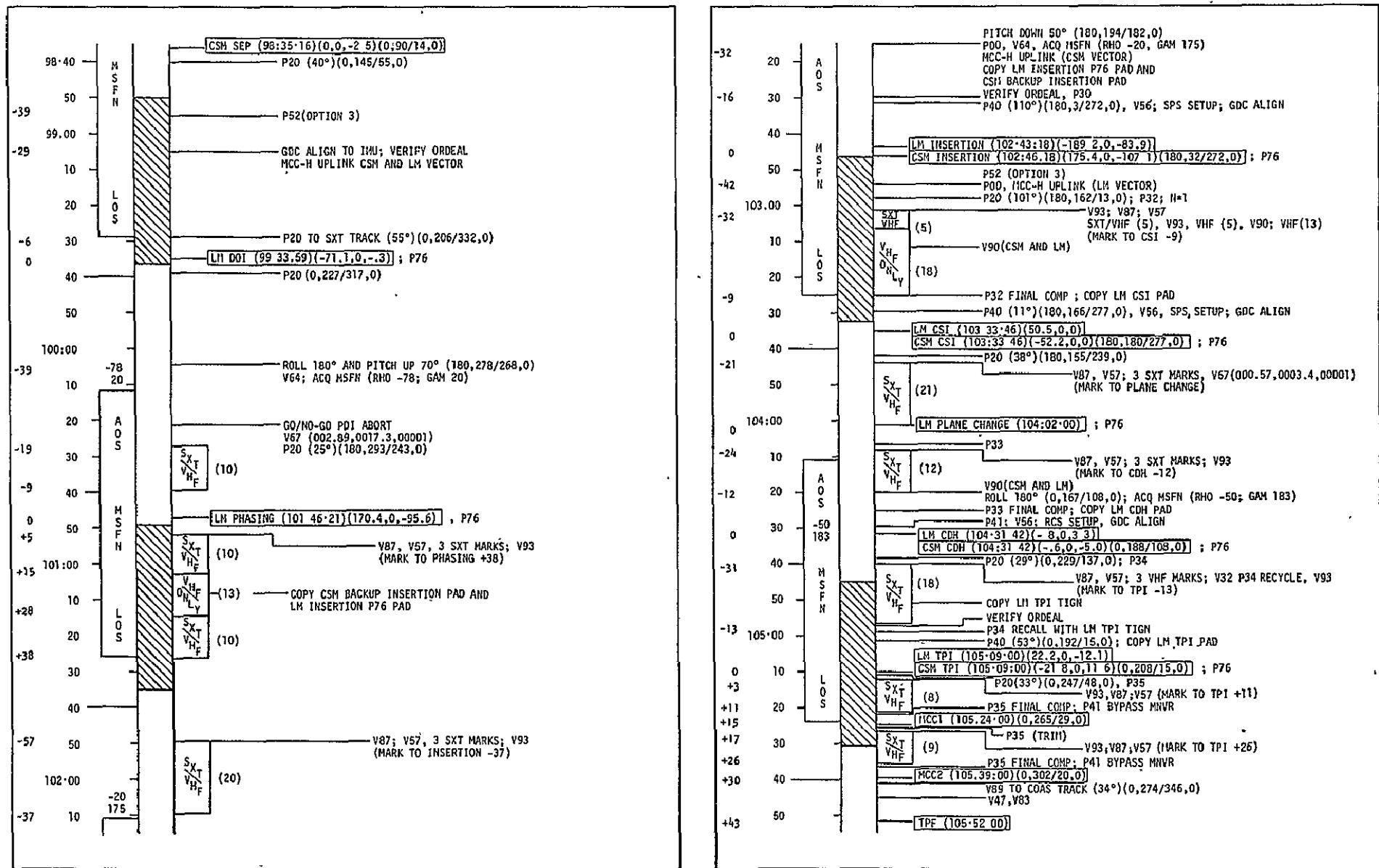
GET	EVENT	PROG	-34-	GET	EVENT	PROG	
(104+34)	TARGET DV PROGRAM (CDH)	(P76)	+05	(105+04)	SPS THRUST SETUP (TPI)	(P40)	
	RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (29 DEG) (0,229/137,0)	(P20)	+03	(105+06)	GDC ALIGN		
	MOVE TO LEB		0	(105+09+00)	LM TPI BURN (22.2,0,-12.1)		
	TPI TARGETING PROGRAM (P34) (P20)				CSM TPI BACKUP BURN (-21.8,0,11.6)		
	LOAD TPI TARGETING DATA WITH ELEV OPTION				(0,208/15,0)		
-31	(104+38)	CALL MARKING ROUTINE(V87,V57)			TARGET DV PROGRAM (TPI)	(P76)	
	AFTER 3 MARKS PROCESSED				RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK	(P20)	
	RECYCLE TPI TARGETING PROGRAM	(P34)	+02		(33 DEG)		
	CALL MARKING ROUTINE(V57)				MOVE TO LEB		
(104+45)	SUNDOWN				MCC TARGETING PROGRAM	(P35)	
(104+50)	COPY LM TPI TIGN				CALL MARKING ROUTINE(V93,V87,V57)		
-13	(104+56)	TERMINATE SXT MARKS			TERMINATE SXT MARKS		
	MOVE TO CMD SEAT		+03		MOVE TO COMMAND SEAT		
	VERIFY ORDEAL(V83)				LOS		
-11	(104+58)	TPI TARGETING PROGRAM (TIGN OPTION WITH LM TIGN)	(P34)	+11			
-07	(105+02)	SPS THRUST PROGRAM V56 AUTO MNVR TO BURN ATTITUDE (53 DEG) (0,192/15,0)	(P40)	+12	(105+21)	MCC1 TARGETING FINAL COMP COPY CSM MCC1 SOLUTION COPY LM MCC1 PAD	(P35)
	COPY LM TPI P76 PAD						

GET	EVENT	PROG	-35-	GET	EVENT	PROG
+13.5	RCS THRUST PROGRAM BYPASS BURN ATTITUDE MNVR	(P41)		+30	(105+39+00)	
+15	(105+24+00)				***** LM MCC2 BURN *****	
	***** LM MCC1 THRUST *****				***** CSM MCC2 BACKUP BURN (0,302/20,0) *****	
	***** CSM MCC1 BACKUP BURN (0,265/29,0) *****				TARGET DV PROGRAM (MCC2) *****	
	TARGET DV PROGRAM (MCC1)	(P76)			AUTO MNVR TO COAS TRACK(V89) (P00) (34 DEG) (0+274/346,0)	
+16	MCC2 TARGETING PROGRAM (P35) (P20) POSSIBLE AUTO MNVR TO SXT TRACK (TRIM)		+39	(105+48)	THRUST MONITORING PROGRAM (P47) (KEY V83 AND MONITOR LM BRAKING) *****	
+17	CALL MARKING ROUTINE(V93,V87,V57)				* BRAKING GATES RETICLE ANGLE * * 30 FPS AT 6000 FT .13. DEG. * * 20 FPS AT 3000 FT .26 DEG. * * 10 FPS AT 2500 FT .54 DEG. * * 5 FPS AT 500 FT 1.60 DEG. * *****	
	(105+32)	SUNUP				
+26	TERMINATE SXT MARKS					
	MOVE TO CMD SEAT		+43	(105+52)		
+27	MCC2 TARGETING FINAL COMP (P35) COPY CSM MCC2 SOLUTION COPY LM MCC2 PAD				***** TPF (0,310/353,0) *****	
+28.5	RCS THRUST PROGRAM BYPASS BURN ATTITUDE MNVR	(P41)				

MISSION F NOMINAL TIMELINE APRIL 25, 1969

5.2 NOMINAL SUMMARY CHECKLIST

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6.0 NOMINAL DETAILED PROCEDURES

The following four sections contain 1) The procedural ground rules assumed when generating procedures, 2) The detailed nominal procedures, 3) A summary of the rendezvous navigation schedule including the angle between the LOS to the Sun and LOS to the LM, and 4) A Summary of the inertial roll gimbal angle and the ORDEAL pitch and inertial pitch gimbal angles during the nominal F Mission.

6.1 Procedures Ground Rules

- 6.1.1 The CSM will be targeted for a TIGN three minutes after the LM TIGN for the insertion burn. No bias will be used for the CSI, CDH, or TPI burns.
- 6.1.2 The CSM will obtain from the LM the CSI, CDH, and TPI times utilized in the LGC targeting programs. These TIGN's loaded into the CMC targeting programs.
- 6.1.3 LM burn data will be incorporated into the CMC LM state vector using Program P76 following each LM burn. No attempt will be made to account for LM burn residuals. The LM burn TIGN loaded in Program P76 will be biased by a fixed number simulating an impulsive LM burn. The non-zero LM burns considered and the corresponding bias times are DOI (20 Sec), Phasing (32 Sec), Insertion (8 Sec), CSI (16 Sec), and TPI (8 Sec).
- 6.1.4 The CSM will compute using V90, the out-of-plane velocity of the LM prior to the CSI, Plane Change, and CDH burns for use in the LM targeting programs.
- 6.1.5 The ordeal setup on FDAO 2 will be verified approximately once per orbit.
- 6.1.6 The GDC will be aligned to the IMU prior to each backup CSM burn.
- 6.1.7 No burn attitude verification using stars or the horizon will be made in the CSM.
- 6.1.8 The PIPA bias determination, EMS DV test, and EMS accelerometer check, identified in the AOH as required

before each SPS burn, need be scheduled only prior to the entire rendezvous sequence.

- 6.1.9 The SM RCS propulsion check identified in the AOH as required before each SPS burn, should not include recording the values monitored.
- 6.1.10 It is assumed that the CMP is able to move to the LEB or command seat during automatic attitude maneuvers and the time required to move is less than one minute.
- 6.1.11 The procedures contained herein reflect the COMANCHE 44, 45 programs and routines.
- 6.1.12 The minimum times to be allowed for the CMC programs are:
 - A. P52 5 Min.
 - B. P76 1 Min.
 - C. P30 2 Min.
 - D. P32 Final Comp 3 Min.
 - E. P33 Final Comp 3 Min.
 - F. P34 Final Comp 3 Min.
 - G. P35 Final Comp 1.5 Min.
 - H. P41 (Excluding ATT Mnvr and Including RCS Setup) 1.5 Min.
 - I. P40 (Excluding ATT Mnvr and Including SPS Setup) 5 Min
- 6.1.13 All automatic maneuvers (DAP) are made at a rate of .5 Deg/Sec.
- 6.1.14 The state vector of the active vehicle (LM) will be updated in the rendezvous navigation Program, P20.

- 6.1.15 The SXT navigation mark frequency during a tracking period is one per minute.
- 6.1.16 Recycles for CMC targeting Programs P32, P33, and P34 during mark periods have not been scheduled. They will be included, as desired for solution comparison purposes, when they do not preclude taking the minimum number of SXT and VHF marks prior to a burn.
- 6.1.17 The delta time between the PRO for Final Comp in Program P35 and the MCC is three minutes.
- 6.1.18 Program P20 will be terminated (V56) in the thrust program prior to each backup CSM burn. This will necessitate calling Program P20 after the burn in addition to a pre-thrust program.
Exceptions: Insertion (P00 does it) and the MCC burns.
- 6.1.19 The automatic star selection routine in Program P52 will be used during each IMU realignment.
- 6.1.20 The CSM attitude has been specified to be compatible with HGA communications when no other attitude constraint prohibits HGA coverage.
- 6.1.21 The procedures have been developed to be compatible with Mission G1 procedural requirements.
- 6.1.22 The SPS gimbal motors will not be activated for a CSM backup of a LM burn until the CSM knows it must become active.

- 6.1.23 Backup charts for the CSI, CDH, TPI, and MCC burns will not be used in the CSM.
- 6.1.24 VHF navigation marks cannot be taken at ranges greater than 327 nautical miles.

6.2 DETAILED PROCEDURES

MISSION F RENDEZVOUS PROCEDURES

ASSUMED INITIAL SWITCH SETTINGS

****CDR SWITCH SETTINGS****

CMC ATT=IMU
FDAI SCALE=5/1
FDAI SELECT=1/2
FDAI SOURCE=ATT SET
ATT SET=IMU
MAN ATT(3)=RATE CMD
LIMIT CYCLE=OFF
ATT DB=MAX
RATE=LOW
THC=LOCKED
RHC=LOCKED
TRANS CONT PWR=ON(UP)
ROT CONTR PWR NORMAL (BOTH)=
AC/DC
ROT CONTR PWR DIRECT (BOTH)=
OFF
SC CONT=CMC
CMC MODE=AUTO
BMAG MODE(3)=RATE2
SPS THRUST DIRECT=NORMAL
DV THRUST(AIR)=OFF

SCS TVC (BOTH)=RATE CMD
SPS GMBL MTR(4)=OFF
DV CG=LM/CSM
IMU CAGE=OFF
EMS ROLL=OFF
.05 G=OFF
LV/SPS IND (BOTH)=PC,GPI
TVC GMBL DRIVE (BOTH)=1
FCSM(A,B)=RESET/OVERRIDE
EMS FUNCTION=OFF
EMS MODE=STBY
UP TLM(CM+IU)=BLOCK
RCS TRNFR=SM
PANEL 8 CB CLOSED EXCEPT
RCS LOGIC(2)
DOCK PROBE(2)
ELS BAT A,BAT B
PL VENT FLT/PL
FLOAT BAG(3)
SECS ARM BAT A,BAT B
AUTO RCS SEL(16)=MNA
EXCEPT
-C4(-PITCH-X)=OFF
-B3(+YAW-X)=OFF
TVC SERVO PWR (BOTH)=OFF
FDAI/GPI PWR=BOTH
LOGIC PWR 2/3=ON(UP)
SCS ELECT PWR=GDC/ECA
SIG COND/DRV RVR BIAS PWR(2)=AC1
BMAG PWR(BOTH)=ON
COAS PWR (L WIND)=ON

FDAI SW1-INERTIAL
FDAI SW2-ORB RATE
EARTH/LUNAR=LUNAR
ALT SET KNOB TO AVE ALT
MODE=OPR/SLOW

LMP SWITCH SETTINGS

G/N PWR-AC1
MN BUS TIE(2)-OFF
NONESS BUS-OFF
PANEL 5 G/N CB(10)-CLOSED
PANEL 229 CB CLOSED EXCEPT
MN REL PYRO(2)

CMP SWITCH SETTINGS

G/N PWR OPTICS-ON
G/N PWR, IMU-ON
RNDZ XPONDER-OFF
OPTICS MODE=CMC
ZERO OPTICS-ZERO
CONT SPEED-LO
CONT COUPLING-RESOLVED
SCT TRUN-SLAVE TO SAT
CONDITION LAMPS-ON
UP TLM(LEB)-ACCEPT
C/W=NORMAL
C/W=CSM
C/W=POWER

* TIMELINED RENDEZVOUS PROCEDURES *

* PROCEDURES THRU UNDOCKING *

(97+03)

MOVE TO CMD SEAT

(97+05)

MCC-H UPLINK RLS (AS REQUIRED)
AND CSM STATE VECTOR

P00

KEY V37E00E

UP TLM(CM) (MDC)=ACCEPT
MONITOR UPLINK ACT LT=ON
MONITOR GND UPLINK
MONITOR UPLINK ACT LT=OFF
UP TLM(CM) (MDC)=BLOCK
CONFIRM COMP LT=OFF
KEY V66E

COPY SEPARATION PAD
DON HELMET AND GLOVES
CSM/LM COMM CHECK

(97+14)

VERIFY DAP WIDE DB FOR
LM RCS CHECKS

KEY V48E

F 04 46 (DAP CONFIGURATION)

VERIFY R1 = XXXIX

PRO

F 06 47 (CSM AND LM WT)

PRO

F 06 48 (SPS GMBL TRIM)

PRO

SET MDC ET COUNTING DOWN TO
SEPARATION TIGN
(98+55+44)

(97+20)

VERIFY AUTO RCS SEL
-C4 (+PITCH=X)=OFF, AND
-B3 (+YAW=X)=OFF (FOR LM
RDR SELF TEST)
VERIFY RNDZ XPNDR=OFF

LOSS OF SIGNAL

(97+27)

CONFIGURE CAMERA:
CM/SEQ/18/CEX=BRKT(RH WIN)
MIR(F11.250, INFINITY) 6FPS,
15 MIN

(07+35)

LOAD DAP
KEY V48E
F 04 46 (DAP CONFIGURATION)
LOAD

21112
11111

PRO
F 06 47 (CSM AND LM WT)
LOAD

+37768
+30847

PRO
F 06 48 (SPS GMBL TRIM)
LOAD

+00052
+00059

PRO
KEY V46E
KEY V62E

(97+38)

SUNUP
VHF AM A-SIMPLEX
VHF RCV ONLY-B DATA

(97+40)

MANEUVER TO UNDOCKING ATTITUDE
(INERTIAL SEP ATT EXCEPT ROLL
AND 14 DEG YAW)

KEY V49E
F 06 22 (COMMANDER R,P,Y)
LOAD(180,14,14)

PRO

F 50 18 (COMMANDER R,P,Y)
PRO

06 18 (COMMANDER R,P,Y)
MONITOR AUTO MANEUVER
F 50 18 (COMMANDER R,P,Y)
(180,282/14,14)
KEY ENTER

(97+46)

CMC MODE=FREE (AS REQ FOR AGS
CALIB)
CMC MODE=AUTO (AFTER 32 SEC)
PERFORM SYSTEMS CHECK
AND SWITCH VERIFICATION

(97+50)

YAW 14 DEG LEFT AFTER
LM AGS CALIBRATION
(180,312/14,0)

(97+52)

MOVE TO LEB

RNDZ XPNDR ACTIVATION
AND SELF TEST

CB RNDZ XPNDR FLT BUS=CLOSE
RNDZ XPNDR PWR=HTR FOR 1 MIN
FOR SELF TEST

RNDZ XPNDR PWR=PWR
SYS TEST (LH)=XPNDR
SYS TEST (RH)=A(RRT XMTR OUT
PWR)
RNDZ XPNDR TEST=TEST (HOLD)
SYS TEST IND=GREATER THAN 1 VDC
SYS TEST (RH)=B(RRT AGC SIG)
RNDZ XPNDR TEST=TEST (HOLD)
SYS TEST IND=GREATER THAN 1
VDC

RNDZ XPNDR TEST-OPERATE
SYS TEST IND=0=4.5 VDC
SYS TEST(RH)=C(RRT FREQ LOCK)
SYS TEST IND
-LESS THAN .8 VDC UNLOCKED
-MORE THAN 4 VDC LOCKED
SYS TEST(RH)=B(RRT AGC SIG
OPERATE)

(98+05) MOVE TO CMD SEAT
GDC ALIGN TO IMU
16 20 KEY V16N20E
(R,P,Y)
ATT SET THUMRWHEELS TO N20
FDAI SELECT-1
NULL ATT ERROR NEEDLES
ON FDAI 1 WITH ATT
SET THUMBWHEELS
FDAI SELECT-1/2
ATT SET-GDC
DEPRESS GDC ALIGN PB
ATT SET-IMU
KEY RELEASE

(98+06) **ORDEAL VERIFICATION**
F 04 06 KEY V82E
(VEHICLE OPTION CODES)
PRO
F 16 44 (HA,HP,TFF)
CALC AVE ALT
PRO
ALT SET KNOB TO AVE ALT

F 16 54 KEY V83E
(R,RDOT,THETA)
VERIFY R=RDOT=0
IF NOT,KEY V66E
SLEW/ADJUST FDAI TO THETA
PRO

(98+07) S/C CONT-SCS
LOAD DAP

F 04 46 KEY V48E
(DAP CONFIGURATION)
LOAD
11102
11111
B
PRO

F 06 47 (CSM AND LM WT)
PRO

F 06 48 (SPS GMBL TRIM)
PRO
KEY V46E

PROCEDURES FOR UNDOCKING

INSTALL DOCKING TARGET
RATE-HI
ROT CONT PWR DIR(BOTH)=MNA/MNB
RMC-ARMED
THC-ARMED
CB ROCK PROBE(2)=CLOSED
EVNT TMR RESET-DOWN
EVNT TMR START (ON CDR SIGNAL)

-25 (98+10+00)

PROBE EXTD/REL-EXTD/REL (HOLD
UNTIL LM SEP PLUS 5 SEC)
PROBE EXTD/REL TB(2)-GRAY TO
BP TO GRAY
MONITOR LM UNDOCKING/MAINTAIN
UNDOCKING ATTITUDE
(180+12/14,0)
PROBE EXTD/REL-RETR
DV CG=CSM
AUTO RCS SELECT
-B3(+Y&W-X)=MNA
-C4(-PITCH-X)=MNA
ROT CONT PWR DIR(BOTH)=OFF
BMAG MODE(3)=ATT 1/RATE 2
CB DOCK PROBE(2)=OPEN

***** -15 (98+20) **LM STATION KEEPING**
* PROCEDURES FOR *
* UNDOCKING THRU SEPARATION *

THC=LOCKED
RHC=LOCKED

CSM STATION KEEPING -12 (98+23)
ATT DB=MIN
RNDZ XPNDR PWR=PWR(VERIFY)
PERFORM STATION KEEPING
MANEUVERS/MAINTAIN 40 FT
SEPARATION
(98+29) **RELOAD DAP**
FDAI SCALE=5/5
ROLL 180 DEG LEFT AT
2 DEG/SEC
(0, 50, 14, 0)
FDAI SCALE=5/1
KEY V48E
F 04 46 (DAP CONFIGURATION)
LOAD
11112
01111
B
PRO
F 06 47 (CSM AND LM WT)
PRO
F 06 48 (SPS GMBL TRIM)
PRO
BMAG MODE(3)-RATE 2
KEY V46E
S/C CONT=CMC

ACQUIRE MSFN
KEY V64E
F 06 51 (RHO, GAMMA, BLANK)
SLEW HI. GAIN ANT
(98+13)
ACQUISITION OF SIGNAL
ACQUIRE MSFN
PRO
ACTIVATE COLOR TV
TAKE PHOTOS
INSPECT LM DURING LM 360 DEG
YAW
-5 (98+30)
P30
KEY V37E30E
F 06 33 (GET OF SEP)
LOAD GET OF SEPARATION
(98+55+64)
PRO

F 06 81 (VG-LV)	-00+40	MONITOR COMP ACTV LT-OUT
LOAD (0,0,-2.5)		
PRO	-00+35	DSKY BLANKS
F 06 42 (HA,HP,VG)	-00+30	
VERIFY VG=2.5		
PRO		
F 16 45 (MKS,TFI,MGA)		16 85 (VG-BODY) (AVG G ON)
SET MDC ET TO TFI,COUNTING		COMP ACTV LT=2 SEC FLASH
DOWN	-00+25	EMS MODE=NORMAL
CONFIRM MGA LESS THAN 45 DEG		
PRO		
F 37 BB		CK VG FOR HI PIPA BIAS (LESS THAN 2.0 FPS PER 5 SEC)
		RHC=ARMED

*3 (98+32) **RCS SEPARATION BURN SETUP**

0+00 (98+35+16)

F 16 85 (VG-BODY)
SET MDC ET COUNTING UP
FROM SEPARATION
INCREASE VG-BODY FROM
(+2.5,0,0) TO (+5.0,0,0)
BY THRUSTING AFT

CSM PERFORMS SEPARATION BURN
(0,0,-2.5)
(0,90/14,0)
WHEN BURN COMPLETE, VOICE
CONFIRMATION TO LM

P41 KEY 41E
F 50 18 (COMMENDED R,P,Y)
PRO
06 18 (COMMENDED R,P,Y)
MONITOR ATT TRIM
F 50 18 (COMMENDED R,P,Y)
RHC=ARMED
ALIGN S/C IN ROLL
PRO
06 18 (COMMENDED R,P,Y)
MONITOR ATT TRIM
F 50 18 (COMMENDED R,P,Y)
KEY ENTER
06 85 (VG-BODY)
MONITOR COMP ACTV LT
BMAG MODE(3)=ATT 1/RATE 2
RATE=LOW
EMS FUNCTION=DV SET
SET DV IND TO +2.5
EMS FUNCTION=DV

-50-

EMS MODE=STBY
RECORD DV IND
EMS FUNCTION=OFF
RHC=LOCKED
THC=LOCKED
BMAG MODE(3)=RATE 2
AUTO RCS SEL A/C ROLL(4)=OFF
PRO

F 37 BB

MISSION F DETAILED RENDEZVOUS PROCEDURES, REVISION A, APRIL 25, 1969

* PROCEDURES FOR *
* SEPARATION THRU PHASING *

(98+39) **MANEUVER TO SXT TRACK ATTITUDE**

P20 KEY 20E
F 50 18 (COMMANDER R,P,Y)
PRO
06 18 (COMMANDER R,P,Y)
MONITOR AUTO MNVR
F 50 18 (COMMANDER R,P,Y)
(0,142/54,0)
KEY ENTER

VHF ANT=RT
EMS FUNCTION-VHF RNG
EMS MODE-VHF RNG
VHF AM A-OFF
VHF AM B-DUPLEX
VHF RCV ONLY-OFF
VHF RANGING=RNG
VHF RNG=RESET
MONITOR EMS FOR RANGE

KEY V83E
F 16 54 (R,RDOT,THETA)
COMPARE EMS AND V83 RANGE

COMPARE LM RR RANGE AND
RANGE RATE WITH EMS
AND V83 RANGE AND RANGE
RATE
PRO

MOVE TO LEB

PERFORM OPTICS CHECKS

ZERO OPTICS-OFF
ZERO OPTICS-ZERO(15 SEC)
ZERO OPTICS-OFF
MONITOR LM IN SCT AND SXT
OPTICS MODE-MAN

(98+50)

SUNDOWN

(98+55)

REALIGN IMU TO REFSMMAT

P52 ADJUST RETICLE BRTNSS
KEY V37E52E
E 04 06 (ALIGN OPTION CODE)
LOAD 00003 IN R2 FOR
REALIGN TO REFSMMAT
PRO
F 50 25 (00015, PERFORM STAR ACQ)
OMC-MANEUVER SCT TO ACQ
TWO SUITABLE STARS
(ALTAIR#40,PEACOCK#42)
PRO
F 01 70 (STAR CODE)
CHECK FIRST STAR CODE
ZERO OPTICS-ZERO(15 SEC)
ZERO OPTICS-OFF

OPTICS MODE=CMC
PRO
06 92 (SHAFT,TRUN,BLANK)
MONITOR OPT DRIVE TO STAR ONE
IDENTIFY STAR ONE
OPTICS MODE=MAN
F 51 BB (PLEASE MARK)
CENTER FIRST STAR IN SXT
MARK ON STAR ONE
F 50 25 (00016, TERMINATE MARK SEQ)
PRO
F 01 71 (MARKED STAR CODE)
PRO
F 01 70 (STAR CODE)
CHECK SECOND STAR CODE
ZERO OPTICS=ZERO (15SEC)
ZERO OPTICS=OFF
OPTICS MODE=CMC
PRO
06 92 (SHAFT,TRUN,BLANK)
MONITOR OPT DRIVE TO STAR TWO
IDENTIFY STAR TWO
OPTICS MODE=MAN
F 51 BB (PLEASE MARK)
CENTER SECOND STAR IN SXT
MARK ON STAR TWO
F 50 25 (00016, TERMINATE MARK SEQ)
PRO
F 01 71 (MARKED STAR CODE)
PRO
F 06 05 (ANGLE DIFF)
COPY DATA ON CHECKLIST
PRO

E 06 93 (GYRO TORQ ANGLES)
COPY DATA ON CHECKLIST
PRO
F 50 25 (00014, PERFORM FINE ALIGN)
PRO
F 50 25 (00015,PERFORM STAR ACQ)
PRO
F 01 70 (STAR CODE)
LOAD THIRD STAR CODE
(RASALHAGUE=35)

ZERO OPTICS=ZERO (15SEC)
ZERO OPTICS=OFF
OPTICS MODE=CMC
PRO
06 92 (SHAFT,TRUN,BLANK)
MONITOR OPTICS DRIVE TO STAR
THREE
ZERO OPTICS=ZERO

KEY V37E00E
SET LEB ET COUNTING DOWN
TO LM DOI TIGN

(99+03)
MOVE TO CMD SEAT
ALIGN GDC TO IMU
KEY V16N20E
16 20 (R,P,Y)
ATT SET THUMBWHEELS TO N20
FDAT SELECT=1

NULL ATT ERROR NEEDLES ON FDAI 1 WITH ATT SET THUMBWHEELS FDAI SELECT-1/2 ATT SET-GDC DEPRESS GDC ALIGN PB. ATT SET-IMU KEY RELEASE	=9 =6	(99+25) (99+28)	LOSS OF SIGNAL **MANEUVER TO SXT TRACK ATITUDE**
 ORDEAL VERIFICATION			
KEY V83E F 16 54 (R,RDOT,THETA) SLEW/ADJUST FDAI TO THETA PRO			P20 F 50 18 (COMMANDER R,P,Y) PRO 06 18 (COMMANDER R,P,Y) MOVE TO LEB DURING AUTO MNVR F 50 18 (COMMANDER R,P,Y) (0,206/332,0) KEY ENTER
			ZERO OPTICS-OFF OPTICS MODE-CMC
-29 (99+05) **MCC-H UPLINK(CSM AND LM VECTOR)**	=4	(99+30)	
UP TLM(CM) (MDC)=ACCEPT MONITOR UPLINK ACT LT=ON MONITOR GND UPLINK MONITOR UPLINK ACT LT=OFF UP TLM(CM) (MDC)=BLOCK CONFIRM COMP ACT LT=OFF SET MDC ET COUNTING DOWN TO LM DOI TIGN	=2	(99+32) P76	VERIFY VHF AM A-SIMPLEX VERIFY PCM BIT RATE=LOW VERIFY TAPE RCDR FWD=FWD KEY V37E76E F 06 84 (DV,S OF LM DOI BURN) LOAD (-71,1,0,-0,3) PRO F 06 33 (GETI OF DOI) LOAD LM DOI TIGN +20 SEC (99+34+19)
(99+14) VHF AM A-SIMPLEX VHF AM B-OFF VHF RCV ONLY-B DATA VHF RANGING-OFF		0+00 (99+33+59)	***** LM DOI BURN (-71,1,0,0,-0,3) *****

	MOVE TO CMD SEAT	-63	(99+44)	
(99+36)	VHF AM A=OFF VHF AM B=DUPLEX VHF RCV ONLY=OFF VHF RANGING=RNG VHF RANGING=RESET(NO VOICE 12 SEC) COMPUTE RDOT FROM VHF RANGE			INITIATE LM TRACK SET LEB ET COUNTING DOWN TO LM PHASING TIGN ZERO OPTICS=ZERO CEASE TRACKING MOVE TO CMD SEAT
	CONFIRM LM BURN COMPLETE	-39	(100+07)	
	PRO/INCORPORATE P76			FDAI SCALE=5/5 RHC=ARMED ROLL 180 DEG AND PITCH UP 70 DEG AT 1 DEG/SEC (180,278/268,0)
F 37 BB				FDAI SCALE=5/1 RHC=LOCKED
	KEY V82E			KEY V37E00E
F 04 06 (VEHICLE OPTION CODES)	LOAD R2=00002			KEY V64E
	PRO	P00		(RHO,GAMMA,BLANK)
F 16 44 (HA,HP,TFF)	PRO		F 06 51	SLEW HI GAIN ANT
F 37 BB				PRO
(99+37)	SUNUP		(100+12)	ACQUISITION OF SIGNAL
(99+38)	**MANEUVER TO TRACK ATTITUDE**			ACQUIRE MSFN
P20	KEY 20E		(100+20)	OBTAI GO/NO GO FOR PDI ABORT
	MOVE TO LEB			MOVE TO LEB
(99+41)	ZERO OPTICS=ZERO(15 SEC)			KEY V67E
	ZERO OPTICS=OFF			F 06 99 (POS,VEL,ERR,OPTION CODE)

LOAD WI

+002.89
+0017.3
+00001

PRO

(100+25)

MANEUVER TO TRACK ATTITUDE

0+00 (100+46+21)

*****LM PHASING BURN*****

LM PHASING BURN
(170.4,0,-95.6)
(180,307/193,0)

(100+47)

CONFIRM LM BURN COMPLETE

P76

KEY V37E76E
F 06 84 (DV,S OF LM PHASING BURN)
LOAD(170.4,0,-95.6)

PRO

F 06 33 (GETI OF PHASING BURN)
LOAD LM PHASING TIGN
+32 SEC
(100+46+53)

PRO

F 37 BB

KEY V82E
F 04 06 (VEHICLE OPTION CODES)
LOAD R2=00002

PRO

F 16 44 (HA,HP,TFF)

PRO

F 37 BB

-9

(100+37)

PRO/PROCESS LAST MARK

OPTICS MODE=CMC

KEY V88E/TERMINATE VHF RNG

* PROCEDURES FOR *
* PHASING THRU INSERTION *

(100+48) **MANEUVER TO TRACK ATTITUDE**

P20 KEY 20E

(100+49) SUNDOWN

(100+51) KEY V87E
KEY V57E

F 51 BB (PLEASE MARK)
OPTICS MODE=MAN
OHC-CENTER LM IN SXT
MAKE 4 MARKS IN NEXT
3 MINUTES

(100+54) KEY V93E (REINITIALIZE W MAT)
MAKE 6 MARKS IN NEXT
7 MINUTES

(101+01) PRO/PROCESS LAST MARK
OPTICS MODE=CMC

(101+05) COPY INITIAL CSM BACKUP
INSERTION PAD

(101+10) COPY LM INSERTION P76 PAD

(101+14)

F 51 BB (PLEASE MARK)
OPTICS MODE=MAN
OHC-CENTER LM IN SXT
MAKE 10 MARKS IN NEXT
10 MINUTES

(101+24)

PRO/PROCESS LAST MARK
KEY V88E/TERMINATE VHF RNG
ZERO OPTICS-ZERO
OPTICS MODE=CMC

MOVE TO CMD SEAT

LOSS OF SIGNAL

(101+35)

SUNUP

(101+40)

ALIGN GDC TO IMU

16 20 (R,P,Y)
ATT SET THUMBWHEELS TO N20
FDI SELECT=1
NULL ATT ERROR NEEDLES
ON FDI 1 WITH ATT
SET THUMBWHEELS
FDI SELECT=1/2
ATT SET=GDC
DEPRESS GDC ALIGN PB
ATT SET=IMU

	ORDEAL VERIFICATION KEY V83E F 16 54 (R, RDOT, THETA) SLEW/ADJUST FDAI TO THETA PRO MOVE TO LEB	(102+11)	ACQUISITION OF SIGNAL PITCH DOWN 50 DEG (180, 194, 182, 0)
-58	(101+48)	-32 (102+14) P00	KEY V37E00E KEY V64E F 06 51 (RHO GAMMA, BLANK) SLEW HI GAIN ANT ACQUIRE MSFN PRO
-57	(101+49)		**MCC-H UPLINK CSM VECTOR** UP TLM(CM) (MDC)=ACCEPT MONITOR UPLINK ACT LT=ON MONITOR GND UPLINK MONITOR UPLINK ACT LT=OFF UP TLM(CM) (MDC)=BLOCK
-54	(101+52)		COPY LM INSERTION P76 PAD UPDATE AS REQUIRED COPY CSM BACKUP INSERTION PAD UPDATE AS REQUIRED
-37	(102+09)		SET MDC ET COUNTING DOWN TO INSERTION USING LM INSERTION TIGN +3 MIN AND MISSION TIMER
		(102+29)	**ORDEAL VERIFICATION** KEY V83E F 16 54 (R, RDOT, THETA) SLEW/ADJUST FDAI TO THETA PRO

-16 (102+30)
TARGET CSM INSERTION BACKUP

P30 KEY V37E30E
F 06 33 (GETI OF CSM INS BACKUP)
LOAD (102+46+18)
PRO
F 06 81 (VG - LV)
LOAD(175.4,0,-107.1)
PRO
F 06 42 (HA,HP,VG)
PRO
F 16 45 (MKS,TFI,MGA)
VERIFY ET=TFI
CONFIRM MGA LESS THAN 45 DEG
PRO

F 37 BB

*5

-14 (102+32)
MANEUVER TO INSERTION ATTITUDE

P40 KEY 40E
F 50 18 (COMMANDED R,P,Y)
PRO
06 18 (COMMANDED R,P,Y)
MONITOR ATT MNVR
F 50 18 (COMMANDED R,P,Y)
(180,3/272,0)

-9 (102+37)

SETUP SPS INSERTION BACKUP

EMS MODE=STBY
EMS FUNCTION=DV SET

LOAD INSERTION BURN VC
EMS FUNCTION=DV
FDATI SCALE=5/5
RATE-HIGH
RHC PWR DIRECT(BOTH)=MNA/MNB
BMAG MODE(3)=ATT1/RATE2
SCS TVC(BOTH)=RATE CMD
TVC GMBL DRIVE(BOTH)=AUTO
AUTO RCS SEL A/C ROLL(4)=MNA
RHC=ARMED

ALIGN S/C TO 180 DEG ROLL
PRO

06 18 (COMMANDED R,P,Y)
MONITOR ATT TRIM
F 50 18 (COMMANDED R,P,Y)

GDC ALIGN

ATT SET THUMBWHEELS TO N18
FDATI SELECT=1
NULL ATT ERROR NEEDLES
ON FDATI 1 WITH ATT
SET THUMBWHEELS
FDATI SELECT=1/2
ATT SET=GDC
GDC ALIGN PB=PUSH
ATT SET=IMU

KEY ENTER

F 50 25 (00204, GMBL DRIVE TEST)

*3 (102+43+18) *-00+05 F 99 40 (TFI, VG, DV)
MONITOR LM ENGINE IGNITION * PRO

LM INSERTION BURN *00+00 (102+46+18)
(-189.2, 0, 1, -83.9) * CSM BACKUP INSERTION BURN
***** * (175.4, 0, -107.1)
* (180.32, 272, 0)

* IF LM CANNOT PERFORM THE BURN *+00+01 06 40 (TFC, VG, DV)
(USE ZERO INSERTION CHECKLIST) * THC-TERMINATE ULLAGE
(FOR PARTIAL LM INSERTION GO TO * MONITOR SPS BURN
PARTIAL INSERTION CHECKLIST FOR * F 16 40 (TFC, VG, DV)
RESCUE 2 BURN AT 103+32) * DV THRUST A=OFF
***** * GMBL MTRS(4)=OFF/SEQUENTIALLY
MN BUS TIE(BOTH)=ON * PRO
NONESS BUS=MNA * F 16 85 (VG=BODY)
SPS HE VLV TB (BOTH)=BP * THC=NULL VGS
SPS HE VLV (BOTH)=AUTO * THC=LOCKED
VERIFY SPS TH LT=OFF * EMS MODE=STBY
GMBL MTRS(4)=ON(SEQUENTIALLY) * TVC SERVO PWR 1=AC1/MNA
TVC SERVO PWR 2=AC2/MNB * TVC SERVO PWR(BOTH)=OFF
KEY ENTER * SPS HE VLV (BOTH)=OFF
06 40 (TFI, VG, DV) * MN BUS TIE(BOTH)=OFF
DV THRUST A=NORMAL * NONESS BUS=OFF
THC=ARMED * PRO
*-00+35 DSKY BLANKS * F 37 BB
*-00+30 06 40 (TFI, VG, DV) (AVE G ON) * (CONTINUE DETAILED PROCEDURES
EMS MODE=NORMAL * BUT DELETE P76)

*-00+15 THC=APPLY ULLAGE *

-60-

CONFIRM LM BURN COMPLETE

P76 KEY V37E76E

F 06 84 (DV,S OF LM INSERTION BURN)

RHC=LOCKED

THC=LOCKED

AUTO RCS SEL, A/C ROLL (4)=OFF

EMS FUNCTION=VHF RNG

EMS MODE=VHF RNG

VHF RNG=RESET

FDAI SCALE=5/1

RATE=LOW

ROT CONT PWR DIRECT (BOTH)=OFF

BMAG MODE(3)=RATE2

TVC GIMBAL DRIVE(BOTH)=1

LOAD (=189.2,0,-83.9)

PRO

F 06 33 (GETI OF INSERTION BURN)

LOAD LM INSERTION TIGN

+8 SEC

(102+43+18)

PRO

F 37 BB

KEY V82E

F 04 06 (VEHICLE OPTION CODES)

LOAD R2=00002

PRO

F 16 44 (HA,HP,TFF)

PRO

F 37 BB

* PROCEDURES FOR *
* INSERTION THRU CSI *

(102⁴⁷)

SUNDOWN
MOVE TO LEB
SET LEB ET COUNTING DOWN TO
CSI USING LM CSI TIGN
AND MISSION TIMER

PS2 **REALIGN IMU TO REFSMMAT**
KEY 52E
ADJUST RETICLE BRTNESS
F 04 06 (ALIGN OPTION CODE)
LOAD 00003 IN R2 FOR
REALIGN TO REFSMMAT
PRO
F 50 25 (00015, PERFORM STAR ACQ)
OHC=MANEUVER SCT TO ACQ
TWO SUITABLE STARS
(ALTAIR=40, FOMALHAUT=45)
PRO
F 01 70 (STAR CODE)
CHECK FIRST STAR CODE
ZERO OPTICS=OFF
PRO
06 92 (SHAFT,TRUN,BLANK)
MONITOR OPT DRIVE TO STAR ONE
IDENTIFY STAR ONE
OPTICS MODE=MAN

F 51 88 (PLEASE MARK)
CENTER FIRST STAR IN SXT
MARK ON STAR ONE
F 50 25 (00016, TERMINATE MARK SEQ)
PRO
F 01 71 (MARKED STAR CODE)
PRO
F 01 70 (STAR CODE)
CHECK SECOND STAR CODE:
ZERO OPTICS=ZERO (15SEC)
ZERO OPTICS=OFF
OPTICS MODE=CMC
PRO
06 92 (SHAFT,TRUN,BLANK)
MONITOR OPT DRIVE TO STAR TWO
IDENTIFY STAR TWO
OPTICS MODE=MAN
F 51 88 (PLEASE MARK)
CENTER SECOND STAR IN SXT
MARK ON STAR TWO
F 50 25 (00016, TERMINATE MARK SEQ)
PRO
F 01 71 (MARKED STAR CODE)
PRO
F 06 05 (ANGLE DIFF)
COPY DATA ON CHECKLIST
PRO
F 06 93 (GYRO TORQ ANGLES)
COPY DATA ON CHECKLIST
PRO
F 50 25 (00014, PERFORM FINE ALIGN)
PRO
F 50 25 (00015, PERFORM STAR ACQ)
PRO

F 01 70 (STAR CODE) LOAD THIRD STAR CODE (RASALHAGUE=35)	(102+59)	KEY ENTER ZERO OPTICS=OFF COPY LM CSI AND TPI TIGNS
ZERO OPTICS=ZERO (15SEC) ZERO OPTICS=OFF OPTICS MODE=CMC PRO	P32	KEY V37E32E
06 92 (SHAFT,TRUN,BLANK) MONITOR OPTICS DRIVE TO STAR THREE ZERO OPTICS=ZERO	F 06 11	(GETI=CSI) LOAD LM CSI TIGN (103+33+46)
(102+52) P00 KEY V37E00E	F 06 55	PRO (N,E,CENTANG) LOAD R2=+208.30 LOAD R3=130.00 PRO
LM STATE VECTOR UPLINK UP TLM(CM) (MDC)=ACCEPT MONITOR UPLINK ACT LT=ON MONITOR GND UPLINK MONITOR UPLINK ACT LT=OFF UP TLM(CM) (MDC)=BLOCK	F 06 37	(GETI=TPI) LOAD LM TPI TIGN (105+09+00) PRO
(102+56) **MANEUVER TO TRACK ATTITUDE**	F 16 45	(MKS,TFI=CSI,-00001) SET LEB ET=TFI KEY V32E
P20 KEY V37E20E F 50 18 (COMMANDED R,P,Y) PRO 06 18 (COMMANDED R,P,Y) MONITOR AUTO MNVR F 50 18 (COMMANDED R,P,Y) (180,162/13,0)	F 06 75 F 06 81 F 06 82 F 16 45	(DH,DT=CSI/CDH,DT=CDH/TPI) PRO (CSI VG=LV) (CDH VG=LV) (MKS,TFI=CSI,-00001)
	-32 (103+02)	KEY V93/REINITIALIZE W MATRIX KEY V87E (VHF RANGING) KEY V57E
	F 51 B8	(PLEASE MARK) OPTICS MODE=MAN OHC=CENTER LM IN SXT

	MAKE 6 MARKS IN NEXT 5 MINUTES	(103+23)	LOSS OF SIGNAL
*27	(103+07)	-9	(103+25)
	PRO/PROCESS LAST MARK		PRO/MAKE FINAL PASS
	F 16 45 (MKS,TFI=CSI,-00001)	F 06 75 (DH,DT=CSI/CDH,DT=CDH/TPI)	COPY DATA ON CHECKLIST
	ZERO OPTICS=ZERO	PRO	
	OPTICS MODE=CMC	F 06 81 (CSI VG=LV)	OVERWRITE N81 VG WITH
	MOVE TO CMD SEAT	(-) CSM YDOT	COPY DATA ON CHECKLIST
*22	(103+12)	PRO	
	KEY V90E	F 06 82 (CDH VG=LV)	COPY DATA ON CHECKLIST
	F 04 12 (VEHICLE OPTION)	PRO	
	LOAD R2=00002	F 16 45 (MKS,TFI=CSI,MGA)	SET MDC ET=TFI
	PRO	PRO	
	F 06 16 (TIME OF EVENT)	F 37 88	
	LOAD LM CSI TIGN	(103+28)	COPY LM CSI PAD (P76)
	(103+33+46)	(103+29)	
	PRO	P40	KEY 40E
	F 06 90 (Y,YDOT,PSI)	F 50 18 (COMMANDER R,P,Y)	
	VOICE LM YDOT TO LM	KEY V56E	
	PRO	PRO	
	F 16 45 (MKS,TFI=CSI,-00001)	06 18 (COMMANDER R,P,Y)	MONITOR MANEUVER
	KEY V90E	F 50 18 (COMMANDER R,P,Y)	(180,166/277,0)
	F 04 12 (VEHICLE OPTION)	-4	(103+30)
	LOAD R2=00001		**SETUP SPS CSI BACKUP**
	PRO		
	F 06 16 (TIME OF EVENT)		EMS MODE=STBY
	LOAD CSM CSI TIGN		EMS FUNCTION=DV SET
	(103+33+46)		LOAD CSI BURN VC
	PRO		
	F 06 90 (Y,YDOT,PSI)		
	COPY (-) CSM YDOT		
	ON CHECKLIST		
	F 16 45 (MKS,TFI=CSI,-00001)		

-64-

EMS FUNCTION=DV
FDI AI SCALE=5/5
RATE=HIGH
RHC PWR DIRECT(BOTH)=MNA/MNB
BMAG MODE(3)=ATT 1/RATE 2
TVC GMBL DRIVE(BOTH)=AUTO
AUTO RCS SEL A/C ROLL (4)=MNA
RHC=ARMED

ALIGN S/C TO 180 DEG ROLL

PRO

06 18 (COMMANDER R,P,Y)

MONITOR ATT TRIM

F 50 18 (COMMANDER R,P,Y)

*3

GDC ALIGN

ATT SET THUMBWHEELS TO N18

FDI AI SELECT=1

NULL ATT ERROR NEEDLES

ON FDI AI 1 WITH ATT

SET THUMBWHEELS

FDI AI SELECT=1/2

ATT SET=GDC

GDC ALIGN PB=PUSH

ATT SET=IMU

KEY ENTER

F 50 25 (00204, GMBL DRIVE TEST)
(103+33)

SUNUP

*2

KEY ENTER

06 40 (TFI, VG, DV)

DV THRUST A=NORMAL

THC=ARMED

-00+35

DSKY BLANKS

-00+30

06 40 (TFI, VG, DV) (AVE G ON)

EMS MODE=NORMAL

-00+05

F 99 40 (TFI, VG, DV)

00+00 (103+33+46)

MONITOR LM ENGINE IGNITION

*****LM CSI BURN

(50.5,0,0)

IF LM CANNOT PERFORM THE BURN

MN BUS TIE (2)=ON(UP)

NONESS BUS=MNA

SPS HE VLV TB (BOTH)=BP

SPS HE VLV (BOTH)=AUTO

VERIFY SPS TH LT=OFF

THC=APPLY ULLAGE

GMBL MTRS(4)=ON(SEQUENTIALLY)

TVC SERVO PWR 1=AC1/MNA

TVC SERVO PWR 2=AC2/MNB

PRO

**0001	06 40 (TFC, VG, DV) THC-TERMINATE ULLAGE MONITOR SPS BURN	*	ROT CONT PWR DIRECT (BOTH)=OFF BMAG MODE(3)=RATE 2 TVC GIMBAL DRIVE(BOTH)=1
F 16 40	(TFC, VG, DV) DV THRUST A=OFF GMBL MTRS(4)=OFF/SEQUENTIALLY PRO	*	(103+37) LOAD(50,5,0,0) PRO
F 16 85	(VG=BODY) THC=NULL VGS THC=LOCKED TVC SERVO PWR(BOTH)=OFF EMS MODE=STBY SPS HE VLV (BOTH)=OFF MN BUS TIE (BOTH)=OFF NONESS BUS=OFF PRO	*	F 06 33 (GETI OF CSI BURN) LOAD LM CSI TIGN +16 SEC (103+33+46) PRO
F 37 BB	(CONTINUE DETAILED PROCEDURES* BUT DELETE P76)	*	F 37 BB KEY V82E F 04 06 (VEHICLE OPTION CODES) LOAD R2=00002 PRO
		*	F 16 44 (HA,HP,TFF) PRO
		*	F 37 BB

CONFIRM LM BURN COMPLETE

P76 KEY V37E76E
F 06 84 (DV,S OF CSI BURN)

RHC=LOCKED
THC=LOCKED
AUTO RCS SEL, A/C ROLL. (4)=OFF
EMS FUNCTION=VHF RNG
EMS MODE=VHF RNG
VHF RNG=RESET
FD/AI SCALE=5/1
RATE=LOW

* PROCEDURES FOR *
* CSI THRU CDH *

(103+38) **MANEUVER TO TRACK ATTITUDE**

P20 KEY 20E
F 50 18 (COMMANDER R,P,Y)
PRO
06 18 (COMMANDER R,P,Y)
MOVE TO LER DURING
AUTO MANEUVER
F 50 18 (COMMANDER R,P,Y)
(180,155/239,0)
KEY ENTER
SET LEB ET COUNTING DOWN TO
CDH USING LM CDH TIGN
AND MISSION TIMER
ZERO OPTICS=OFF

(103+41) KEY V87E (VHF TRACKING)
KEY V57E

F 51 BB (PLEASE MARK)
OPTICS MODE=MAN
OHC=CENTER LM IN SXT
MAKE 3 MARKS IN NEXT
2 MINUTES

(103+43) PRO/PROCESS LAST MARK

KEY V67E
F 06 99 (POS ERR,VEL ERR,OPTION CODE)
LOAD WR
(000.57,0003.4,00001)
PRO
(103+44) KEY V57E
F 51 BB (PLEASE MARK)
MAKE 11 MARKS IN NEXT
11 MINUTES
(103+55) PRO/PROCESS LAST MARK
OPTICS MODE=CMC
OBTAIN LM PC TIGN
KEY V90E
F 04 12 (VEHICLE OPTION)
LOAD R2=00002
PRO
F 05 16 (TIME OF EVENT)
LOAD LM PC TIGN
(104+02+00)
PRO
F 06 90 (Y, YDOT, PSI)
VOICE LM YDOT TO LM
PRO
(103+57) COPY LM PC PAD
KEY V57E
F 51 BB (PLEASE MARK)
OPTICS MODE=MAN

	OHG-CENTER LM IN SXT MAKE 5 MARKS IN NEXT 5 MINUTES		F 06 81 (CDH VG-LV) PRO
-30	(104+02+00)		F 16 45 (MKS,TFI=CDH,-00001) VERIFY ET=TFI
	PRO/PROCESS LAST MARK OPTICS MODE=CMC	-24 (104+08)	
	***** LM PLANE CHANGE(0,0,0) *****		KEY V87E (VHF RANGING) KEY V57E
P76	KEY V37E76E		OPTICS MODE=MAN
	F 06 84 (DV,S OF LM PC BURN) LOAD LM PLANE CHANGE DV,S PRO		OHG-CENTER LM IN SXT TAKE 4 MARKS IN NEXT 3 MINUTES
	F 06 33 (GETI OF PC BURN) LOAD GETI=PC BURN (104+02+00)		KEY V93E (REINITIALIZE W MAT) TAKE 8 MARKS IN NEXT 9 MINUTES
	CONFIRM LM PC BURN		PRO / PROCESS LAST MARK (MKS,TFI,-00001)
	PRO/INCORPORATE P76	(104+09)	ZERO OPTICS-ZERO OPTICS MODE=CMC
	F 37 BB (104+05)	-12 (104+20)	ACQUISITION OF SIGNAL
P33	KEY 33E		KEY V90E
	F 06 13 (GETI=CDH) OBTAIN LM CDH TIGN LOAD LM CDH TIGN (104+31+42)		F 04 12 (VEHICLE OPTION) LOAD R2=00002 PRO
	COPY DATA ON CHECKLIST PRO		F 06 16 (TIME OF EVENT) LOAD LM CDH TIGN (104+31+42) PRO
	F 16 45 (MKS,TFI=CDH,-00001) KEY V32E		F 06 90 (Y,YDOT,PSI) VOICE LM YDOT TO LM PRO
	F 06 75 (DH,DT,CDH/TPI,DT-TPI/TPI) PRO		F 16 45 (MKS,TFI=CDH,-00001)

	MOVE TO CMD SEAT	(104+28)	
(104+22)	FDAI SCALE=5/5	-3	COPY LM CDH P76 PAD
	RHC=ARMED		
	ROLL 180 DEG AT 2 DEG/SEC		
	(0,167/108,0)		
	RHC-LOCKED		
	FDAI SCALE=5/1.		
	ACQUIRE MSFN		
	(RHO=50, GAMMA=183)		
		-2	(104+29)
			P41
			KEY 41E
			F 50 18 (COMMANDED R,P,Y)
			KEY V56E
			KEY ENTER(BYPASS MNVR)
			06 85 (VG=BODY)
~7	(104+25)		(104+30)
	PRO/MAKE FINAL PASS		**SETUP RCS CDH BACKUP**
	F 06 75 (DH=DT=CDH/TPI,DT=TPi/TPI)		
	COPY DATA ON CHECKLIST		
	PRO		
	F 06 81 (CDH VG=LV)		
	KEY V90E		
	F 04 12 (VEHICLE OPTION)		
	LOAD R2=00001		
	PRO		
	F 06 16 (TIME OF EVENT)		
	LOAD LM CDH TIGN		
	(104+31+42)		
	PRO		
	F 06 90 (Y,YDOT,PSI)		
	COPY (-) CSM YDOT ON CHECKLIST		
	PRO		
	F 06 81 (CDH VG=LV).		
	OVER WRITE N1 WITH		
	(-) CSM YDOT		
	COPY DATA ON CHECKLIST		
	PRO		
	F 16 45 (MKS,TFI=CDH,MGA)		
	SET MDC ET=TFI		
	PRO		
	F 37 88		
		-00+35	
		-00+30	
			(104+31)
			ALIGN GDC TO IMU
			16 20 (R,P,Y)
			ATT SET THUMBWHEELS TO N20
			FDAI SELECT=1
			NULL ATT ERROR NEEDLES
			ON FDAI 1 WITH ATT
			SET THUMBWHEELS
			FDAI SELECT=1/2
			ATT SET=GDC
			DEPRESS GDC ALIGN PB
			ATT SET=IMU
			KEY RELEASE
			DSKY BLANKS
			16 85 (VG=BODY) (AVE G ON)
			THC=ARMED
			RHC=ARMED
			EMS MODE=NORMAL

0+00 (104+31+42)
F 16 85 (VG=BODY)
MONITOR LM IGNITION

LM CDH BURN (-0.8,0,3,3)

* IF LM CANNOT PERFORM THE BURN *
* ***** *
* CSM BACKUP CDH BURN *
* (-.6,0,-5,0) *
* (0,188/108,0) *
* ***** *
* THC=NULL: VGS *
* THC=LOCKED *
* PRO *
* F 37 BB *
* (CONTINUE DETAILED PROCEDURES*
* BUT DELETE P76 *

CONFIRM LM BURN COMPLETE
PRO (TO BYPASS CSM=CDH BURN)
F 37 BB
RHC=LOCKED
AUTO RCS SEL A/C ROLL (4)=OFF
BMAG MODE (3)=RATE 2

P76 KEY 76E
F 06 84 (DV,S OF LM CDH BURN)
LOAD LM CDH DV,S
LOAD (-0.5,0,+5.8)
PRO
F 06 33 (GETI OF CDH BURN)
LOAD LM CDH GETI
(104+31+42)
PRO
F 37 BB

* PROCEDURES FOR *
* CDH, THRU TPI *

-35 (104+34)

TARGET CSM TPI BACKUP
P20 KEY 20E
F 50 18 (COMMANDED R,P,Y)
PRO
06 18 (COMMANDED R,P,Y)
MOVE TO LEB DURING
AUTO MANEUVER
F 50 18 (COMMANDED R,P,Y)
(0,229/137+0)
KEY ENTER

P34 KEY V37E34E
F 06 37 (GETI=TPI)
LOAD TPI, TIGN(105+09+00)
PRO
F 06 55 (BLANK,ELANG,CENTANG)
LOAD R2 = +208,30 (DEG)
LOAD R3 = +130.00 (DEG)
PRO
F 16 45 (MKS,TFI=00001)
SET LEB ET TO TFI
COUNTING DOWN

-31 (104+38)
ZERO OPTICS=OFF
KEY V87E (VHF RANGING)
KEY V57E
F 51 BB (PLEASE MARK)
KEY V16N45E
F 16 45 (MKS,TFI=00001)

F 51 BB (PLEASE MARK)
OPTICS MODE=MAN
OHC-CENTER LM IN SXT
MAKE 4 MARKS IN NEXT
3 MINUTES (INCORPORATE 4)
OPTICS MODE=CMC

KEY V32E
F 06 37 (GETI=TPI)
COPY DATA ON CHECKLIST
PRO
F 06 58 (HP,DVTPI,DVTPF)
COPY DATA ON CHECKLIST
PRO
F 06 59 (VG=LOS)
COPY DATA ON CHECKLIST
PRO
F 16 45 (MKS,TFI=00001)

KEY V93E (REINITIALIZE W MAT)
KEY V57E
F 51 BB (PLEASE MARK)
OPTICS MODE=MANUAL
OHC-CENTER LM IN SXT
TAKE 14 MARKS IN NEXT
14 MINUTES

(104+45)
SUNDOWN

-19 (104+50)
COPY LM TPI TIGN ON CHECKLIST

-13 (104+56)
PRO/PROCESS LAST MARK
F 16 45 (MKS,TFI=00001)
ZERO OPTICS=ZERO
OPTICS MODE=CMC

MOVE TO CMD SEAT
VERIFY ORDEAL
KEY V83E
F 16 54 (R, RDOT, THETA)
SLEW/ADJUST FDAI TO THETA
PRO
F 16 45 (MKS, TFI, =00001)
-11 (104+58)
KEY V37E34E
F 06 37 (TPI TIGN)
LOAD LM TPI TIGN
(105+09+00)
PRO
F 06 55 (BLNK, ELANG, CENTANG)
LOAD R2 = +000.00
PRO
F 16 45 (MKS, TFI, =00001)
PRO/MAKE FINAL PASS
F 06 55 (BLNK, ELANG, CENTANG)
COPY DATA ON CHECKLIST
PRO
F 06 58 (HP, DVTP1, DVTPF)
COPY DATA ON CHECKLIST
PRO
F 06 81 (VG-LV)
COPY DATA ON CHECKLIST
PRO
F 06 59 (VG-LOS)
COPY DATA ON CHECKLIST
PRO
F 16 45 (MKS, TFI, MGA)
RESET MDC ET WITH TFI
PRO

F 37 88
-7 (105+02)
MANEUVER TO TPI BACKUP ATTITUDE
P40
KEY 40E
F 50 18 (COMMANDER R,P,Y)
KEY V56E:
PRO
06 18 (COMMANDER R,P,Y)
MONITOR ATT MNVR
F 50 18 (COMMANDER R,P,Y)
(0, 192/15, 0)
COPY LM TPI PAD (P76)
-5 (105+04)
SETUP SPS TPI BACKUP
EMS MODE=STBY
EMS FUNCTION=DV SET
LOAD INSERTION BURN VC
EMS FUNCTION=DV
FDATI SCALE=5/5
RATE=HIGH
RMC PWR DIRECT (BOTH)=MNA/MNB
BMAG MODE (3)=ATT1/RATE2
TVC GMBL DRIVE (BOTH)=AUTO
AUTO RCS SEL A/C ROLL (4)=MNA
RMC ARMED
ALIGN S/C TO ZERO ROLL
PRO
06 18 (COMMANDER R,P,Y)
MONITOR ATT TRIM
F 50 18 (COMMANDER R,P,Y)

2

GDC ALIGN

ATT SET THUMBWHEELS TO N18
FDAAI SELECT=1.
NULL ATT ERROR NEEDLES
ON FDAAI 1 WITH ATT
SET THUMBWHEELS
FDAAI SELECT=1/2
ATT SET=GDC
DEPRESS GDC ALIGN PB
ATT SET=IMU

F 50 25 (00204, GMBL DRIVE TEST)
KEY ENTER
06 40 (TFI, VG, DV)
DV THRUST A=NORMAL
THC=ARMED

-00+35

-00 +30

06 40 (TFI, VG, DV) (AVE G ON)
EMS MODE=NORMAL

-00♦05

F 99 40 (TFI, VG, DV)

(105*09*00)

MONITOR LM ENGINE IGNITION

LM TPI BURN(22.0,0-12.1)

2

IF LM CANNOT PERFORM THE BURN

MN RUS TIE (2)=ON (UP)
NONESS BUS-MNA
SPS HE VLV (BOTH)=AUTO
SPS HE VLV TB (BOTH)=BP
VERIFY SPS TH LT=OFF

THC=APPLY ULLAGE
GMBL MTRS(4)=ON(SEQUENTIAL)
TVC SERVO PWR 1=AC1/MNA
TVC SERVO PWR 2=AC2/MNB
PRO

CSM TPI BACKUP BURN
(*21.8:0,11.6)
(09208/15.0)

06 40 (TFC, VG, DV)
THC-TERMINATE ULLAGE
MONITOR SPS BURN

F. 16-40 (TFC, VG, DV)
DV THRUST A-OFF
GMBL MTRS(4)-OFF/SEQUENTIALLY

PRO
E 16 BE (VA-BODA)

TVG=0.0001
THC=NULL VGS
THC=LOCKED
EMS MODE=STBY
TVC SERVO PWR (BOTH)=OFF
SPS HE VLV (BOTH)=OFF
MN BUS TIE (BOTH)=OFF

NONESS BUSES-OFF

PRO

F 37 BB

(CONTINUE DETAIL PROCEDURES)

BUT DELETE P76)

CONFIRM LM BURN COMPLETE

P76 KEY V37E76E

F 06 84 (DV,S OF LM TPI BURN)

RHC-LOCKED

THC-LOCKED

FDI SCALE=5/1

RATE=LOW

ROT CONT PWR DIRECT (BOTH)=OFF

BMAG MODE (3)=RATE 2

TVC GIMBAL DRIVE (BOTH)=OFF

AUTO RCS SEL A/C ROLL (4)=OFF

EMS FUNCTION=VHF RNG

EMS MODE=VHF RNG

VHF RNG=RESET

LOAD (22.2,0-12.1)

PRO

F 06 33 (GETI OF TPI BURN)

LOAD LM GET=TPI

+8 SEC

(105+09+08)

PRO

F 37 BB

* PROCEDURES FOR *
* TPI THRU TPF *
***** +11 (105+20) OHG-CENTER LM IN SXT
MAKE 8 MARKS IN NEXT
8 MINUTES

*2 **MANEUVER TO SXT TRACK**
P20 KEY 20E
F 50 18 (COMMENDED R,P,Y) MOVE TO COMMAND SEAT
PRO
06 18 (COMMENDED R,P,Y) +12+00 (TIGN=3.0 MIN) LOSS OF SIGNAL
MOVE TO LER DURING AUTO
MANEUVER
F 50 18 (COMMENDED R,P,Y)
(0,247/48,0)
KEY ENTER
TARGET CSM MCC1 BACKUP
P35 KEY V37E35E
F 16 45 (MKS,TFI,-00001)
(TIME FROM TPI)
SET LEB ET TO TFI,
COUNTING UP
ZERO OPTICS-OFF
F 16 45 (MKS,TFI,MGA)
PRO
F 37 88
COPY LM MCC1 PAD (P76)
*3 KEY V93E (REINITIALIZE W MAT) +13+30
KEY V87E (VHF RANGING) P41
KEY V57E
F 51 88 (PLEASE MARK)
KEY V16N45E
F 16 45 (MKS,TFI,-00001)
OPTICS MODE-MAN
F 50 18 (COMMENDED R,P,Y)
KEY ENTER (BYPASS MNVR)
06 85 (VG-BODY)
SET UP MCC1 BACKUP
DSKY BLANKS

+14+25

*14*30

16 85 (VG=BODY) (AVE G ON)
THC-ARMED
RHC-ARMED

*15 (105*24*00)

F 16 85 (VG=BODY)

LM PERFORMS MCC1 BURN

CSM MCC1 BACKUP BURN
(0,265/29,0)

CONFIRM LM BURN COMPLETE

F 37 BB PRO (TO BYPASS CSM-MCC1 BURN)

THC-LOCKED
RHC-LOCKED

P76

F 06 84 (DV,S OF LM MCC1 BURN)
LOAD LM MCC1 DV,S
PRO

F 06 33 (GETI OF MCC1 BURN)
LOAD LM GET-MCC1
(105*24*00)

PRO

F 37 BB MOVE TO LEB

+17

P35

TARGET CSM MCC2 BACKUP
KEY 35E
F 16 45 (MKS,TFI,-00001)
ZERO OPTICS=OFF
KEY V93E (REINITIALIZE W MAT)
KEY V87E (VHF RANGING)
KEY V57E
F 51 BB (PLEASE MARK)
KEY V16N45E
F 16 45 (MKS,TFI,-00001)
OMC-CENTER LM IN SXT
OPTICS MODE=MAN
MAKE 9 MARKS IN NEXT
9 MINUTES

(105*32)

SUNUP

PRO/PROCESS LAST MARK
F 16 45 (MARKS,TFI,-00001)
ZERO OPTICS=ZERO

MOVE TO COMMAND SEAT

+26

(TIGN=3.0 MIN)

PRO/MAKE FINAL PASS
F 06 81 (VG-LV)
COPY DATA ON CHECKLIST
PRO

F 06 59 (VG-LOS)
COPY DATA ON CHECKLIST
PRO

F 16 45 (MARKS,TFI,MGA)
PRO

F 37 BB

COPY LM MCC2 PAD
COMPARE CSM SOLN WITH LM
SOLUTION

*28+30
P41
KEY 41E
F 50 18 (COMMANDED R,P,Y)
KEY V56E
KEY ENTER(BYPASS MNVR,
06 85 (VG-BODY)

SET UP MCC2 BURN

*29+25
DSKY BLANKS

*29+30
16 85 (VG-BODY) (AVE G ON)
THC-ARMED
RHC-ARMED

*30+00 (105+39+00)
F 16 85 (VG-BODY)

LM PERFORMS MCC2 BURN

CSM MCC2 BACKUP BURN
(0,302/20,0)

CONFIRM LM BURN COMPLETE

PRO (TO BYPASS CSM-MCC2 BURN)
THC-LOCKED
RHC-LOCKED

F 37 BB

P76
KEY 76E
F 06 84 (DV,S OF LM MCC2 BURN)
LOAD LM MCC2 DV,S
PRO
F 06 33 (GETI OF MCC2 BURN)
LOAD LM GET=MCC2
(105+39+00)
PRO
F 37 BB

*31
MANEUVER TO COAS TRACK ATTITUDE

P00
KEY 00E
KEY V89E
F 04 06 (00003,00001,BLNK)
LOAD R2 = 00002
PRO
F 06 18 (COMMANDED R,P,Y)
PRO
F 50 18 (COMMANDED R,P,Y)
PRO
06 18 (COMMANDED R,P,Y)
MONITOR ATT NMVR
F 50 18 (COMMANDED R,P,Y)
ALIGN S/C TO ZERO DEG ROLL
PRO
06 18 (COMMANDED R,P,Y)
F 50 18 (COMMANDED R,P,Y)
(0,274/346,0)
KEY ENTER

* PROCEDURES FOR BRAKING *

*34

RHC-ARMED
CENTER LM IN RETICLE
BMAG MODE(3)-ATT/RATE 2
MONITOR EMS FOR RANGE

*39

P47 KEY V37E47E

F 16 83 (DV-BODY)

KEY V83E

F 16 54 (R,RDOT,THETA)

THC-ARMED

MONITOR LOS CONTROL

MONITOR R AND R DOT

* BRAKING GATES AND RET ANG,S *

*
*
* 30 FPS AT 6000 FT=-.13 DEG *

* 20 FPS AT 3000 FT=-.26 DEG *

* 10 FPS AT 1500 FT=-.54 DEG *

* 5 FPS AT 500 FT=-1.6 DEG *

USE RANGE ON DSKY TO CHECK EMS
RANGE INDICATOR. USE
RETICLE ANGLE AS THIRD
VOTE.

*43 (105+52)

TPF(0,310/353,0)

6.3 P20 NAVIGATION SUMMARY WITH SUN ANGLES

IT - INITIATE TRACK
CT - CEASE TRACK
(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE
		(L.O.S. TO SUN) DEG
98:35	CSM SEP	
98:50	SUNSET	
99:34	LM DOI	
99:37	SUNRISE	
100:27	IT (SXT/VHF) (10/10)	207
100:32		186
100:37	CT (SXT/VHF)	174
100:46	LM PHASING	
100:49	SUNSET	
100:51	IT (SXT/VHF) (4/3,V93,6/7)	
101:01	CT (SXT)	
101:14	IT (SXT/VHF) (10/10)	
101:24	CT (SXT/VHF)	
101:35	SUNSET	
101:49	(T (SXT/VHF) (4/3,V93,16/17)	131
101:54		117
101:59		111
102:04		102
102:09	CT (SXT/VHF)	93
102:43	LM INSERTION	
102:47	SUNSET	
103:02	IT (SXT/VHF) (V93,6/5,V93,17/18)	
103:07	CT (SXT)	
103:25	CT (VHF)	
103:33	SUNRISE	
103:34	LM CSI	
103:41	IT (SXT/VHF) (3/3;V67,00057,00034,00001; 16/16)	186
103:46		171
103:51		156
103:55		144
103:57		138
104:02	CT	123
104:02	LM PLANE CHANGE	
104:08	IT (SXT/VHF) (4/3,V93,8/9)	105
104:14		87
104:20	CT (SXT/VHF)	70
104:32	LM CDH	
104:38	IT (SXT/VHF) (4/3,V93,14/15)	17
104:43		4
104:45	SUNSET	5
104:56	CT (SXT/VHF)	
105:09	LM TPI	
105:12	IT (SXT/VHF) (V93,8/8)	
105:20	CT (SXT/VHF)	
105:24	LM MCC1	
105:26	IT (SXT/VHF) (V93,9/9)	
105:32	SUNRISE	91
105:35	CT (SXT/VHF)	92
105:39	MCC2	
105:52	TPF	

6.4 CSM ATTITUDE SUMMARY
INERTIAL AND ORDEAL FDAI GIMBAL ANGLE
PROFILE FOR THE NOMINAL F MISSION
(YAW = 0.0)

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
97:40	180	282	14 (Yaw = 14.0)	
97:50	180	312	14	
98:10	180	12	14	
98:12	0	21	14	
98:35	0	90	14	
98:39	0	142	54	40
98:55	0	170	34	
99:28	0	206	332	55
99:39	0	227	317	
100:07	180	278	268	70
100:25	180	293	243	25
100:48	180	308	188	
101:14	180	337	138	
101:49	180	108	162	
102:11	180	174	162	50
102:36	180	3	272	110
102:46	180	32	272	
102:56	180	162	13	101
103:05	180	161	346	
103:29	180	166	277	11
103:34	180	180	277	
103:38	180	155	239	39
104:02	180	156	167	
104:05	180	157	158	
104:25	0	167	108	
104:32	0	188	108	
104:36	0	229	137	29
104:58	0	235	75	
105:02	0	192	15	53
105:09	0	208	15	
105:11	0	247	48	33
105:24	0	265	29	
105:39	0	302	20	
105:42	0	274	346	34
105:52	0	310	353	

7.0 DISCUSSION OF LM ABORT AND RESCUE CASES

7.1 LM Active PDI Abort

A LM active PDI abort rendezvous is planned if a failure occurs after DOI which precludes continuation of the nominal rendezvous mission. The LM PDI abort burn is targeted to occur at LM perilune one-half revolution after LM DOI. This burn provides phasing such that the LM TPI burn occurs one revolution earlier than the nominal TPI time. The LM CSI burn occurs one-half revolution after the PDI abort burn, and the LM CDH burn occurs one-half revolution after the CSI burn. The CDH burn is targeted to produce a 15 nautical mile delta height. The LM TPI burn will be cued at an elevation angle of 26.6 degrees and will occur near the midpoint of darkness.

Since the LM is active for this rendezvous, the function of the CSM will be to target the CSI, CDH, and TPI burns and backup the LM burns.

The onboard data for this case is valid if the CSM becomes active following the LM PDI abort burn.

7.2 Partial LM Phasing Greater Than or Equal to 40 FPS

This abort and rescue case was developed to provide a rendezvous capability assuming the LM could not complete the Phasing burn but could apply at least 40 FPS. This is the most likely partial Phasing situation since the LM RCS could be used to apply 40 FPS even if a failure of both the LM DPS and APS engines precluded application of the burn. The rendezvous sequence assuming a partial Phasing burn greater than or equal to 40 FPS requires the CSM to perform a rescue burn approximately one revolution after the LM DOI burn. This rescue burn is sized, based upon the magnitude of the partial Phasing burn, so that the terminal rendezvous approach is similar to that in the nominal case. A canned rescue burn (48.7 FPS retrograde)

will be provided onboard which is valid for the partial phasing equal to 40 FPS, the most likely failure situation. This burn will nominally be updated by the ground 20 minutes after the failed LM burn based upon the LM DV applied at phasing. Updating the Rescue burn preserves the nominal terminal rendezvous approach.

The CSM Rescue burn at 101:33 lowers the CSM perilune to about 20 nautical miles and provides phasing which results in TPI occurring near the nominal TPI time (105:09). The CSM CSI One burn is targeted so that the CDH burn occurs one revolution after the CSI One burn. A CSI Two burn (nominally zero) is scheduled half way between the CSI One and CDH burns. The CSM TPI burn at a delta height of 15 nautical miles will be cued on an elevation angle of 208.3 degrees which results in TPI at the midpoint of darkness. There exists the possibility that the LM could become active following the CSM Rescue burn. This would require slightly different targeting for the terminal rendezvous phase; however, the CSM onboard data would not be affected significantly.

7.3 Partial LM Phasing Less Than 40 FPS

In the event the LM cannot apply 40 FPS of the Phasing burn (the DPS, APS, and RCS have failed), a rendezvous plan similar to that in Section 7.2 will be utilized. A CSM rescue burn is applied at 101:33, one revolution after the DOI burn, and the magnitude of the rescue burn is a function of the LM DV applied (zero to 40 FPS). A canned rescue burn (57.1 FPS retrograde) is also provided onboard for this case. This canned burn is valid for a LM DV of 20 FPS and will be updated after the failed LM burn. A partial LM burn of 20 FPS was chosen simply because it is midway in the zero to 40 FPS partial burn range.

This rendezvous plan differs from the LM partial Phasing greater than 40 FPS plan in that there are two revolutions between the CSM

CSI One and CSM CDH burns. A CSI Two burn is scheduled half way between the CSI One and CDH burns and a CSI Three burn is scheduled half way between the CSI Two and CDH burns. Both the CSI Two and CSI Three burns are nominally zero. The CDH burn for this rescue is targeted so that the delta height between CDH and TPI is 10 nautical miles instead of the nominal 15 nautical miles. The TPI burn is cued as in the nominal case at an elevation angle of 208.3 degrees. The LM could be active after the CSM rescue burn.

7.4 Partial LM Insertion

Should the LM APS fail during the Insertion burn DV be outside the LM-RCS capability either to complete the Insertion burn or to null out the applied DV (i.e., $DV > 40$ or $V_G > .80$) a CSM active rescue will be initiated. For LM applied DV of less than 45 FPS the LM will use the RCS to burn the DV back to zero. If the velocity-to-go is less than 80, the LM will use the RCS to complete the burn. For other LM-applied DV's, the LM will use the RCS to apply an additional 80 FPS along the burn vector to prevent the subsequent maximum ranges from becoming greater than 400 nautical miles. Based upon the LM DV applied, the MCC-H will target the CSM Rescue burn magnitude to preserve the nominal Terminal Phase characteristics, but will result in a TPI one revolution later than nominal. A pad containing this Rescue burn, as well as CSI, CDH, and TPI burn data will be voiced to the CSM approximately 20 minutes after the failed LM Insertion burn. Should the CSM fail to receive these pads, the CSM will apply a canned Rescue burn of 44.9 FPS (regrograde) at DOI + 238 min. (2 revs after DOI). This Rescue burn is valid for a LM applied DV of 120 FPS at the

partial Insertion burn, which is approximately in the middle of the range of LM-applied DV's for which this rescue would be initiated.

After the CSM Rescue burn, the usual concentric rendezvous sequencing of CSI, CDH, and TPI is initiated, with CSI scheduled to occur at one-half revolution after the Rescue burn, and CDH one revolution after the CSI burn. Half-way between CSI and CDH a second, nominally zero, CSI is scheduled, followed by a Plane Change burn half-way between the second CSI and CDH. TPI occurs 35 minutes after CDH at an elevation angle of 208.3, approximately 23 minutes before sunrise. TPF occurs approximately two and one-half revolutions after the failed Insertion burn. The LM could become active after the CSM Rescue burn.

7.5 Zero LM Insertion

Should the LM APS fail at the Insertion burn with less than 45 FPS being applied, the LM will use the RCS to reduce the total DV to zero. The CSM will initiate a ground supplied backup Insertion burn at 102:46:18, three minutes after the LM Insertion TIGN.

This burn preserves the relative trajectory, producing a CSI, CDH, TPI sequence similar to the nominal case; however, due to the increased orbital period of the CSM, the delta times between burns will be somewhat larger than nominal. The resulting TPI time is seven minutes later than nominal and 23 minutes before sunrise. TPI occurs at an elevation angle of 208.3, and at a delta height of 15 nautical miles. The LM could become active after the CSM Backup Insertion burn.

8.0 LM RESCUE ONBOARD DATA

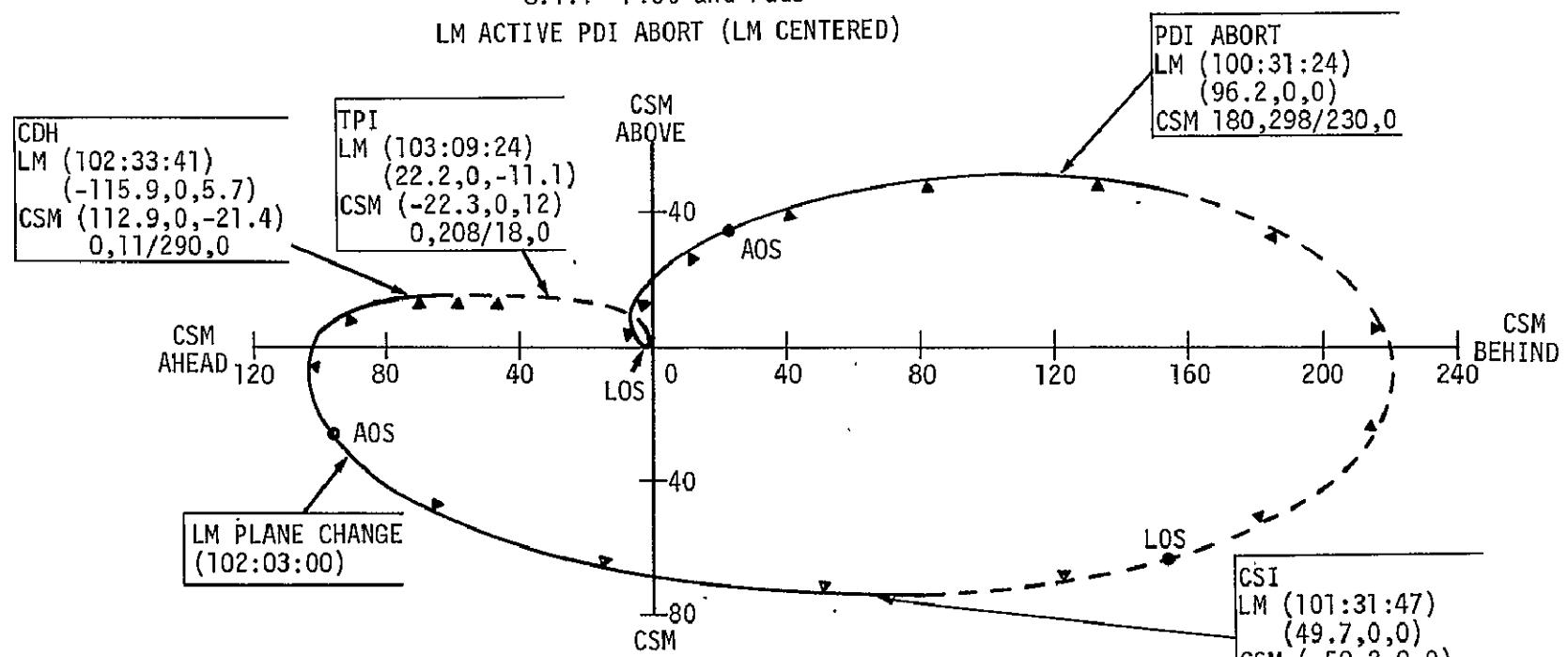
The onboard procedures data required to perform CSM active LM rescues will take the form of a one-page summary checklist which covers the complete timeline for each rescue case. Each summary checklist will show all activities from the failed LM burn to TPF required to complete the LM rescue including program requirements, navigation schedules, and burn schedules. In addition, sunrise/sunset, LOS/AOS, and tracking and burn attitudes will be indicated. In order to condense activities into one page, considerable abbreviation from the nominal checklist format is required. This may necessitate the crew referring to the nominal checklist if additional detail on a particular activity is desired.

Accompanying each one page checklist will be a relative motion plot (LM centered) showing the orbital position of the CSM with respect to the LM. In addition to the plot, maneuver pads for copying onboard data will be provided for each burn during the five contingency rendezvous situations.

The rescue checklist timelines, relative motion plots, and burn data for the two partial LM Phasing cases and one partial LM Insertion case were formulated assuming the CSM applied the canned rescue burn. The onboard data for the zero LM Insertion rescue case assumes a nominal trajectory to Insertion and a ground targeted CSM Backup Insertion burn. The PDI abort case assumes the LM DOI burn was nominal and the LM is active for the rendezvous. The CSM in this case provides a backup to the LM maneuvers after the PDI abort burn.

8.1 · LM PDI Abort

8.1.1 Plot and Pads
LM ACTIVE PDI ABORT (LM CENTERED)



PDI ABORT P76 PAD

84	.	.	.
33	:	:	:

CSM CSI COPY

11	.	.	.
37	.	.	.
75	.	.	.
81	.	.	.
82	.	.	.

LM CSI P76

84	.	.	.
33	:	:	:

CSM PC COPY

33	.	.	.
81	.	.	.

LM PC P76

84	.	.	.
33	:	:	:

CSM CDH COPY

13	.	.	.
75	.	.	.
81	.	.	.

LM CDH P76

84	.	.	.
33	:	:	:

CSM TPI COPY

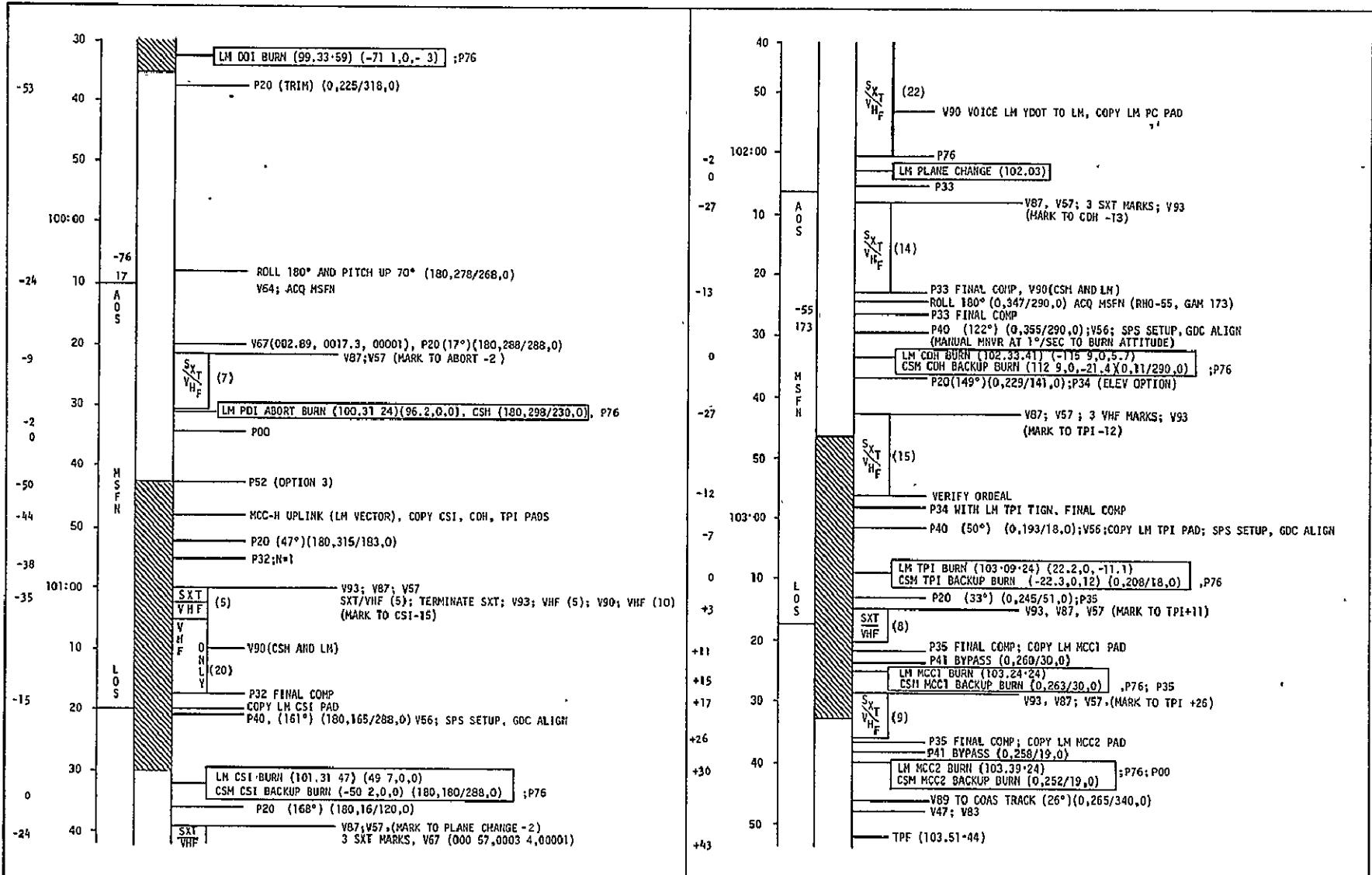
37	.	.	.
58	.	.	.
81	.	.	.
59	.	.	.

LM TPI P76

84	.	.	.
33	:	:	:

8.1.2 Checklist
LM ACTIVE DOI ABORT

-87-

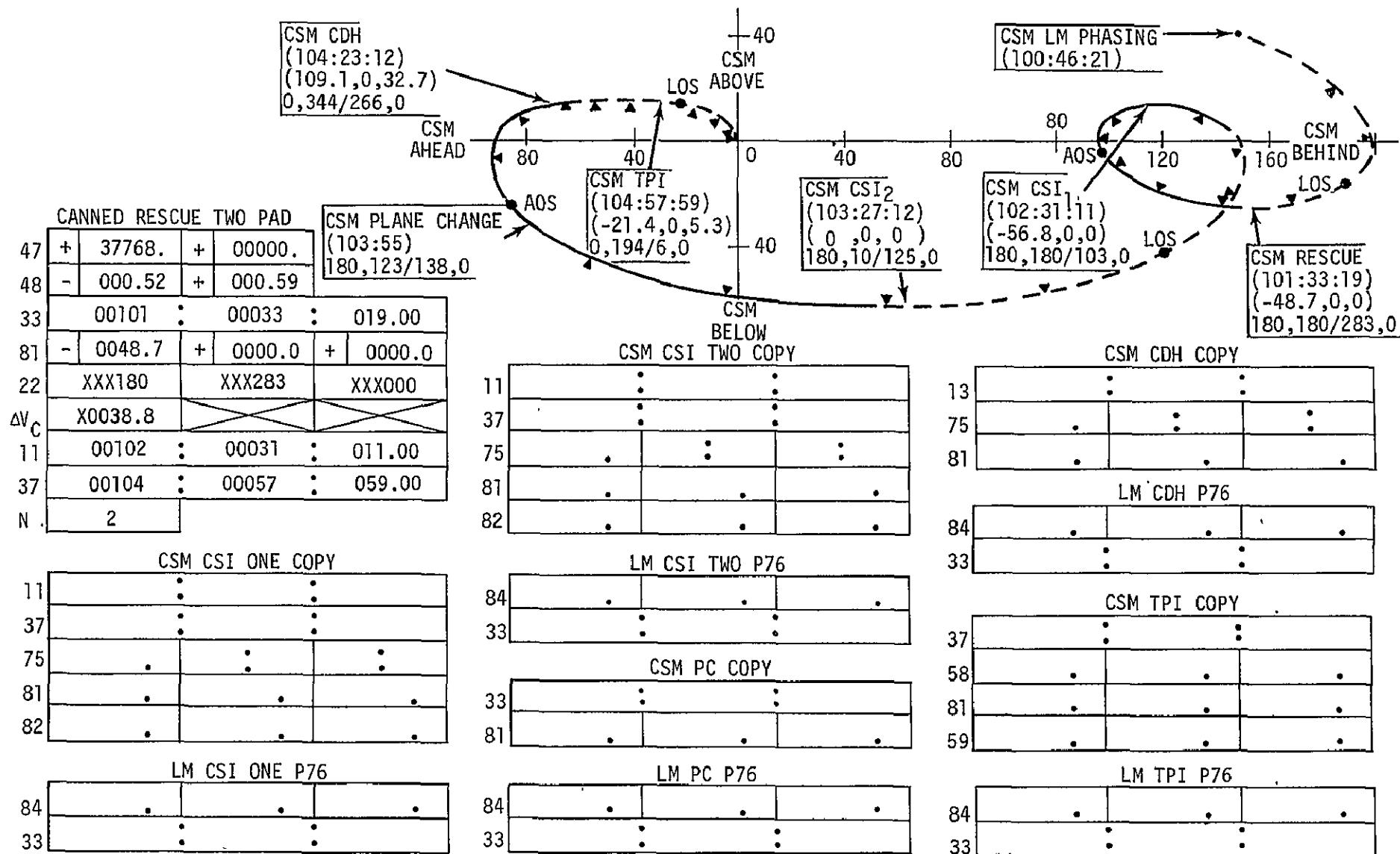


8.2 LM Partial Phasing Greater or Equal To 40 FPS

8.2.1 Plot and Pads

CSM ACTIVE LM RESCUE PARTIAL PHASING ≥40fps (LM CENTERED)

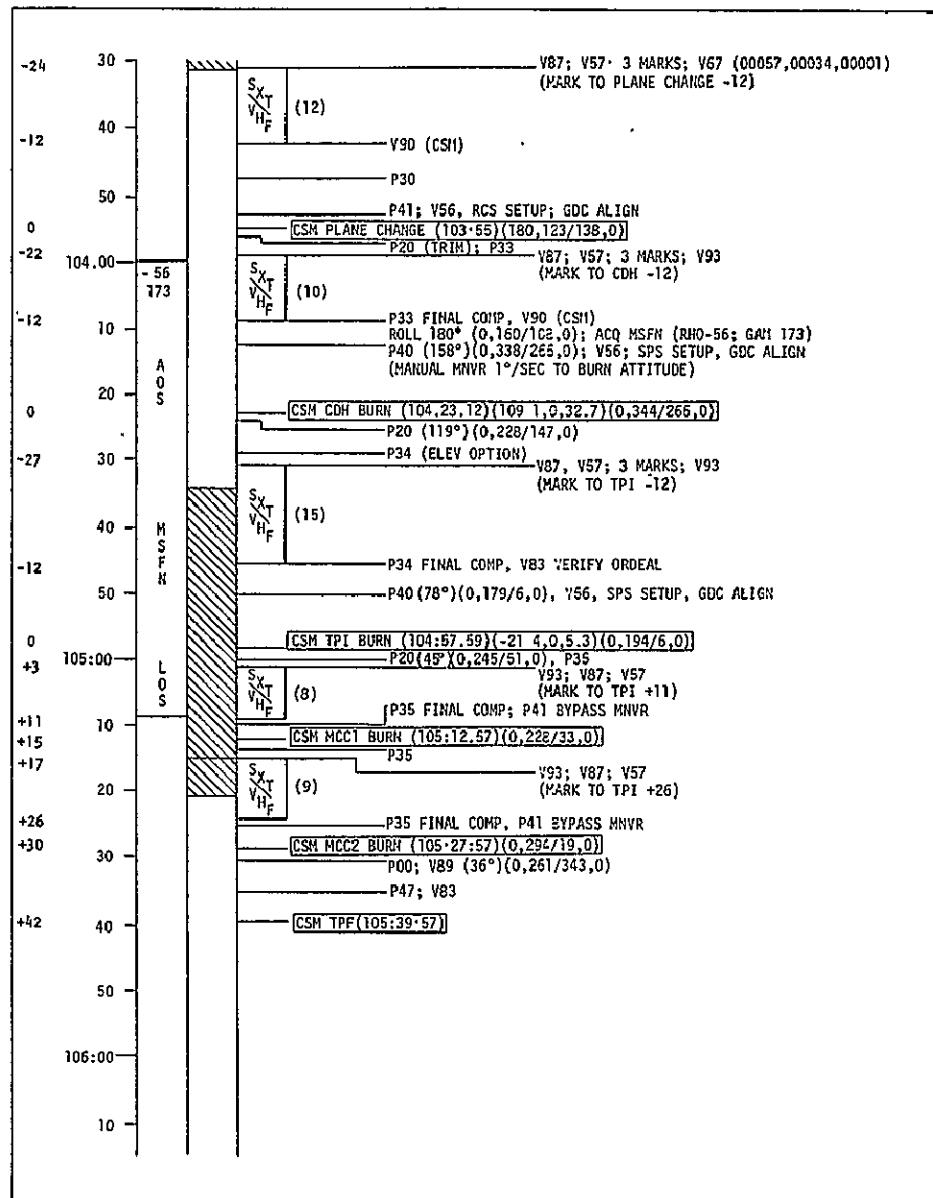
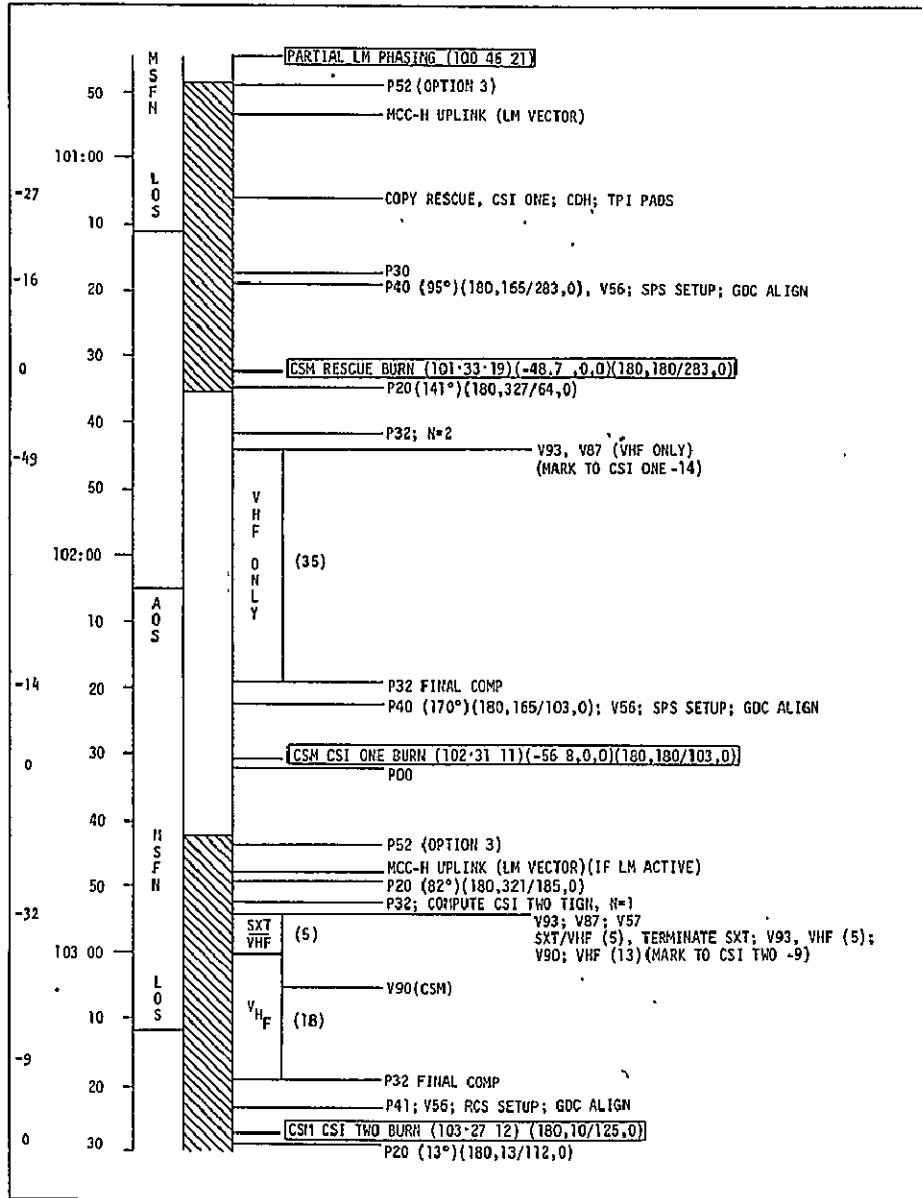
-89-



8.2.2 Checklist

CSM ACTIVE LM RESCUE - LM PHASING \geq 40 FPS (ΔV APPLIED = 40 FPS)

-90-



8.3 LM Partial Phasing Less Than 40 FPS

8.3.1 Plot and Pads
PARTIAL LM PHASING <40 FPS

-92-

CANNED RESCUE TWO PAD

47	+	37768.	+	00000.
48	-	000.52	+	000.59
33		00101	:	00032
			:	059.00
81	-	0057.1	+	0000.0
22		XXX180		XXX283
				XXX000
ΔV_C		X0047.1		
11		00102	:	00030
37		00106	:	00050
			:	057.00
N		4		

CSM CSI ONE COPY

11		•		•
37		•		•
75		•	•	•
81		•	•	•
82		•	•	•

LM CSI ONE P76

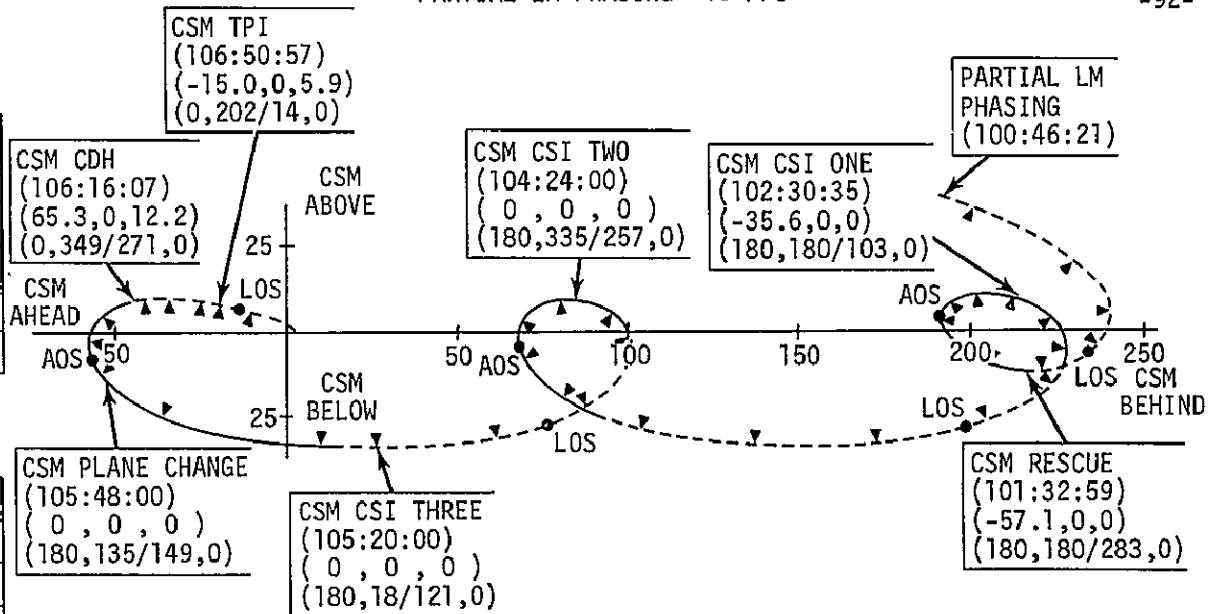
84		•		•
33		•	•	•

CSM CSI TWO COPY

11		•		•
37		•		•
75		•	•	•
81		•	•	•
82		•	•	•

LM CSI TWO P76

84		•		•
33		•	•	•



CSM CSI THREE COPY

11		•		•
37		•		•
75		•	•	•
81		•	•	•
82		•	•	•

LM CSI THREE P76

84		•		•
33		•	•	•

CSM PC COPY

33		•		•
81		•		•

LM PC P76

84		•		•
33		•	•	•

CSM CDH COPY

13		•		•
75		•	•	•
81		•	•	•

LM CDH P76

84		•		•
33		•	•	•

CSM TPI COPY

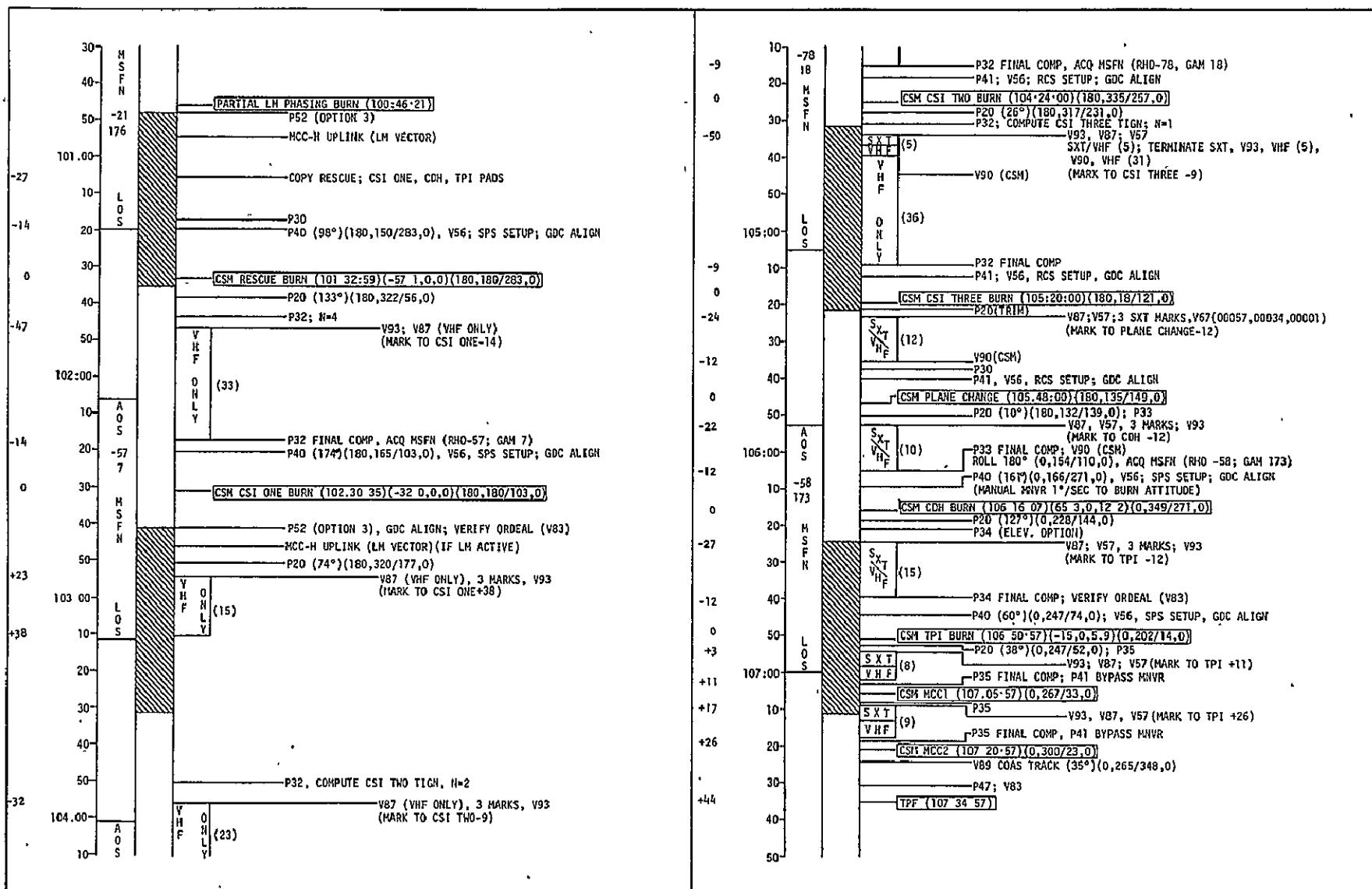
37		•		•
58		•		•
81		•	•	•
59		•	•	•

LM TPI P76

84		•		•
33		•	•	•

8.3.2 Checklist
CSM ACTIVE LM RESCUE PHASING LESS THAN 40 FPS

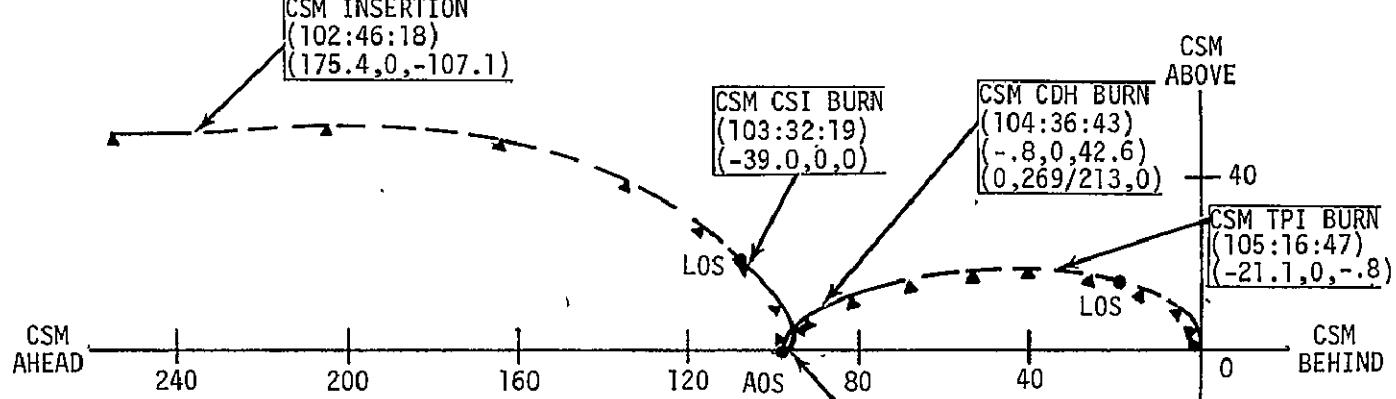
-93-



8.4 LM Zero Insertion

8.4.1 Plot and Pads
CSM ACTIVE LM RESCUE-ZERO INSERTION (LM CENTERED)

-95-



CSM BACKUP
INSERTION PAD
INITIAL

47	+	37768.	+	00000.		
48	-	000.52	+	000.59		
33		00102	:	00046	:	018.00
81	+	175.4	+	0000.0	-	107.1
22		XXX180		XXX272		XXX00
ΔV_C		X192.7		X		X
11		00103	:	00032	:	019.
37		00105	:	00016	:	047.
N		1				

CSM CSI COPY

11	:	:	:
37	:	:	:
75	.	:	:
81	.	.	.
82	.	.	.

LM CSI P76

84	.	.	.
33	:	:	:
84	.	.	.
33	:	:	:
84	.	.	.

CSM PC COPY

33	:	:	:
81	.	.	.
84	.	.	.
33	:	:	:
84	.	.	.

LM CDH P76

84	.	.	.
33	:	:	:
84	.	.	.
33	:	:	:
84	.	.	.

LM PC P76

84	.	.	.
33	:	:	:
84	.	.	.
33	:	:	:
84	.	.	.

CSM CDH COPY

13	.	.	.
75	.	:	:
81	.	.	.
84	.	.	.
33	:	:	:

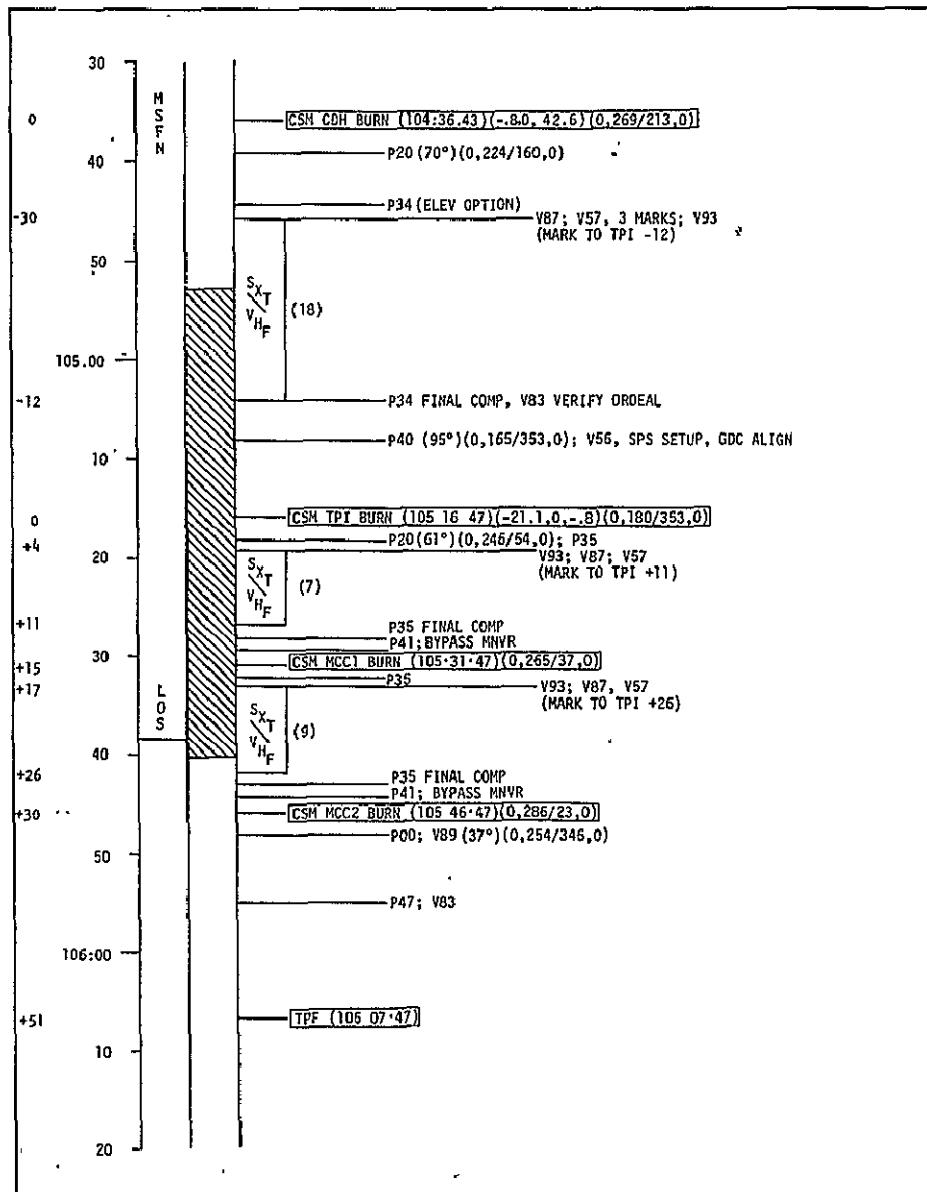
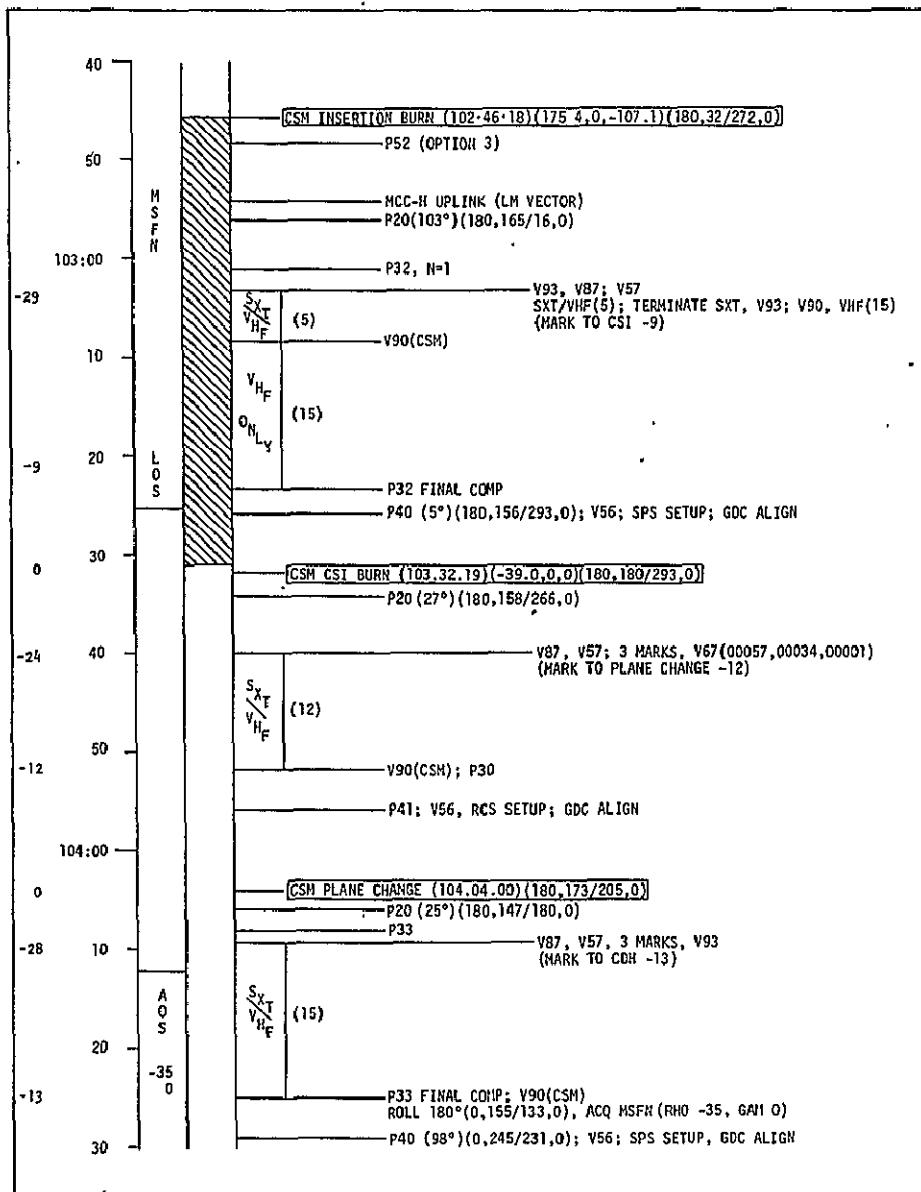
LM TPI P76

84	.	.	.
33	:	:	:
84	.	.	.
33	:	:	:
84	.	.	.

8.4.2 Checklist

CSM ACTIVE LM RESCUE - LM INSERTION = 0

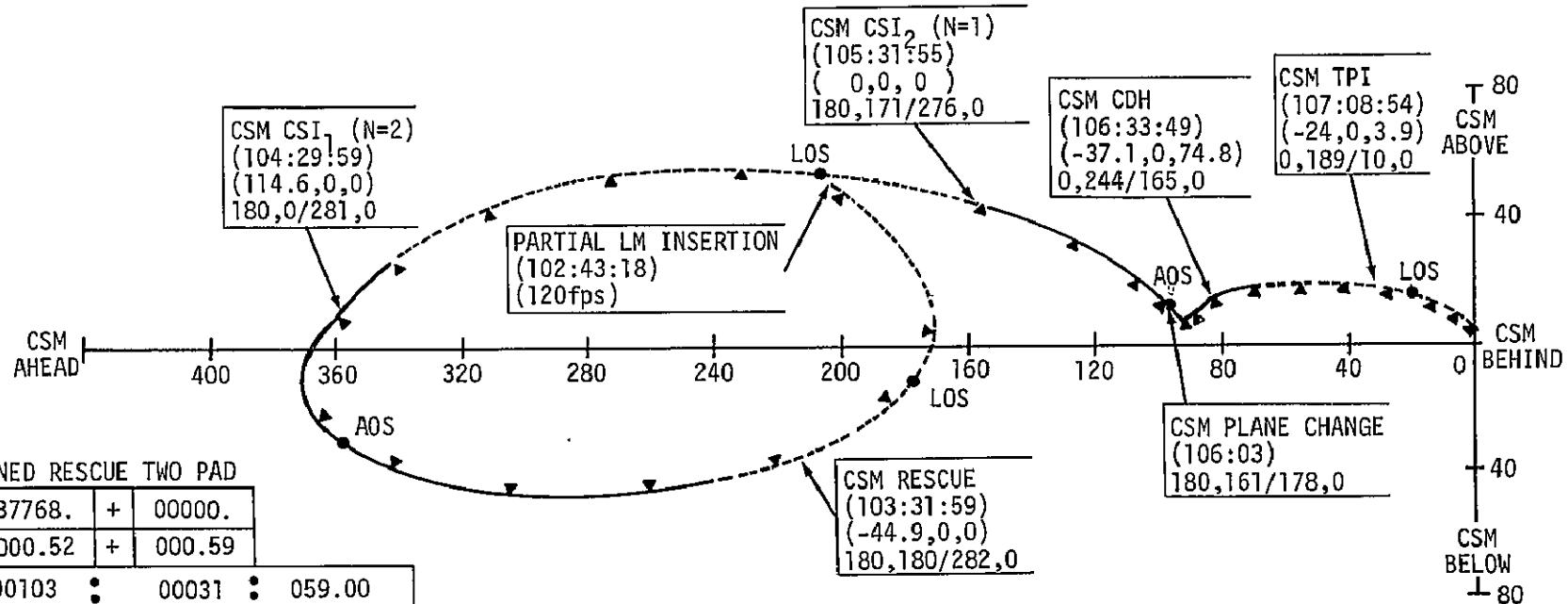
-96-



8.5 LM Partial Insertion

8.5.1 Plot and Pads
CSM ACTIVE LM - PARTIAL LM INSERTION (LM CENTERED)

-98-



CANNED RESCUE TWO PAD					
47	+	37768.	+	00000.	
48	-	000.52	+	000.59	
33		00103	:	00031	:
		0044.9	+	0000.0	+
81				00000.	0000.0
22		XXX180		XXX282	
V _C		X0035.0		X0035.0	
11		00104	:	00029	:
37		00107	:	00008	:
N		2			

CSM CSI ONE COPY					
11		•		•	
37		•		•	
75		•		•	
81		•		•	
82		•		•	

LM CSI TWO P76					
84		•		•	
33		•		•	

CSM TPI COPY					
37		•		•	
58		•		•	
81		•		•	
59		•		•	

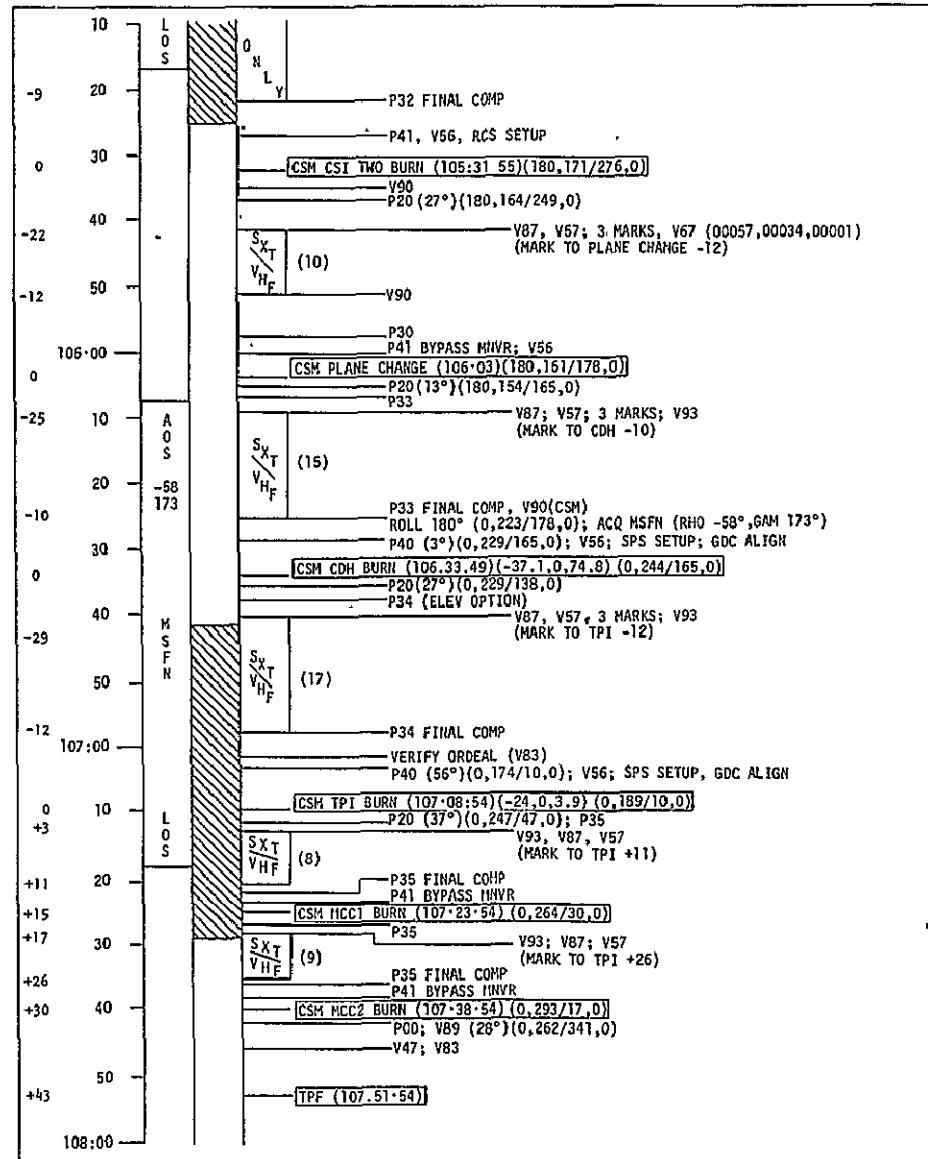
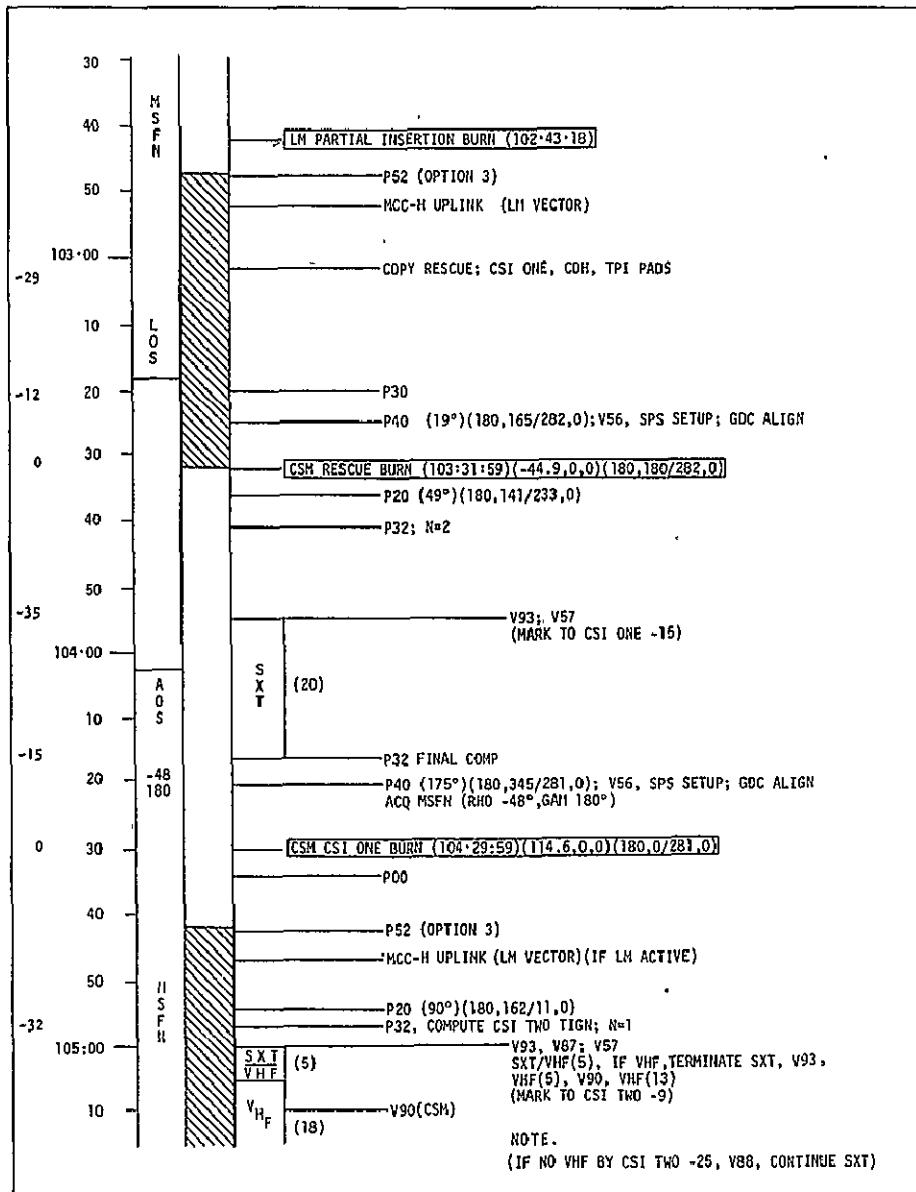
LM CSI ONE P76					
84		•		•	
33		•		•	

LM PC P76					
84		•		•	
33		•		•	

LM TPI P76					
84		•		•	
33		•		•	

8.5.2 Checklist
CSM ACTIVE LM RESCUE - PARTIAL LM INSERTION (ΔV APPLIED = 120 FPS)

-99-



9.0 LM RESCUE PROCEDURES SUMMARY

9.1 Rescue Procedures Ground Rules

- 9.1.1 The GDC will be aligned to the IMU before all SPS burns.
- 9.1.2 P20 will be terminated with a verb 56 before all SPS burns and the Plane Change burn.
- 9.1.3 A rescue pad will be received by the CSM 20 minutes after LM failure in all cases where a Rescue rather than Backup burn is used. The pad will include CSI, CDH, and TPI burn data.
- 9.1.4 The detailed procedures for SPS burns are listed on the timelines because gimbal motor checks are included which are not employed in the nominal backups.
- 9.1.5 Moves to the LEB or CMD seat are not made during auto maneuvers unless time is limited.
- 9.1.6 The 180-degree option should be taken in P32 by loading in a non-zero central angle of 130 degrees.
- 9.1.7 TIGN for CSI Two (and Three) burns will be computed onboard to be half way between the previous CSI TIGN and P32 CDH TIGN.
- 9.1.8 The LM state vector will always be updated unless notified otherwise by the ground.

9.2 LM PDI ABORT

9.2.1 SUMMARY TIMELINE

-102-

GET	EVENT	PROG	GET	EVENT	PROG
	PDT ABORT	-11	(100+20)	V67 LOAD WI(002.89,0017.3,00001) MOVE TO LEB	
(99+33+59)	***** LM DOI BURN(-71.1,0,-.3) ***** CONFIRM LM DOI INCORPORATE P76 DATA (DOI)	-10	(100+21)	RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (17 DEG) (180,288/251,0)	(P20)
-53	(99+37)	-9	(100+22)	CALL MARKING ROUTINES(V87,V57)	
	SUNUP	-2	(100+29)	TERMINATE MARKS (7)	
	(99+38)	0	(100+31+24)	***** LM POI ABORT BURN (96.2,0,0) (180,298/230,0) ***** TARGET DV PROGRAM	(P76)
-47	(99+44)		(100+34)	CMC IDLING PROGRAM	(P00)
-24	(100+07)	-50	100+42)	SUNDOWN IMU REALIGN TO REFSMMAT. (OPTION 3) COPY GYRO TORQUE ANGLES	(P52)
	MOVE TO CMD SEAT ROLL 180 DEG AND PITCH UP 70 DEG AT 1 DEG/SEC ACQUIRE MSFN (V64)	(P00)	(100+48)	MCC-H UPLINK(LM VECTOR) COPY CSI,CDH,TPI PAWS	(P00)
	(100+09)	-44			
	AOS RHO=-76 , GAMMA=17				

GET	EVENT	PROG	-103-	GET	EVENT	PROG
-40	(100+52) RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (47 DEG) (180,315/183,0)	(P20)		-11	(101+21) SPS THRUST PROGRAM, V56 AUTO MNVR TO BURN ATTITUDE (161 DEG) (180,165/288,0)	(P40)
-38	(100+54) CSI TARGETING PROGRAM (P32) (P20) OBTAIN LM CSI AND TPI TIGNS, LOAD N55 WITH R1=00001 R2=20830 R3=13000 MOVE TO LEB			-5	(101+27) SPS THRUST SETUP GDC ALIGN	(P40)
-35	(100+57) CALL MARKING ROUTINE (V93,V87,V57) (AFTER 5 MARKS V93)			-2	(101+30) SUNUP	
-30	(101+02) TERMINATE SXT MARKS, V93 CONTINUE VHF			0	(101+31+47) ***** LM CSI BURN, (49.7,0,0) *****	
-25	(101+07) OUT OF PLANE DATA (V90) VOICE LM Y DOT TO LM CONTINUE VHF				***** CSM CSI BACKUP BURN, (-50.2,0,0) (180,180/288,0) *****	
-15	(101+17) TERMINATE MARKS (20) MOVE TO CMD SEAT CSI TARGETING FINAL COMP COPY CSM CSI SOLUTION OVERWRITE N81 WITH (-) CSM YDOT (V90)	(P32)		-30	(101+33) TERRENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (168 DEG) (180,16/120,0)	(P20)
-12	(101+20) COPY LM CSI PAD LOS			-24	(101+39) CALL MARKING ROUTINES (V87,V57) AFTER 3 SXT MARKS PROCESSED V67 LOAD WR(000.57,1003.4,00001)	

GFT		EVENT		PROG		GFT		EVENT		PROG	
-10	(101+53)	COPY LM PC TIGN OUT OF PLANE DATA(V90) VOICE LM YROT TO LM COPY LM PC PAD				-13	(102+21)	TERMINATE MARKS (14) OUT OF PLANE DATA(V00) VOICE LM YROT TO LM MOVE TO CME SFAT ROLL 180 DEG AT 2 DEG/SEC (P00) ACQ MSFN RH0=-55, GAMMA=173 (0,347/290,0)			
-2	(102+01)	TERMINATE MARKS (22) LOAD TARGET DV PROGRAM (P76)				0	(102+03+00)	CDH TARGETING FINAL COMP (P33) COPY CSM CDH SOLUTION COPY LM CDH PAD			
0	(102+03+00)	***** LM PLANE CHANGE(0,0,0) ***** INCORPORATE P76 DATA (PLANE CHANGE)				-9	(102+25)	SPS THRUST PROGRAM, V56 (P40) MANUAL MNVR AT 1 DEG/SEC TO BURN ATTITUDE (122 DEG) (0,356/290,0)			
-29	(102+05)	CDH TARGETING PROGRAM (P33) POSSIBLE AUTO MNVR TO SXT TRACK (TRTM) OBTAIN LM CDH TIGN LOAD CDH TARGETING DATA				-5	(102+29)	(SPS THRUST SETUP (P40) GDC ALIGN			
-27	(102+07)	CALL MARKING ROUTINE(V87,V57) (PROCESS THREE MARKS,V93) AOS				0	(102+33+41)	***** LM CDH BURN (-115.9,0+5.7), ***** ***** CSM CDH BACKUP BURN (112.9,0,-21.4 (0,11/290,0) *****			

GET	EVENT	PROG	-105-	GET	EVENT	PROG
-33	(102+36) TARGET DV PROGRAM (FDH) RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (149 DEG) (0,22°/141,0) MOVE TO LEB TPI TARGETING PROGRAM (P34) (P20) LOAD N55 WITH R2=20830 R3=13000	(P76)	-5	(103+04)	SPS THRUST SETUP (TPI) GDC ALIGN	(P40)
-27	(102+42) CALL MARKING ROUTINE (V87,V57) (PROCESS THREE MARKS,V93)	(P20)	0	(103+09+24)	LM TPI BURN (22.2,0,-11,1) ***** CSM TPI BACKUP BURN (-22.3,0,12) (0,208/18,0) ***** TARGET DV PROGRAM (TPI)	(P76)
-22	(102+47) SUNDOWN		+02		RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (33 DEG) (0,245/51,0)	(P20)
-17	(102+52) COPY LM TPI TIGN				MCC TARGETING PROGRAM MOVE TO LEB	(P35)
-12	(102+57) TERMINATE MARKS (15) MOVE TO CMD SEAT VERIFY ORDEAL (V83)				CALL MARKING ROUTINE (V93,V87,V57)	
-10	(102+59) TPI TARGETING PROGRAM (P34) FINAL COMP (TIGN OPTION WITH LM TIGN)	(P20)	+03	(103+18)	LOS	
-7	(103+02) SPS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE (50 DEG) (0,193/18,0) COPY LM TPI P76 PAD	(P40)	+11		TERMINATE MARKS (8)	
			+12		MCC1 TARGETING FINAL COMP (P35) COPY CSM MCC1 SOLUTION COPY LM MCC1 PAD	

MISSION F RESCUE TIMELINE APRIL 25, 1969

GET	EVENT	PROG	-106-	GET	EVENT	PROG
13.5	RCS THRUST PROGRAM RYPASS BURN ATTITUDE MNVR (0,260/30,0)	(P41)	+30	(103+39+24)	***** LM MCC2 UURN ***** ***** CSM MCC2 BACKUP BURN (0,252/19,0)	
15	(103+24+24) ***** LM MCC1 THRUST ***** ***** CSM MCC1 BACKUP BURN (0,263/30,0) ***** TARGET DV PROGRAM (MCC1)	(P76)	+39	(103+48)	***** TARGET DV PROGRAM (MCC2) AUTO MNVR TO COAS TRACK(V89) (P00) (26 DEG) (0,265/340,0)	
16	MCC2 TARGETING PROGRAM (P35) (P20) POSSIBLE AUTO MNVR TO SXT TRACK (TRIM) (0,264/27,0)		+39	(103+48)	THRUST MONITORING PROGRAM (P47) ***** * BRAKING GATES * * 30 FPS AT 6000 FT .13 DEG. * * 20 FPS AT 3000 FT .26 DEG. * * 10 FPS AT 2500 FT .54 DEG. * * 5 FPS AT 500 FT 1.60 DEG. *	
17	CALL MARKING ROUTINE(V93,V87,V57)		+43	(103+52)	***** TPF *****	
103+33)	SUNUP					
26	TERMINATE MARKS(9) MOVE TO CMD SEAT					
27	MCC2 TARGETING FINAL COMP (P35) COPY CSM MCC2 SOLUTION COPY LM MCC2 PAD					
28.5	RCS THRUST PROGRAM RYPASS BURN ATTITUDE MNVR (0,258/19,0)	(P41)				

9.2.2 RENDEZVOUS NAVIGATION SUMMARY WITH SUN ANGLES
LM ACTIVE - PDI ABORT
(MISSION F)
IT - INITIATE TRACK
CT - CEASE TRACK
(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE (LOS TO SUN) DEG
-99:34	LM DOI	
99:37	SUNRISE	
100:22	IT (SXT/VHF) (7/7)	156
100:29	CT (SXT/VHF)	171
100:31	LM PDI ABORT	
100:42	SUNSET	
100:57	IT (SXT/VHF) (V93,6/5,V93,14/15)	
101:02	CT (SXT)	
101:17	CT (VHF)	
101:30	SUNRISE	
101:32	LM CSI	
101:39	IT (SXT/VHF) (4/3,V67,00057,00034,00001;18/19)	80
101:44		91
101:49		101
101:54		103
101:59		101
102:01	CT (SXT/VHF)	99
102:03	LM PLANE CHANGE	
102:07	IT (SXT/VHF) (4/3,V93,10/11)	90
102:12		83
102:17		71
102:21	CT (SXT/VHF)	63
102:34	LM CDH	
102:42	IT (SXT/VHF) (4/3,V93/11/12)	8
102:47	SUNSET	
102:57	CT (SXT/VHF)	
103:09	LM TPI	
103:12	IT (SXT/VHF) (V93,8/8)	
103:20	CT (SXT/VHF)	
103:24	LM MCC1	
103:26	IT (SXT/VHF) (V93,9/9)	
103:33	SUNRISE	96
103:35	CT (SXT/VHF)	97
103:39	LM MCC2	
103:52	TPF	

9.2.3 CSM ATTITUDE SUMMARY

** _____ **

INERTIAL AND ORDEAL FDAI BALL GIMBAL ANGLE PROFILE FOR THE LM
ACTIVE PDI ABORT RESCUE CASE (MISSION F) (YAW = 0.0)

** _____ **

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
99:34	0	215	320	
99:38	0	225	318	
100:07	180	278	268	70
100:21	180	288	251	17
100:31	180	298	230	
100:52	180	315	183	47
101:21	180	165	288	161
101:32	180	180	288	
101:33	180	16	120	168
102:03	180	130	143	
102:21	0	347	290	
102:25	0	356	290	122
102:34	0	11	290	
102:36	0	229	141	149
103:02	0	193	18	50
103:09	0	208	18	
103:11	0	245	51	33
103:23	0	260	30	
103:24	0	263	30	
103:25	0	264	27	
103:37	0	258	19	
103:39	0	252	19	
103:40	0	265	340	26
103:52	0	277	340	

9.3 LM PARTIAL PHASING GREATER THAN
OR EQUAL TO 40 FPS

9.3.1 Summary Timeline

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LM PHASING GREATER THAN 40 FPS RESCUE TIMELINE

GET	EVENT	PROG	GET	EVENT	PROG
100+46+21)	LM PHASING HIJRN(34.9,0,-10.55)	*****		RHC-LOCKED	*****
(100+48)	SUNSET	*****	06 18	(COMMANDED R,P,V)	*****
	MOVE TO LER	*****		MONITOR MNVR	*****
(100+49)	IMH REALIGN TO REFSMMAT (OPTION 3)	(P52)	F 50 18	(COMMANDED R,P,V)	*****
(100+54)	MOC-H UPLINK(LM VECTOR)	(P00)		MN BIS TTF(?)=ON(UP)	*****
	MOVE TO CMD SEAT	*****		SPS HE VI V TR(BOTH)=BP	*****
(101+06)	COPY RESCUE,CSI ONE,CDH,TPT PADS	*****		SPS HE VI V(BOTH)=AUTO	*****
-16 (101+17)	EXTERNAL UV PROGRAM	(P30) (P20)	06 18	(COMMANDED R,P,V)	*****
14 (101+19)	SPS THRUST PROGRAM,V56 AUTO MNVR TO HURN ATTITUDE (95 DEG) (180,165/283,0)	(P40)	F 50 18	(COMMANDED R,P,V)	*****
	LOS	*****		SPS THRUST SETUP (P40)	*****
14 (101+19)	KEY 40F F 50 18 (COMMANDED R,P,V) KEY V56E RHC-ARMED ALIGN ROLL TO 180 DEG.	*****		***GDC ALIGN**	*****
		*****		ALT SET KNOB TO 60 NM	*****
		*****		FDAI SELECT-1	*****
		*****		ATT SET THUMBWHEELS TO N18	*****
		*****		NULL ATT ERROR NEEDLES	*****
		*****		ON FDAT 1 WITH ATT	*****
		*****		SFT THUMBWHEELS	*****
		*****		ATT SFT-GDC	*****
		*****		DEPRESS GDC ALIGN OR	*****
		*****		FDAI SELECT-1/2	*****

GET	EVENT	PROG	GET	EVENT	PROG
	ATT SET-TMU	* *		MONITOR GMBL DRIVE	*
	MTVC CHECKS	* *		SEQ AND. TRIM	*
		* *	06 40 (TFI,VG,DVM)		*
		* *-02*00			*
	BMAG MODE(3)-ATT1/RATE?	* *		FDAL SCALE-5/5	*
	TVC SERVO PWR 1-AC1/MNA	* *		VERIFY SPS TH LT-OFF	*
	TVC SERVO PWR 2-AC2/MNR	* *		EMS MODE STBY	*
	TRANS CONTR PWR-ON	* *		EMS FUNCTION-DV SET	*
	RHC PWR NORM 2-AC	* *		LOAD RESCUE VC	*
	GMBL MTRS PITCH 1 STRT-ON	* *		EMS FUNCTION-DV	*
	GMBL MTRS YAW 1 STRT-ON	* *		THG-ARMED	*
	THC-CLOCKWISE	* *		RHC-ARMED	*
	RHC-ARMED	* *		DV THRUST(A)-NORMAL	*
	RHC-VERIFY NO MTVC	* *-00*35		DSKY BLANKS	*
	GMBL MTRS PITCH 2=STRT-ON	* *	06 40 (TFI,VG,DVM)		*
	GMBL MTRS YAW 2=STRT-ON	* *-00*30		EMS MODE=NORMAL	*
	SET SPS GIMBALS TW(2)-	* *		PERFORM ULLAGE	*
	PITCH = -.52	* *	F 99 40 (REQUEST FOR ENGINE ENABLE)		*
	YAW = +.59	* *-00*15	PRO		*
	RHC-VERIFY MTVC	* *	06 40 (TFI,VG,DVM)		*
	THC-NEUTRAL	* *-00*05		*****	*
	RHC PWR NORM 2-AC/DC	* *		*****	*
	RHC-LOCKED	* *		*****	*
	PRO	* *		*****	*
06 18	(COMMANDED R,P,Y)	* *****		*****	*
	MONITOR ATT TRIM	* 0 (101*33*19)		*****	*
F 50 18	(COMMANDED R,P,Y)	* *****		*****	*
	KEY ENTER	* *****		*****	*
F 50 25	(00204,GMBL DRIVE TEST)	* *****		*****	*
	RHC PWR DIRECT(BOTH)=MNA/MNB	* *****		*****	*
	RATE=HIGH	* *****		*****	*
	AUTO RCS SEL A/C ROLL(4)=MNA	* *****		*****	*
	PRO	* *****		*****	*

MISSION F RESCUE TIMELINE APRIL 25, 1969

GET	EVENT	PPNG	APT	EVENT	PPNG
	*****	-112-			
	SET MUC ET-KFSFT-START	* *		FMS MODE-VHF RNG	*
	MONITOR	* *		VHF RNG-PRESET	*
	SPS THRUST IT-ON	* *		VN BIS TTF (?) -OFF	*
	DV INDICATOR-DECREASING	* *		PRO	*
+00+01	VILLAGE-OFF	*		F 37 BR	*
	06 40 (TFC, VG, DVM)	*		MOVE TO LER	*
	MONITOR SPS OPERATION	*	(101+34)	SUNRSE	
	PC INDICATOR=95-105 PSJ	*	(101+35)	RENDFZVOIS NAV PROGRAM	(P20)
	MONITOR SPS ENGIN CUTOFF	*		AUTO MNVR TO SXT TRACK	
	SPS THRUST IT-OFF	*		(141 DEG)	
F 16 42	(TFC, VG, DVM)	*		(180,327/64,0)	
	DV THRUST(BOTH)-OFF	*		CST TARGTING PROGRAM (P35) (P20)	
	MONITOR	*	(101+40)	LOAD N55 R1=+00002	
	PC INDICATOR = 0	*		R2=+20830	
	SPS INJ VLV IND(4)-CLOSE	*		R3=+13000	
	SPS HE VLV TR(BOTH)-HP	*		CALL MARKING ROUTINE(V93,V87)	
	SPS GMPLS RETURN TO SERVO	*		AOS	
	NULL	* -49	(101+42)	TERMINATE MARKS(35)	
	GMPL MTRS-OFF (SEQUENTIALLY)	*	(102+07)	CST TARGTING FINAL	(P32) (P20)
	TVC-SERVO PWR(BOTH)-OFF	*		COMP	
	FDAI SCALE-5/1	*	(102+17)	MOVE TO CMD SEAT	
	RATE-LOW	* -14		SPS THRUST PROGRAM,V56 (P40)	
	ROT CONT PWR DIRECT(BOTH)-OFF	*	(102+20)	AUTO MNVR TO BURN ATTITUDE	
	PRO	*		(170 DEG)	
F 16 85	(VG-BODY)	*		(180,165/103,0)	
	THC-NULL VG COMPONENTS	*			
	THC-LOCKED	* -11			
	PHC-LOCKED	*			
	FMS FUNCTION-VHF RNG	*			

GET	EVENT	PROG	GET	EVENT	PROG
11 (102+20)		*		NULL ATT ERROR NEEDLES ON FDAO,1 WITH ATT SET THUMBWHEELS	*
	" KEY 40E	*		ATT SET-GDC	*
	F 50 18 (COMMANDED R,P,Y)	*		DEPRESS GDC ALIGN PB	*
	KEY V56E	*		FDAO SELECT-1/2	*
	RHC=ARMED	*		ATT SET-TMU	*
	ALIGN ROLL TO 180 DEG.	*		**MTVC CHECKS**	*
	RHC=LOCKED	*		BMAG MODE(3)-ATT1/RATE2	*
	PRO	*		TVC SERVO PWR 1-AC1/MNA	*
	06 18 (COMMANDED R,P,Y)	*		TVC SERVO PWR 2-AC2/MNB	*
	MONITOR MNVR	*		TRANS CONTR PWR-ON	*
	F 50 18 (COMMANDED R,P,Y)	*		RHC PWR NORM 2-AC	*
	MN BUS TIE(2)-ON(UP)	*		GMBL MTRS PITCH 1 STRT=ON	*
	SPS HE VLV TB(ROTH)-BP	*		GMBL MTRS YAW 1 STRT=ON	*
	SPS HE VLV(BOTH)-AUTO	*		THC-CLOCKWISE	*
	NONESS BUS-MNA	*		RHC=ARMED	*
	PRO	*		RHC-VERIFY NO MTVC	*
	06 18 (COMMANDED R,P,Y)	*		GMBL MTRS PITCH 2-STRT=ON	*
	MONITOR ATT TRIM	*		GMBL MTRS YAW 2-STRT=ON	*
	F 50 18 (COMMANDED R,P,Y)	*		SET SPS GIMBALS TW(2)-	*
15 (102+26)	SPS THRUST SETUP	(P40)	*	PITCH = -.52	*
15 (102+26)				YAW = +.59	*
	GDC ALIGN	*		RHC-VERIFY MTVC	*
	ALT SET KNOB TO 60 NM	*		THC-NEUTRAL	*
	FDAO SELECT-1	*		RHC PWR NORM 2-AC/DC	*
	ATT SET THUMBWHEELS TO N18	*		RHC=LOCKED	*
		*		PRO	*
		*	06 18 (COMMANDED R,P,V)	*	
		*	MONITOR ATT TRIM	*	
		*	F 50 18 (COMMANDED R,P,V)	*	
		*	KEY ENTER	*	

GET	EVENT	PROG	-114-	GET	EVENT	PROG
	F 50 26 (00204,GMNL,DRIVE TEST)	*	0	(102+31+11)	*****	
	RHC PWR DIRECT(BOTH)-MNA/MNR	*		CSM CSI ONE TURN(-54.8,0,0)		
	RATE-HIGH	*		(180,180/103,0)		
	AUTO RCS SEL A/C ROLL(4)-MNA	*		*****		
	PRO	*		*****		
	MONITOR GMNL DRIVE	*		SET MDC ET-RESET,START	*	
	SEN AND TRIM	*		MONITOR	*	
-02+00	06 40 (TFI,VG,DVM)	*		SPS THRUST LT-ON	*	
	FDAI SCALE-5/5	*		DV INDICATOR-DECREASING	*	
	VERIFY SPS TH LT-OFF	*		ULLAGE-OFF	*	
	FMS MODE STBY	*		06 40 (TFC,VG,DVM)	*	
	EMS FUNCTION-DV SET	*		MONITOR SPS OPERATION	*	
	LOAD CSI VC	*		PC INDICATOR 95-105 PSI	*	
	FMS FUNCTION-DV	*		MONITOR SPS ENGINE CUTOFF	*	
	RHC-ARMED	*		SPS THRUST LT-OFF	*	
	RHC-ARMED	*		F 16 40 (TFC,VG,DVM)	*	
	DV THRUST(A)-NORMAL	*		DV THRUST(BOTH)-OFF	*	
-00+35	DSKY BLANKS	*		MONITOR	*	
-00+30	06 40 (TFI,VG,DVM)	*		PC INDICATOR = 0	*	
	EMS MODE-NORMAL	*		SPS INJ VLV IND(4)-CLOSE	*	
-00+15	PERFORM ULLAGE	*		SPS HE VLV TB(BOTH)-BP	*	
-00+05	F 99 40 (REQUEST FOR ENGINE ENABLE)	*		SPS GMRLS RETURN TO SERVO	*	
	PRO	*		NULL	*	
	06 40 (TFI,VG,DVM)	*		GMRL MTRS-OFF(SEQUENTIALLY)	*	
*****	*****	*		TVC SERVO PWR(BOTH)-OFF	*	
		*		FDAI SCALE-5/1	*	
		*		RATE-LOW	*	
		*		ROT CONT PWR DIRECT(BOTH)-OFF	*	
		*		PRO	*	
		*		F 16 85 (VG-READY)	*	
		*		THC-NULL VG COMPONENTS	*	
		*		THC-LOCKED	*	

GET	EVENT	PROG	GET	EVENT	PROG
	RHC-LOCKED	*		(103+00)	
	EMS FUNCTION-VHF RNG	*		TERMINATE SXT,CONT. VHF	
	EMS MODE-VHF RNG	*		OUT OF PLANE DATA(V60)	
	VHF RNG-RESET	*		COPY CSM YDOT	
	MN BUS TIE(2)-OFF	*		LOS	
	PRO	*			
	F 37 RR	*			
(102+32)	*****	-9			
	CMC IDLING PROGRAM	(P00)		TERMINATE MARKS(23)	
	MOVE TO LER			CSI TARGETING FINAL	(P32) (P20)
(102+42)	SUNSET			COMP	
	IMU REALIGN TO REFSMMAT	(P52)		OVERWRITE NAI WITH(-)CSM YDOT	
	(OPTION 3)			RCS THRUST PROGRAM,V56	(P41)
(102+47)	MOVE TO CMD SEAT			AUTO MNVR TO BURN ATTITUDE	
	MCC-H UPLINK(LM VECTOR)			RCS THRUST SETUP	(P41)
	(IF LM ACTIVE)			GDC ALIGN	
(102+49)	RENDEZVOUS NAV PROGRAM	(P20)		*****	
	AUTO MNVR TO SXT TRACK			CSM CSI TWO BURN(, ,)	
	(82 DEG)			(180,10/125,0)	
	(180,321/185,0)			*****	
(102+52)	CSI TARGETING PROGRAM	(P32) (P20)		RENDEZVOUS NAV PROGRAM	(P20)
	LOAD NSS R1=+00001			AUTO MNVR TO SXT TRACK	
	R2=+20830			(13 DEG)	
	R3=+13000			(180+13/112,0)	
	MOVE TO LER	-24		SUNRISE	
-32 (102+55)	CALL MARKING ROUTINE(V93,V87,V57)			CALL MARKING ROUTINE(V87,V57)	
	(V93 AFTER 5 MKS PROCESSED)			AFTER 3 MARKS PROCESSED	
				V67,LOAD WR(00057,00034,00001)	

		-116-				
	GET .	EVENT	PROG	GET	EVENT	PROG
-12	(103+43)	TERMINATE MARKS(12)		(104+13)	OUT OF PLANE DATA(V66)	
		OUT OF PLANE DATA(V90)			OVERWRITE NR1 YITH(-) CSM YDOT)	
	(103+47)	EXTERNAL DV TARGETTING (P30) (P20)			SPS THRUST PROGRAM,V56 (P40)	
-8	(103+53)	RCS THRUST PROGRAM,V56 (P41)			MANUAL MNVR (1 DEG/SEC)	
		BY PASS BURN ATTITUDE MNVR			TO BURN ATT	
		RCS THRUST SET-UP (P41)	*****		(158 DEG)	
		GNC ALIGN	**-8		(0.338/266.0)	
6	(103+55)	*	*	(104+13)	KEY 40E	*
		*****	*		F 50 18 (COMMANDED R,P,V)	*
		CSM PLANE CHANGE(180,123/138,0)	*		KEY V56F	*
		*****	*		RHC-ARMED	*
	(103+56)	RENDEZVOUS NAV PROGRAM (P20)	*		ALIGN ROLL TO 0 DEG.	*
		POSSIBLE AUTO MNVR TO SXT TRACK (TRIM)	*		RHC-LOCKED	*
		*	*		PRO	*
	(103+57)	CDH TARGETTING PROGRAM (P33) (P20)	*	06 18	(COMMANDED R,P,V)	*
	(104+00)	Ans	*		MONITOR MNVR	*
-22	(104+01)	CALL MARKING ROUTINE(V87,V57) (V93 AFTER 3MKS PROCESSED)	*	F 50 18	(COMMANDED R,P,V)	*
-12	(104+11)	TERMINATE MARKS(10)	*		MN BUS TIE(2)-ON(UP)	*
		MOVE TO CMD SEAT	*		SPS HE VLV TB(BOTH)-BP	*
		ROLL 180 DEG,ACQ MNVR (0.160/108,0)	*		SPS HE VLV(BOTH)-AUTO	*
		(RHO =56, GAM 173)	*		NONESS BUS-MNA	*
		CDH TARGETTING FINAL COMP	(P33)(P20)		PRO	*
				06 18	(COMMANDED R,P,V)	*
					MONITOR ATT TRIM	*
				F 50 18	(COMMANDED R,P,V)	*
				*****	SPS THRUST SFTUP	(P40)
				-5		

GET	EVENT	PROG	GET	EVENT	PROG
**5	(104+18)	*		THC-NEUTRAL	*
	GDC ALIGN	*		RHC PWR NORM 2-AC/DC	*
		*		RHC-LOCKED	*
		*		PRO	*
	ALT SET KNOB TO 60 NM	*	06 18	(COMMANDED R,P,Y)	*
	FDAI SELECT-1	*		MONITOR ATT TRIM	*
	ATT SET THUMBWHEELS TO N18	*	F 50 18	(COMMANDED R,P,Y)	*
	NULL ATT ERROR NEEDLES	*		KEY ENTER	*
	ON FDAI 1 WITH ATT	*	F 50 25	(00204 GMBL DRIVE TEST)	*
	SET THUMBWHEELS	*		RHC PWR DIRECT(ROT) -MNA/MNB	*
	ATT SET-GDC	*		RATE-HIGH	*
	DEPRESS GUC ALIGN PB	*		AUTO RCS SEL A/C ROLL(4)-MNA	*
	FDAI SELECT-1/2	*		PRO	*
	ATT SET-TMU	*		MONITOR GMBL DRIVE	*
		*		SEQ AND TRIM	*
	MTVC CHECKS	*	06 40	(TFI,VG,DVM)	*
		*			*
	BMAG MODE (3)-ATT1/RATE2	*	FDAI SCALE-5/5	*	
	MTVC SERVO PWR 1-AC1/MNA	*	VERIFY SPS TH LT-OFF	*	
	MTVC SERVO PWR 2-AC2/MNB	*	EMS MODE-STBY	*	
	TRANS CONTR PWR-ON	*	EMS FUNCTION-DV SET	*	
	RHC PWR NORM 2-AC	*	LOAD CDH VC	*	
	GMBL MTRS PITCH 1 STRT-ON	*	EMS FUNCTION-DV	*	
	GMBL MTRS YAW 1 STRT-ON	*	THC-ARMED	*	
	THC-CLOCKWISE	*	RHC-ARMED	*	
	RHC-ARMED	*	OV THRIEST(A)-NORMAL	*	
	RHC-VERIFY NO MTVC	*		*	
		*	DSKY BLANKS	*	
	GMBL MTRS PITCH 2-STR-ON	*		*	
	GMBL MTRS YAW 2-STR-ON	*	06 40 (TFI,VG,DVM)	*	
	SET SPS GIMBALS TW(2)-	*		*	
	PITCH = -.52	*	EMS MODE-NORMAL	*	
	YAW = +.59	*		*	
	RHC-VERIFY MTVC	*	PERFORM ULLAGE	*	

GET	EVENT	PROG	ACT	EVENT	PROG
*-00+05		* *		ROT CONT PWR DIRECT(BOTH)-OFF*	
	F 99 40 (REQUEST FOR ENGINE FAIRING)	* *		PRO	*
	PRO	* *		F 16 85 (VG-RODY)	*
	06 40 (TFI, VG, DVM)	* *		THC-NIJL VG COMPONENTS	*
		* *		THC-LOCKED	*
	06 40 (TFI, VG, DVM)	* *		RHC-LOCKED	*
		* *		EMS FUNCTION-VHF RNG	*
	06 40 (TFI, VG, DVM)	* *		EMS MODE-VHF RNG	*
		* *		VHF RNG-RESET	*
	06 40 (TFI, VG, DVM)	* *		NN BUS TTF(21)-OFF	*
		* *		PRO	*
	06 40 (TFI, VG, DVM)	* *		F 37 BR	*
	SET MUC ET-RESET, START	* *		RENDEZVOUS NAV PROGRAM (P20)	
	MONITOR	* *		AUTO MNVR TO SXT TRACK	
	SPS THRUST LT-ON	* *		(119 DEG)	
	DV INDICATOR-DECREASING	* *		(0.228/147,0)	
*+00+01					
	HILLAGE-OFF	* *		TPI TARGETING PROGRAM (P34) (P20)	
	06 40 (TFC, VG, DVM)	* *		LOAD R2=+20830	
	MONITOR-SPS OPERATION	* *		R3=+13000	
	PC INDICATOR=95-105 PSI	* *		MOVE TO LER	
	MONITOR SPS ENGINE CUTOFF	* *		CALL MARKING ROUTINE(VB7,V57)	
	SPS THRUST LT-OFF	* *		(V93 AFTER 3 MKS PROCESSED)	
	F 16 40 (TFC, VG, DVM)	* *		SUNSET	
	DV THRUST(BOTH)-OFF	* 27			
	MONITOR	* *		TERMINATE MARKS(15)	
	PC INDICATOR = 0	* *		MOVE TO CMR SEAT	
	SPS INJ VLV TND(4)-CLOSE	* *		VERIFY ORDLAL(V83)	
	SPS HE VLV TR(BOTH)-RP	* *		TPI TARGETING FINAL (P34) (P20)	
	SPS GMRLS RETURN TO SERVO	* -12		COMP	
	NULL	* *			
	GMRL MTRS-OFF(SEQUENTIALLY)	* *			
	TVC SERVO PWR(BOTH)-OFF	* *			
	FDAI SCALE-5/1	* *			
	RATE-LOW	* *			

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GET	EVENT	PROG	GET	EVENT	PROG
-7	(104+51)			FDAI SELECT-1	*
	SPS THRUST PROGRAM,V56	(P40)		ATT SET THUMBWHEELS TO N18	*
	AUTO MNVR TO BURN ATTITUDE			NULL ATT ERROR NEEDLES	*
	(78 DEG)			ON FDAI 1 WITH ATT	*
	(0,179/6,0)			SET THUMBWHEELS	*

-7	(104+51)			ATT SET-GDC	*
	KEY 40E			DEPRESS GDC ALIGN PB	*
	F 50 18 (COMMENDED R,P,Y)			FDAI SELECT-1/2	*
	KEY V56E			ATT SET-TMU	*
	RHC-ARMED			**MTVC CHECKS**	
	ALIGN ROLL TO 0 DEG.			BMAG MODE(3)-ATT1/RATE2	*
	RHC-LOCKED			TVC SERVO PWR 1-AC1/MNA	*
	PRO			TVC SERVO PWR 2-AC2/MNB	*
	06 18 (COMMENDED R,P,Y)			TRANS CONTR PWR-ON	*
	MONITOR MNVR			RHC PWR NORM 2-AC	*
	F 50 18 (COMMENDED R,P,Y)			GMBL MTRS PITCH 1 STRT-ON	*
	MN BUS TIE(2)-ON(UP)			GMBL MTRS YAW 1 STRT-ON	*
	SPS HE VLV TB(BOTH)-BP			THC-CLOCKWISE	*
	SPS HE VLV(BOTH)-AUTO			RHC-ARMED	*
	NONESS BUS-MNA			RHC-VERIFY NO MTVC	*
	PRO			GMBL MTRS PITCH 2-STR-ON	*
	06 18 (COMMENDED R,P,Y)			GMBL MTRS YAW 2-STR-ON	*
	MONITOR ATT TRIM			SET SPS GIMBALS TW(2)-	*
	F 50 18 (COMMENDED R,P,Y)			PITCH = -.52	*
-5	(104+53)			YAW = +.59	*
	SPS THRUST SETUP			RHC-VERIFY MTVC	*

-5	(104+53)			THC-NEUTRAL	*
	GDC ALIGN			RHC PWR NORM 2-AC/DC	*
				RHC-LOCKED	*
				PRO	*
	ALT SET KNOB TO 60 NM			06 18 (COMMENDED R,P,Y)	*
				MONITOR ATT TRIM	*

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-120-

GET	EVENT	PPUG	GET	EVENT	PPUG
	F 50 19 (COMMANDER) R,P,Y)	*	0	(104+57+59)	
	KEY ENTER	*		*****	
	F 50 25 (00204,GMPL DRTVF TEST)	*		CSM TPI BURN(-21.4,6,5.3)	
	RHC PWR DIRECT(BOTH)-MNA/MNB	*		(0,194/6,0)	
	RATE-HIGH	*		*****	
	AUTO RCS SEL AYC ROLL(4)-MNA	*		*****	
	PRO	*	*	SET MDC ET-RESET,START	*
	MONITOR GMPL DRTVF	*	*	MONITOR	*
	SEU AND TRIM	*	*	SPS THRUST LT-ON	*
-02+00	06 40 (TFI+VG+DVM)	*	*	DV INDICATOR DECREASING	*
	FDAI SCALE-5/5	*	**00+01		*
	VERIFY SPS TH LT-OFF	*	*	ULLAGE-OFF	*
	EMS MODE STBY	*	*	06 40 (TFC+VG+DVM)	*
	EMS FUNCTION-DV SET	*	*	MONITOR SPS OPERATION	*
	LOAD TPI VC	*	*	PC INDICATOR 95-105 PST	*
	EMS FUNCTION-DV	*	*	MONITOR SPS ENGINE CUTOFF	*
	THC-ARMED	*	*	SPS THRUST LT-OFF	*
	RHC-ARMED	*	*	F 16 40 (TFC+VG+DVM)	*
	DV THRUST(A)-NORMAL	*	*	DV THRUST(BOTH)-OFF	*
-00+35	DSKY BLANKS	*	*	MONITOR	*
-00+30	06 40 (TFI+VG+DVM)	*	*	PC INDICATOR = 0	*
	FMS MODE-NORMAL	*	*	SPS INJ VLV IND(4)-CLOSE	*
-00+15	PERFORM ULLAGE	*	*	SPS HF VLV TR(BOTH)-BP	*
-00+05	F 99 41 (REQUEST FOR ENGINE ENABLE)	*	*	SPS GMPLS RETURN TO SERVO	*
	PRO	*	*	NULL	*
	06 40 (TFI+VG+DVM)	*	*	GMPL MTRS-OFF(SEQUENTIALLY)	*
	*****	*	*	TVC SERVO PWR(BOTH)-OFF	*
		*	*	FDAI SCALE-5/1	*
		*	*	RATE-LOW	*
		*	*	ROT CONT PWR DIRECT(BOTH)-OFF	*
		*	*	PRO	*
		*	*	F 16 85 (VG-HODY)	*
		*	*	THC-MHIL VG COMPONENTS	*
		*	*	THC-LOCKED	*

GET	EVENT	PROG	GFT	EVENT	PROG
*	RHC-LOCKED	*	+16		
*	EMS FUNCTION-VHF RNG	*		MCC TARGETING PROGRAM (P35) (P20)	
*	EMS MODE-VHF RNG	*		POSSIBLE MNVR TO SXT TRACK	
*	VHF RNG-RESET	*	+17		
*	MN BIS TIE (2)-OFF	*		CALL MARKING ROUTINE(V93,V87,V57)	
*	PRO	*	+23		
*	F 37 RR	*		SUNRISE	
+2	*****	*****	+26		
	RENDEZVOUS NAV PROGRAM (P20)			TERMINATE MARKS(9)	
	AUTO MNVR TO SXT TRACK		+27	MOVE TO CMD SEAT	
	(45 DEG)			MCC TARGETING FINAL (P35) (P20)	
	(0,245/51,0)			COMP	
	MOVE TO LER DURING AUTO MNVR	+28.5			
	MCC TARGETING PROGRAM (P35) (P20)				
+3	CALL MARKING ROUTINE(V93,V87,V57)		+30	RCS THRUST PROGRAM (P41)	
	(105+08)			BYPASS BURN ATTITUDE MNVR	
	LOS			RCS THRUST SETUP (P41)	
+11	TERMINATE MARKS(8)			*****	
+12	MCC TARGETING FINAL (P35) (P20)	+31		MCC2 BURN(0,294/19,6)	
+13.5				*****	
	RCS THRUST PROGRAM (P41)			AUTO MNVR TO COAS TRACK(V89) (P00)	
	BYPASS BURN ATTITUDE MNVR			(36 DEG)	
	RCS THRUST SETUP (P41)			(0,261/343,0)	
+15	(105+12+57)			CALL P47 WHEN R=1.25 N.M.	
	*****			THRUST MONITOR PROGRAM (P47)	
	MCC1 BURN(0,228/33,0)			V83, PERFORM BRAKING AND	
	*****			LOS CONTROL	

GET EVENT

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PROG

* BRAKING GATES RETICLE ANGLE *
* 30 FPS AT 6000 FT .13 DEG. *
* 20 FPS AT 3000 FT .26 DEG. *
* 10 FPS AT 2500 FT .54 DEG. *
* 5 FPS AT 500 FT 1.60 DEG. *

*42 (105+39+57)

TPF

9.3.2 NAVIGATION SUMMARY
LM PARTIAL PHASING \geq 40 FPS
P20 NAVIGATION SUMMARY WITH SUN ANGLES
IT-INITIATE TRACK
CT-CEASE TRACK
(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE (LOS TO SUN) DEG
100:46	LM PHASING	
100:48	SUNSET	
101:33	CSM RESCUE BURN	
101:34	SUNRISE	
101:42	IT (VHF) (V93,35/35)	108
102:17	CT (VHF)	144
102:31	CSM CSI ONE BURN	
102:42	SUNSET	
102:55	IT (SXT/VHF) (V93,6/5,V93,14/15)	
103:00	CT (SXT)	
103:15	CT (VHF)	
103:27	CSM CSI TWO BURN	
103:31	SUNRISE	
	IT (SXT/VHF) (4/3;V67,00057,00034,00001;8/9)	66
103:43	CT (SXT/VHF)	83
103:55	CSM PLANE CHANGE	
104:01	IT (SXT/VHF) (4/3,V93,6/7)	82
104:06		72
104:11	CT (SXT/VHF)	61
104:23	CSM CDH BURN	
104:31	IT (SXT/VHF) (4/3,V93,11/12)	1
104:34	SUNSET	14
104:46	CT (SXT/VHF)	
104:58	CSM TPI BURN	
105:01	IT (SXT/VHF) (V93,8/8)	
105:09	CT (SXT/VHF)	
105:13	CSM MCC1 BURN	
105:15	IT (SXT/VHF) (V93,9/9)	
105:21	SUNRISE	97
105:24	CT (SXT/VHF)	100
105:28	CSM MCC2 BURN	
105:40	TPF	

9.3.3 CSM ATTITUDE SUMMARY.
INERTIAL AND ORDEAL FDAI BALL GIMBAL ANGLE
PROFILE FOR LM PHASING \geq 40 FPS (MISSION F)
(YAW = 0.0)

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
100:46	180	307	193	
101:19	180	165	283	95
101:33	180	180	283	
101:35	180	327	64	141
102:20	180	165	103	170
102:31	180	180	103	
102:49	180	321	185	82
103:27	180	10	125	
103:28	180	13	112	13
103:55	180	123	138	
104:11	0	160	108	
104:13	0	338	266	158
104:23	0	344	266	
104:24	0	228	147	119
104:51	0	179	6	78
104:58	0	194	6	
105:00	0	245	51	45
105:13	0	228	33	
105:28	0	294	19	
105:29	0	261	343	36

9.4 LM PARTIAL PHASING LESS THAN 40 FPS

9.4.1 SUMMARY TIMELINE

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GET	EVENT	PROG	GET	EVENT	PROG
	PHASING LESS THAN 45° FRS	*		RHC-LOCKED	*
		*		PRO	*
		*	06 18	(COMMANDED R,P,Y)	*
		*		MONITOR MNVR	*
	(100+46+21)	*****	F 50 18	(COMMANDED R,P,Y)	*
	PARALLEL LM PHASING PURN	*		MN BINS TTE(2)-ON(UP)	*
	*****	*		SPS HF VLV TR(BOTH)-BP	*
-45	(100+48)	*		SPS HF VI V(BOTH)-AUTO	*
	SUNSET	*		ONESS BINS-MNA	*
	INIT REALIGN TO RFFSMAT	(P52)	*	PRO	*
	(OPTION 3)	*	06 18	(COMMANDED R,P,Y)	*
-39	(100+54)			MONITOR ATT TRIM	*
	MCC-H UPLINK(LM VECTOR)	(P00)	F 50 18	(COMMANDED R,P,Y)	*
	RHO=-21, GAMMA=176	*	*****	*****	*
-27	(101+06)		-10 (101+23)	SPS THRUST SETUP	(P40)
	COPY RESCUE,CSI ONE,CDH,TPT PADS	*	*****	*****	*
-16	(101+17)			*****	*
	EXTERNAL DV TARGETTING	(P30)	-10 (101+23)	*****	*
-14	(101+19)			**GDC ALTGN**	*
	SPS THRUST PROGRAM,V56	(P40)			*
	AUTO MNVR TO HURN ATTITUDE	*		ALT SET KNOB TO 60 NM	*
	(90 DEG)	*		FDAI SELECT-1	*
	(180,150/283,0)	*		ATT SET THUMBWHEELS TO N18	*
	LOS	*		NULL ATT ERROR NEEDLES	*
	*****	*		ON FDAI 1 WITH ATT	*
-14	(101+19)	*		SET THUMBWHEELS	*
	KEY 40F	*		ATT SET-ADC	*
	F 50 18 (COMMANDED R,P,Y)	*		DEPRESS GDC ALIGN PR	*
	KEY V56	*		FDAI SELECT-1/2	*
	RHC-ARMED	*			*
	ALTGN ROLL TO	*			*

GET	EVENT	PROG	GET	EVENT	PROG
	ATT SET-TMU	-127-		MONITOR GMBL DRIVE	*
*		*		SEQ AND TRIM	*
*	##MTVC CHECKS##	*	06 40	(TFI,VG,DVM)	*
*	RMAG MODE (3) -ATT1/RATE?	*		FDAT SCALE-5/5	*
*	TVC SERVO PWR 1-AC1/MNA	*		VERIFY SPS TH LT-OFF	*
*	TVC SERVO PWR 2-AC2/MNR	*		EMS MODE STBY	*
*	TRANS CONTR PWR-ON	*		EMS FUNCTION-DV SET	*
*	RHC PWR NORM 2-AC	*		LOAD RESCUE VC	*
*	GMBL MTRS-PITCH 1 STRT-ON	*		EMS FUNCTION-DV	*
*	GMBL MTRS YAW 1 STRT-ON	*		THC-ARMED	*
*	THC-CLOCKWISE	*		RHC-ARMED	*
*	RHC-ARMED	*		DV THRUST(A)-NORMAL	*
*	RHC-VERIFY NO MTVC	*		DSKY BLANKS	*
*	GMBL MTRS-PITCH 2-STR-ON	*		06 40 (TFI,VG,DVM)	*
*	GMBL MTRS YAW 2-STR-ON	*		EMS MODE-NORMAL	*
*	SET SPS GIMBALS TW(2)-	*		PERFORM ULLAGE	*
*	PITCH = -.52	*		F 99 40 (REQUEST FOR ENGINE ENABLE)	*
*	YAW = +.59	*		PRO	*
*	RHC-VERIFY MTVC	*	06 40	(TFI,VG,DVM)	*
*	THC-NEUTRAL	*		*****	*
*	RHC PWR NORM 2-AC/DC	*		(101+32+59)	*
*	RHC-LOCKED	*		*****	*
*	PRO	*		CSM RESCUE BURN(-57,1,0,0)	*
*	06 18 (COMMANDER R,P,Y)	*****		(180,180/283,0)	*
*	MONITOR ATT TRIM	*****		*****	*
F 50 18	(COMMANDER R,P,Y)	*****		SET MDC ET-RESET,START	*
*	KEY ENTER	*		*****	*
F 50 25	(00204,GMBL DRIVE TEST)	*		*****	*
*	RHC PWR DIRECT(ROT) -MNA/MNB	*		*****	*
*	RATE-HIGH	*		*****	*
*	AUTO RCS SEL A/C ROLL(4)-MNA	*		*****	*
*	PRO	*		*****	*

GET	EVENT	PROG	GET	EVENT	PROG
	F 50 1R (COMMANDED R,P,Y)	* *		FDI SELECT-1/2	*
	KEY V56E	* *		ATT SET-TMU.	*
	RHC-ARMED	* *			*
	ALIGN ROLL TO	* *		**MTVC CHECKS**	*
	RHC-LOCKED	* *			*
	PRO	* *		RMAG MODE (3)-ATT1/RATE2	*
06	1R (COMMANDED R,P,Y)	* *		TVC SERVO PWR 1-AC1/MNA	*
	MONITOR MNVR	* *		TVC SERVO PWR 2-AC2/MNB	*
F 50	1R (COMMANDED R,P,Y)	* *		TRANS CONTR PWR-ON	*
	MN BUS TIE(2)-ON(UP)	* *		RHC PWR NORM 2-AC	*
	SPS HE VLV TA(BOTH)-BP	* *		GMBL MTRS PITCH 1 STRT-ON	*
	SPS HE VLV(BOTH)-AUTO	* *		GMBL MTRS YAW 1 STRT-ON	*
	NONESS BUS-MNA	* *		THC-CLOCKWISE	*
	PRO	* *		RHC-ARMED	*
06	1R (COMMANDED R,P,Y)	* *		RHC-VERIFY NO MTVC	*
	MONITOR ATT TRIM	* *		GMBL MTRS PITCH 2-STRT-ON	*
F 50	1R (COMMANDED R,P,Y)	* *		GMBL MTRS YAW 2-STRT-ON	*
**	(102+26)	*		SET SPS GIMBALS TW(2)-	*
	SPS THRUST SETUP (P40)	*		PITCH = -.52	*
**	(102+26)	*		YAW = +.59	*
	GUC ALIGN	*		RHC-VERIFY MTVC	*
	ALT SET KNOB TO 60 NM	* *		THC-NEUTRAL	*
	FDI SELECT-1	* *		RHC PWR NORM 2-AC/DC	*
	ATT SET THUMBWHEELS TO N18	* *		RHC-LOCKED	*
	NULL ATT ERROR NEEDLES	* *		PRO	*
	ON FDI 1-WITH ATT	* *	06 18	(COMMANDED R,P,Y)	*
	SET THUMBWHEELS	* *		MONITOR ATT TRIM	*
	ATT SET-GUC	* *	F 50 18	(COMMANDED R,P,Y)	*
	DEPRESS GUC ALIGN PB	* *		KEY ENTER	*
			F 50 25	(00204,GMBL DRIVE TEST)	*
				RHC PWR DIRECT(BOTH)-MNA/MNB	*
				RATE-HIGH	*
				AUTO RPS SEL A/R ROLL(4)-MNA	*
				PRO	*

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APT	EVENT	PPRG
	MONITOR GMRLS DPFVF SEN AND TRIM.	
06 40 (TFI, VG, DVM)		
02+00	FDAI SCALE-5/5 VERIFY SPS TH LT-OFF FMS MODE STBY FMS FUNCTION-DV SET LOAD CSI (VC EMS FUNCTION-DV / ? THC-ARMED RHC-ARMED DV THRUST (A)-NORMAL	
00+35	DSKY BLANKS	
00+30	06 40 (TFI, VG, DVM) FMS MODE-NORMAL	
00+15	PERFORM ULLAGE	
00+05	F 99 40 (REQUEST FOR ENGINE ENARLE) PRO	
06 40 (TFI, VG, DVM)		
(102+30+35)	***** CSM CST ONE BURN (-35.6,0,0) (180,180/103,0) *****	
	SET MDC FT-RESFT-START MONITOR SPS THRUST LT-ON DV INDICATOR-DECREASING	
	ULLAGE-OFF	
06 40 (TFI, VG, DVM)		
	MONITOR SPS OPERATION PC INDICATOR-95-105 PSI	
	MONITOR SPS ENGINE CUTOFF	
	SPS THRUST LT-OFF	
F 16 40 (TFI, VG, DVM)		
	DV THRUST (BOTH)-OFF MONITOR PC INDICATOR = 0	
	SPS INJ VLV TRND(4)-CLOSE	
	SPS HF VLV TR (BOTH)-BP	
	SPS GMRLS RETURN TO SERVO	
	NULL	
	GMRL MTRS-OFF (SEQUENTIALLY)	
	TVC SERVO PWR (BOTH)-OFF	
	FDAI SCALE-5/1	
	RATE-LOW	
	ROT CONT PWR DIRECT (BOTH)-OFF	
	PRO	
F 16 85 (VG-READY)		
	THC-NULL VG COMPONENTS	
	THC-LOCKED	
	RHC-LOCKED	
	EMS FUNCTION-VHF RNG	
	EMS MODE-VHF RNG	
	VHF RNG-RESET	
	MN BLS TTF(2)-OFF	

GET	EVENT	PROG	GET	EVENT	PROG
*	PRO	* -32	(103+52)	MOVE TO LEB	
*	F 37 RR	*		CALL MARKING ROUTINE (V87)	
*****	*****			(V93 AFTER 3 MARKS PROCESSED)	
(102+32)					
(102+40)	CMC IDLING PROGRAM	(P00)	(104+00)	AOS	
	SUNSET	-9	(104+15)	TERMINATE MARKS (23)	
	IMU REALIGN TO REFSMMAT	(P52)		MOVE TO CMO SEAT	
	(OPTION 3)			CSI TARGETING FINAL COMP (P32)	
	GDC ALIGN			ACQ. MSFN, $RHO = -77.5$, $GAMMA = 18.1$	
	VFRIFY ORDEAL (V83)				
(102+46)	MCC-H UPLINK(LM VECTOR, IF LM ACTIVE)	-6	(104+18)	RCS THRUST PROGRAM, V56 (P41)	
(102+51)	RENDEZVOUS NAV PROGRAM	(P20)		AUTO MNVR TO BURN ATTITUDE	
	AUTO MNVR TO SXT TRACK	-2	(104+22)		
	(74 DEG)			RCS THRUST SETUP, GDC ALIGN (P41)	
	(180,320/177,0)			(104+24+00)	
*23	(102+54)			CSM CSM TWO BURN (180,335/257,0)	
	CALL MARKING ROUTINE (V87)	-54	(104+26)	*****	
	THREE MARKS, V93			*****	
*38	(103+09)			RENDEZVOUS NAV PROGRAM (P20)	
	TERMINATE MARKS (15)			AUTO MNVR TO SXT TRACK	
(103+11)				(26 DEG)	
	LOS	-53	(104+27)	(180,317/231,0)	
(103+29)				CSI TARGETING PROGRAM (P32)	
	SUNRTSE			LOAD N55 WITH	
-38	(103+46)			R1 = +00001	
	CST TARGETING PROGRAM	(P32)		R2 = +20830	
	LOAD N55 WITH			R3 = +13000	
	R1 = +00002			MOVE TO LEB	
	R2 = +20830				
	R3 = +13000				

GET	EVENT	GET	EVENT	PROG
-50	(104+30) CALL MARKING ROUTINE(V93,V87,V57) (V93 AFTER 5 MARKS PROCESSED)	-25	(105+26)	RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (TRIM) (180,18/120,0)
(104+33)	SUS-SET	-24	(105+24)	MOVE TO LEP CALL MARKING ROUTINE(V87,V57)
-45	(104+35), TERMINATE SXT MARKS,V93 CONTINUE VHF	"	"	AFTER 3 SXT MARKS PROCESSED V67,LOAD WD(00057,00034.00001)
-40	(104+40) OUT OF PLANE DATA(V90) VOICE LM YDOT TO LM CONTINUE VHF	-12	(105+36)	TERMINATE MARKS(12) MOVE TO CMD SEAT OUT OF PLANE DATA(V90)
(105+04)	LOS	-9	(105+39)	EXTERNAL DV TARGETING
-9	(105+11) TERMINATE MARKS(41) MOVE TO CMD SEAT CSM TARGETTING FINAL COMP	-6	(105+42)	(P30) RCS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE
-5	(105+15) RCS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE	-2	(105+46)	(P41) RCS THRUST SETUP,GDC ALIGN
-2	(105+18) RCS THRUST SETUP,GDC ALIGN	0	(105+48+00)	(P41)
0	(105+20+00) ***** CSM CSM THREE TURN (180,18/121,0)	-26	(105+50)	***** CSM PLANE CHANGE (180,135/149,0)
(105+21)	SUSRTSF	-25	(105+51)	***** RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (10 DEG) (180,132/139,0)
				ODH TARGETTING PROGRAM MOVE TO LEP

	GET	EVENT	-133-	PROG	GET	EVENT	PROG
	(105+52)			*		SPS HE VLV(BOTH)-AUTO	*
	AOS			*		NONESS BUS-MNA	*
-22	(105+54)	CALL MARKING ROUTINE(V87,V57) (V93 AFTER 3 MARKS PROCESSED)		*		PRO	*
-12	(106+04)	TERMINATE MARKS(10). MOVE TO CMD SEAT CDH TARGETING FINAL COMP (P33) OUT OF PLANE DATA(V90) OVERWRITE N81 WITH (-) CSM YDOT ROLL 180 DEG, ACQ MSFN RH0=-58, GAMMA=173 (0.154/110,0)		*	06 18 (COMMANDER R,P,Y) MONITOR ATT TRIM		*
				*	F 50 18 (COMMANDER R,P,Y)	*	
				*	*****	*	
				-5	(106+11)		
					SPS THRUST SETUP (P40)		
					*****	*	
				-5	(106+11)	*	
					##GDC ALIGN##	*	
-8	(106+08)	SPS THRUST PROGRAM,V56 (P40) MANUAL MNVR TO 1 DEG/SET TO BURN		*		ALT SET KNOB TO 60 NM	*
		ATTITUDE (161 DEG) (0,166/271,0)		*	FDATI SELECT-1	*	
		*****		*	ATT SET THUMBWHEELS TO N18	*	
		*****		*	NULL ATT ERROR NEEDLES	*	
		*****		*	ON FDATI 1 WITH ATT	*	
		*****		*	SET THUMBWHEELS	*	
		*****		*	ATT SET-GDC	*	
		KEY 40E		*	DEPRESS GDC ALIGN PB	*	
		F 50 18 (COMMANDER R,P,Y)		*	FDATI SELECT-1/2	*	
		KEY V56E		*	ATT SET-TMU	*	
		RHC-ARMED		*	##MTVC CHECKS##	*	
		ALIGN ROLL TO		*		*	
		RHC-LOCKED		*	BMAG MODE(3)-ATT1/RATE2	*	
		PRO		*	TVC SERVO PWR 1-AC1/MNA	*	
		06 18 (COMMANDER R,P,Y)		*	TVC SERVO PWR 2-AC2/MNB	*	
		MONITOR MNVR		*	TRANS CONTR PWR-ON	*	
		F 50 18 (COMMANDER R,P,Y)		*	RHC PWR NORM 2-AC	*	
		MN BUS TIE(2)-ON(UP)		*	GMBL MTRS PITCH 1 STRT-ON	*	
		SPS HE VLV TB(BOTH)-BP		*		*	

GFT	EVENT	PROM	AFT	EVENT	PROG
	GMRL MTRS YAW 1 STRT-ON	* *		RHC-ARMED	*
	RHC-CLOCKWISE	* *		HV THRUST(A)-NORMAL	*
	RHC-ARMED	* *-00+35		OSKY BLANKS	*
	RHC-VERIFY NO OTVC	* *		06 40 (TFI,VG,DVM)	*
	GMRL MTRS PITCH 2-STRT-ON	* *-00+30		EMS MODE-NORMAL	*
	GMRL MTRS YAW 2-STRT-ON	* *		PERFORM ULLAGE	*
	SET SPS GIMBALS TW(2)-	* *		F 99 40 (REQUEST FOR ENGINE ENARL)	*
	PITCH = -.52	* *-00+15		PRO	*
	YAW = +.59	* *		06 40 (TFI,VG,DVM)	*
	RHC-VERIFY MTVC	* *-00+05		*****	*
	RHC-NEUTRAL	* *		*****	*
	RHC PWR NORM 2-AC/DC	* *		*****	*
	RHC-LOCKED	* *		*****	*
	PRO	* *-00+00		*****	*
	06 18 (COMMANDER R,P,Y)	* 0 (106+16+07)		*****	*
	MONITOR ALT THRM	* *		*****	*
F 50 18 (COMMANDER R,P,Y)				CSM CDH BURN(65.3+0,12.2)	*
	KEY ENTER			(0,349/271,0)	*
F 50 25 (00204,GMRL DRIVE TEST)				*****	*
	RHC PWR DIRECT(BOTH)-MNA/MNB	* #####		SET MDC ET-RESET+START	*
	RATE-HIGH	* *		MONITOR	*
	AUTO RCS SEL A/C ROLL(4)-MNA	* *		SPS THRUST LT-ON	*
	PRO	* *		DV INDICATOR-DECREASING	*
	MONITOR GMRL DRIVE	* *		ULLAGE-OFF	*
	SEN AND TRIM	* *+00+01		06 40 (TFC,VG,DVM)	*
06 40 (TFI,VG,DVM)				MONITOR SPS OPERATION	*
02+00				PC INDICATOR-95-105 PSI	*
	FDAL SCALE-5/5	* *		MONITOR SPS ENGINE CUTOFF	*
	VERIFY SPS TH LT-OFF	* *		SPS THRUST LT-OFF	*
	EMS MODE STBY	* *		F 16 40 (TFC,VG,DVM)	*
	FMS FUNCTION-DV SET	* *		DV THRUST(BOTH)-OFF	*
	LOAD CDH VC	* *		MONITOR	*
	EMS FUNCTION-DV	* *			
	RHC-ARMED	* *			

GET	EVENT	PROG	GET	EVENT	PROG
	PC INDICATOR = 0	* -27	(106+24)	CALL MARKING ROUTINE(V87,V57)	
	SPS INJ VLV IND(4)-CLOSE	*		(V93 AFTER 3 MARKS PROCESSED)	
	SPS HE VLV TB(BOTH)-BP	*		SUNSET	
	SPS GM8LS RETURN TO SERVO	*	(106+26)		
	NULL	*		TERMINATE MARKS(15)	
	GMAL MTRS-OFF(SEQUENTIALLY)	* -12	(106+39)	MOVE TO CMN SEAT	
	TVC SERVO PWR(BOTH)-OFF	*		TPI TARGETING FINAL COMP	(P34)
	FOAI SCALE45/1	*		VERIFY ORDFAL(V83)	
	RATE-LOW	*			
	ROT CONT PWR DIRECT(BOTH)-OFF	*			
	PRO	* -7	(106+44)	SPS THRUST PROGRAM,V56	(P40)
F. 16 85	(VG-BODY)	*		AUTO MNVR TO BURN ATTITUDE	
	THC-NULL VG COMPONENTS	*		(60 DEG)	
	THC-LOCKED	*		(0,247/74,0)	
	RHC-LOCKED	*			
	EMS FUNCTION-VHF RNG	*		KEY 40E	
	EMS MODE-VHF RNG	*		F 50 18 (COMMANDER R,P,̄Y)	
	VHF RNG-RESET	*		KEY V56E	
	MN BUS TIE(2)-OFF	*		RHC-ARMED	
	PRO	*		ALIGN BOLL TO	
F. 37 88	"	*		RHC-LOCKED	
-33	(106+18)			PRO	
	RENDEZVOUS NAV PROGRAM	(P20)	*	06 18 (COMMANDER R,P,̄Y)	
	AUTO MNVR TO SXT TRACK		*	MONITOR MNVR	
	(127 DEG)		*	F 50 18 (COMMANDER R,P,̄Y)	
	(0,228/144,0)		*	MN BUS TIE(2)-ON(UP)	
	MOVE TO LEB		*	SPS HE VLV TB(BOTH)-BP	
-29	(106+22)			SPS HE VLV(BOTH)-AUTO	
	TPI TARGETING PROGRAM	(P34)	*	NONESS BUS-MNA	
	LOAD N55	*	*	PRO	
	R2=+20830	*	*	06 18 (COMMANDER R,P,̄Y)	
	R3=+13000	*	*	MONITOR ATT TRIM	

CHT	EVENT	PROG	CHT	EVENT	PROG
*	F 50 14 (COMMANDED R,P,Y)	*		GMRL MTRS YAW 2-STRT-ON	*
-5 (106+46)	SPS THRUST SET UP	(P40)	*	SET SPS GMRLS TW(2)-	*
-5 (106+46)			*	PITCH = -.52	*
*	**GDC ALIGN**		*	YAW = +.59	*
*	ALT SET KNOB TO 60 NM	*		RHC-VERIFY MTVC	*
*	FDAI SELECT-1	*		RHC-NEUTRAL	*
*	ATT SET THUMBWHEELS TO 0118	*		RHC PWR NORM. 2-AC/DC	*
*	FULL ATT-ERROR NEEDLES	*		RHC-LOCKED	*
*	ON FDAI 1 WITH ATT-1	*		PRO	*
*	SET THUMBWHEELS	*	06 18	(COMMANDED R,P,Y)	*
*	ATT SET-GDC	*		MONITOR ATT TRIM	*
*	DEPRESS GDC ALIGN PR	*	F 50 18	(COMMANDED R,P,Y)	*
*	FDAI SELECT-1/2	*		KEY ENTER	*
*	ATT SET-TMU	*	F 50 25	(00204,GMRL DRIVE TEST)	*
*	**TVC CHECKS**	*-02+00		RHC PWR DIRECT (BOTH)-MNA/MNB	*
*	R'AG MODE (3)-ATT1/RATE2	*		RATE-HIGH	*
*	TVC SERVO PWR 1-AC1/MNA	*		AUTO, RCS SEL A/F ROLL(4)-MNA	*
*	TVC SERVO PWR 2-AC2/MNB	*		PRO	*
*	TRANS CONTR PWR-ON	*		MONITOR GMRL DRIVE	*
*	RHC PWR NORM 2-AC	*		SFO AND TRIM	*
*	GMRL MTRS PITCH 1 STRT-ON	*	06 40	(TFI, VG, DVM)	*
*	GMRL MTRS YAW 1 STRT-ON	*		FDAI SCAL F-5/5	*
*	THC-CLOCK N ISF	*		VERIFY SPS TH LT-OFF	*
*	RHC-ARMED	*-00+35		EMS MODE- STBY	*
*	RHC-VERIFY NO MTVC	*		EMS FUNCTION-DV SET	*
*	GMRL MTRS PITCH 2-STRT-ON	*		LOAD TH VC	*
*				EMS FUNCTION-DV	*
*				THC-ARMED	*
*				RHC-ARMED	*
*				DV THRUST(A)-NORMAL	*
*				DSKY BLANKS	*

GET	EVENT	PROG	GET	EVENT	PROG
000+30		# *		SPS GMBLS RETURN TO SERVO *	
	06 40 (TFI, VG, DVM)	# *		NULL *	
	EMS MODE=NORMAL	# *		GMBL MTRS=OFF (SEQUENTIALLY) *	
00+15		# *		TVC SERVO PWR(BOTH)=OFF *	
	PERFORM ULLAGE	# *		FDAI SCALE=5/1 *	
00+05		# *		RATE=LOW *	
	F 99 40 (REQUEST FOR ENGINE ENABLE)	# *		ROT CONT PWR DIRECT(BOTH)=OFF *	
	PRO	# *		PRO *	
	06 40 (TFI, VG, DVM)	# *		F 16 85 (VG-BODY) *	
	***** (106+50+57)	#		THC=NULL VG COMPONENTS *	
	CSM TPI BURN(-15,0,5.9)	#		THC=LOCKED *	
	(0,202/14,0)	#		RHC=LOCKED *	
	*****	#		EMS FUNCTION-VHF RNG *	
	SET MDC ET=RESET,START	# *		EMS MODE-VHF RNG *	
	MONITOR:	# *		VHF RNG=RESET *	
	SPS THRUST LT=ON	# *		MN BUS TIE(2)=OFF *	
	DV INDICATOR=DECREASING	# +2		PRO *	
00+01		#		F 37 BB	
	ULLAGE=OFF	#		*****	
	06 40 (TFC, VG, DVM)	#		RENDEZVOUS NAV PROGRAM (P20)	
	MONITOR SPS OPERATION:	#		AUTO MNVR TO SXT TRACK	
	PC INDICATOR=95-105 PSI	#		(38 DEG)	
	MONITOR SPS ENGINE CUTOFF:	#		(0,247/52,0)	
	SPS THRUST LT=OFF	# +3		MCCI TARGETTING PROGRAM (P35) (P20)	
	F 16 40 (TFC, VG, DVM)	#		MOVE TO LEB	
	DV THRUST(BOTH)=OFF	#		CALL MARKING ROUTINE(V93, V87, V57)	
	MONITOR:	#		LOS	
	PC INDICATOR = 0	# +11		TERMINATE MARKS(8)	
	SPS INJ VLV IND(4)=CLOSE	#		MCCI TARGETTING FINAL COMP (P35)	
	SRS HE VLV TR(BOTH)=BP	# +12			

(107+00)

GET	EVENT	PROG	-138-	GET	EVENT	PROG
+13.5	RCS THROTTLE PROGRAM BYPASS BURN ATTITUDE MNVR RCS THRUST SETUP	(P41)	+33	(107+24)	THRUST MONITORING PROGRAM (P47) V83, PERFORM BRAKING AND LOS CONTROL	
+15	(107+05+57)				***** CSM MCC1 (0,267/33,0) *****	***** # BRAKING GATES PARTICLE ANGLE # # 30 FPS AT 6000 FT .13 DEG. # # 20 FPS AT 3000 FT .26 DEG. # # 10 FPS AT 2500 FT .54 DEG. # # 5 FPS AT 500 FT 1.60 DEG. # *****
+16	MCC2 TARGETING PROGRAM (P35) (P20)					
+17	CALL MARKING ROUTINE (V93, V87, V57)					
(107+14)	SUNRTSF		+44	(107+34+57)		
+26	TERMINATE MARKS (9) MOVE TO CMD SEAT				***** TPF *****	
+27	MCC2 TARGETING FINAL COMP (P35)					
+28.5	RCS THRUST PROGRAM BYPASS BURN ATTITUDE MNVR RCS THRUST SETUP	(P41)				
+30	(107+20+57)				***** CSM MCC2 (0,300/23,0) *****	
+32	MNVR TO COAS TRACK (V89) (35 DEG) (0,265/348,0)	(P00)				

9.4.2 RENDEZVOUS NAVIGATION SUMMARY WITH SUN ANGLES
CSM ACTIVE LM RESCUE - PARTIAL PHASING < 40 FPS
(MISSION F)
IT - INITIATE TRACK
CT - CEASE TRACK
(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE (LOS TO SUN) DEG
100:46	PARTIAL LM PHASING	
100:48	SUNSET	
101:33	CSM RESCUE	
101:35	SUNRISE	
101:44	IT (VHF) (V93,33/33)	15
101:49		30
101:54		46
101:59		63
102:04		80
102:09		97
102:14		113
102:17	CT (VHF)	123
102:31	CSM CSI ONE	
102:40	SUNSET	
102:54	IT (VHF) (4/3,V93,11/12)	
103:09	CT (VHF)	
103:29	SUNRISE	
103:52	IT (VHF) (4/3,V93,19/20)	48
103:57		68
104:02		98
104:07		105
104:12		129
104:15	CT (VHF)	141
104:24	CSM CSI TWO	
104:30	IT (SXT/VHF) (V93,6/5,V93,35/36)	171
104:33	SUNSET	163
104:35	CT (SXT)	
105:11	CT (VHF)	
105:20	CSM CSI THREE	
105:21	SUNRISE	
105:24	IT (SXT/VHF) (4/3,V67,00057,00034,00001;8/9)	77
105:29		87
105:33		98
105:36	CT (SXT/VHF)	101
105:48	CSM PLANE CHANGE	
105:54	IT (SXT/VHF) (4/3,V93,6/7)	84
105:59		73
106:04	CT (SXT/VHF)	62
106:16	CSM CDH	
106:24	IT (SXT/VHF) (4/3,V93,11/12)	4
106:26	SUNSET	9
106:39	CT (SXT/VHF)	
106:51	CSM TPI	
106:54	IT (SXT/VHF) (V93,8/8)	
107:02	CT (SXT/VHF)	
107:06	CSM MCC1	
107:08	IT (SXT/VHF) (V93,9/9)	
107:14	SUNRISE	96
107:17	CT (SXT/VHF)	96
107:21	CSM MCC2	
107:35	TPF	

9.4.3 CSM ATTITUDE SUMMARY

**

INERTIAL AND ORDEAL FDAI BALL GIMBAL ANGLE PROFILE
FOR THE LM PARTIAL PHASING < 40 FPS RESCUE CASE
(MISSION F) (YAW = 0.0)

**

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
100:46	180	307	193	
101:19	180	150	283	98
101:33	180	180	283	
101:36	180	322	56	133
102:19	180	165	103	174
102:31	180	180	103	
102:51	180	320	177	74
104:24	180	335	257	
104:26	180	317	231	26
105:20	180	18	121	
105:22	180	18	120	
105:48	180	135	149	
105:50	180	132	139	10
106:04	0	154	110	
106:08	0	166	271	161
106:16	0	349	271	
106:18	0	228	144	127
106:44	0	247	74	60
106:51	0	202	14	
106:53	0	247	52	38
107:06	0	267	33	
107:21	0	300	23	
107:23	0	265	348	35
107:35	0	298	348	

9.5 LM ZERO INSERTION

9.5.1 Summary Timeline

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LM INSERTION EQUALS 0 FPS RESCUE TIMELINE

GET	EVENT	PROG	GET	EVENT	PROG
(102+46+18)	***** CSM INSERTION BURN(175.4,0,-107.1) (180,32/272,0)		(103+26)	MOVE TO CMD SEAT CSI TARGETING FINAL COMP OVERWRITE N81 WITH(=)CSM YDOT	(P32) (P20)
	***** SUNSET			SPS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE (5 DEG)	(P40)
(102+48)	IMU REALIGN TO REFSMMAT (OPTION 3)	(P52)		(180,156/293,0)	
(102+53)	MCC-H UPLINK(LM VECTOR) MOVE TO LEB	(P00)	*****		
(102+56)	RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (103 DEG) (180,165/16,0)	(P20)	*-6	F 50 18 (COMMANDER R,P,Y) KEY V56E RHC-ARMED ALIGN ROLL TO 180 DEG RHC-LOCKED	
(103+01)	CSI TARGETING PROGRAM LOAD N55 R1=+00001 R2=+20830 R3=+13000	(P32) (P20)	*	PRO 06 18 (COMMANDER R,P,Y) MONITOR MNVR	
-29	(103+03)	CALL MARKING ROUTINE(V93,V87,V57) (V93 AFTER 5 MKS PROCESSED)	*	F 50 18 (COMMANDER R,P,Y) MN BUS TIE(2)=ON(UP) SRS HE VLV TB(BOTH)=BP SPS HE VLV(BOTH)=AUTO NONESS BUS=MNA	
	(103+08)	TERMINATE SXT,CONT,VHF V90(CSM) COPY CSM YDOT	*	PRO 06 18 (COMMANDER R,P,Y) MONITOR ATT TRIM	
-9	(103+23)	TERMINATE MARKS(20)	*	F 50 18 (COMMANDER R,P,Y)	

GET	EVENT	PROG	GET	EVENT	PROG
(103+25)		*		SET SPS GIMBALS TW(2)=	*
-5	LOS	*		PITCH = -.52	*
		*		YAW = +.59	*
-5	(103+27)	*		RHC-VERIFY MTVC	*
	SPS THRUST SETUP	*		THC-NEUTRAL	*
-5	(103+27)	*		RHC PWR NORM 2-AC/DC	*
*	**GDC ALIGN**	*		RHC-LOCKED	*
*		*		PRO	*
*		*	06 18	(COMMENDED R,P,Y)	*
*	ALT SET KNOB TO 60 NM	*		MONITOR ATT TRIM	*
*	FDAI SELECT-1	*	F 50 18	(COMMENDED R,P,Y)	*
*	ATT SET THUMBWHEELS TO N18	*		KEY ENTER	*
*	NULL ATT ERROR NEEDLES	*	F 50 25	(00204,GMBL DRIVE TEST)	*
*	ON FDAI 1 WITH AIT	*		RHC PWR DIRECT (BOTH)=MNA/MNB	*
*	SET THUMBWHEELS	*		RATE=HIGH	*
*	ATT SET=GDC	*		AUTO RCS SEL A/C ROLL (4)=MNA	*
*	DEPRESS GDC ALIGN PB	*		PRO	*
*	FDAI SELECT-1/2	*		MONITOR GMBL DRIVE	*
*	ATT SET-TMU	*		SEQ AND TRIM	*
*	**MTVC CHECKS**	*	06 40	(TFI,VG,DVM)	*
*		-02+00			*
*	BMAG MODE (3)=ATT1/RATE2	*		FDAI SCALE=5/5	*
*	TV C SERVO PWR 1=AC1/MNA	*		VERIFY SPS TH LT-OFF	*
*	TV C SERVO PWR 2=AC2/MNB	*		EMS MODE STBY	*
*	TRANS CONTR PWR-ON	*		EMS FUNCTION=DV SET	*
*	RHC PWR NORM 2=AC	*		LOAD CSI VC	*
*	GMBL MTRS PITCH 1 STRT-ON	*		EMS FUNCTION=DV	*
*	GMBL MTRS YAW 1 STRT-ON	*		THC-ARMED	*
*	THC=CLOCKWISE	*		RHC-ARMED	*
*	RHC-ARMED	*		DV THRUST (A)=NORMAL	*
*	RHC-VERIFY NO MTVC	*			*
*	GMBL MTRS PITCH 2=STRT-ON	*		DSKY BLANKS	*
*	GMBL MTRS YAW 2=STRT-ON	*			*
*		-00+35			*
*		*			*
*		-00+30			*
*		*			*
*			06 40	(TFI,VG,DVM)	*

MISSION F RESCUE TIMELINE APRIL 25, 1969

GET	EVENT	PROG	GET	EVENT	PROG
*	EMS MODE=NORMAL	**		GMBL MTRS=OFF (SEQUENTIALLY)	*
**00+15	PERFORM ULLAGE	**		TVC SERVO PWR(BOTH)=OFF	*
**00+05	F 99 40 (REQUEST FOR ENGINE ENABLE)	**		FDAI SCALE=5/1	*
*	PRO	**		RATE=LOW	*
*	06 40 (TFC,VG,DVM)	**		ROT CONT PWR DIRECT(BOTH)=OFF	*
0 (103+32+19)	*****	**	F 16 85	PRO (VG-BODY)	*
	*****	**		THC=NULL, VG COMPONENTS	*
	CSM CSI BURN(-39,0,0,0)	**		THC=LOCKED	*
	(180,180/293,0)	**		RHC=LOCKED	*
	*****	**		EMS FUNCTION=VHF RNG	*
	SUNRISE	**		EMS MODE=VHF RNG	*
	*****	**		VHF RNG=RESET	*
	SET MDC ET=RESET,START	**		MN BUS TIE(2)=OFF	*
	MONITOR	**		PRO	*
	SPS THRUST LT=ON	**	F 37 88	*****	*
	DV INDICATOR=DECREASING	**	(103+34)	RENDEZVOUS NAV PROGRAM (P20)	
**00+01	ULLAGE=OFF	**		AUTO MNVR TO SXT TRACK	
	06 40 (TFC,VG,DVM)	**		(27 DEG)	
	MONITOR SPS OPERATION	**		(180,158/266,0)	
	PC INDICATOR=95-105 PSI	** -24	(103+40)	MOVE TO LEB	
	MONITOR SPS ENGINE CUTOFF	**		CALL MARKING ROUTINE(V87,V57)	
	SPS THRUST LT=OFF	**		AFTER 3 MARKS PROCESSED	
F 16 40 (TFC,VG,DVM)	DV THRUST(BOTH)=OFF	** -12	(103+52)	V67,LOAD WR(+00057,+00034,+00001)	
	MONITOR	**		TERMINATE MARKS(12)	
	PC INDICATOR = 0	**		OUT OF PLANE DATA(V90)	
	SPS INJ VLV IND(4)=CLOSE	**		EXTERNAL DV PROGRAM (P30) (P20)	
	SPS HE VLV TB(BOTH)=BP	**		LOAD OUT OF PLANE DATA	
	SPS GMBLS RETURN TO SERVO	**			
	NULL	**			

GET	EVENT	PROG	GET	EVENT	PROG
-8	(103+56) RCS THRUST PROGRAM,V56 BYPASS BURN ATTITUDE MNVR RCS THRUST SETUP	(P41)	-9	(104+28) SPS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE (98 DEG) (0,245/231,0)	(P40)
	(104+04+00) ***** CSM PLANE CHANGE(180,173/205,0) *****	**	-9	(104+28)	*
	(104+05) RENDEZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (25 DEG) (180,147/180,0)	(P20)	*	E 50 18 (COMMANDED R,P,Y) KEY 40E KEY V56E RHC=ARMED ALIGN ROLL TO 0 DEG, RHC=LOCKED PRO	*
-30	(104+07) CDH TARGETING PROGRAM (P33) (P20)	*	*	06 18 (COMMANDED R,P,Y)	*
-28	(104+09) CALL MARKING ROUTINE(V87,V57) (V93 AFTER 3 SXT MARKS PROCESSED)	*	*	MONITOR MNVR F 50 18 (COMMANDED R,P,Y) MN BUS TIE(2)=ON(UP)	*
	(104+12) AOS	*	*	SPS HE VLV TB(BOTH)=BP SPS HE VLV(BOTH)=AUTO NONESS BUS=MNA PRO	*
-13	(104+24) TERMINATE MARKS(15) MOVE TO CMD SEAT ROLL 180 DEG,ACQ MSFN (0,155/133,0) (RHO -35, GAM 0) CDH TARGETING FINAL (P33) (P20) COMP OUT OF PLANE DATA(V90) OVERWRITE N81 WITH(=)CSM YDOT	*	*	06 18 (COMMANDED R,P,Y) MONITOR ATT TRIM F 50 18 (COMMANDED R,P,Y) *****	*
		-5	(104+32)	SPS THRUST SETUP *****	(P40)
		*	-5	(104+32) **GDC ALIGN**	*
		*	*		*
		*		ALT SET KNOB TO 60 NM	*

GET	EVENT	PROG	GET	EVENT	PROG
*	FDAI SELECT-1	*	F 50 18	(COMMANDED R,P,Y)	*
*	ATT SET THUMBWHEELS TO N18	*	KEY ENTER	*	*
*	NULL ATT ERROR NEEDLES	*	F 50 25	(00204,GMBL DRIVE TEST)	*
*	ON FDAI 1 WITH ATT	*	RHC PWR DIRECT (BOTH)-MNA/MNB	*	*
*	SET THUMBWHEELS	*	RATE-HIGH	*	*
*	ATT SET-GDC	*	AUTO RCS SEL A/C ROLL (4)-MNA	*	*
*	DEPRESS GDC ALIGN PB	*	PRO	*	*
*	FDAI SELECT-1/2	*	MONITOR GMBL DRIVE	*	*
*	ATT SET-TMU	*	SEQ AND TRIM	*	*
*	**MTVC CHECKS**	*.02+00	06 40	(TFI,VG,DVM)	*
*	B MAG MODE (3)-ATT1/RATE2	*	FDAI SCALE-5/5	*	*
*	TVC SERVO PWR 1-AC1/MNA	*	VERIFY SPS TH LT-OFF	*	*
*	TVC SERVO PWR 2-AC2/MNB	*	EMS MODE STBY	*	*
*	TRANS CONTR PWR-ON	*	EMS FUNCTION=DV SET	*	*
*	RHC PWR NORM 2-AC	*	LOAD COH VC	*	*
*	GMBL MTRS PITCH 1 STRT-ON	*	EMS FUNCTION=DV	*	*
*	GMBL MTRS YAW 1 STRT-ON	*	THC-ARMED	*	*
*	THC-CLOCKWISE	*	RHC-ARMED	*	*
*	RHC-ARMED	*.00+35	DV THRUST (A)-NORMAL	*	*
*	RHC-VERIFY NO MTVC	*	DSKY BLANKS	*	*
*	GMBL MTRS PITCH 2-STRT-ON	*.00+30	06 40 (TFI,VG,DVM)	*	*
*	GMBL MTRS YAW 2-STRT-ON	*	EMS MODE=NORMAL	*	*
*	SET SPS GIMBALS TW(2)-	*	PERFORM ULLAGE	*	*
*	PITCH = -.52	*.00+15	F 99 40 (REQUEST FOR ENGINE ENABLE)	*	*
*	YAW = +.59	*	PRO	*	*
*	RHC-VERIFY MTVC	*.00+05	06 40 (TFI,VG,DVM)	*	*
*	THC-NEUTRAL	*	*****	*	*
*	RHC PWR NORM 2-AC/DC	*		*	*
*	RHC-LOCKED	*		*	*
*	PRO	*		*	*
*	06 18 (COMMANDED R,P,Y)	*		*	*
*	MONITOR ATT TRIM	*		*	*

GET	EVENT	PROG	GET	EVENT	PROG
0	(104+36+43)	*		RHC-LOCKED	*
	*****	*		EMS FUNCTION-VHF RNG	*
	CSM CDH BURN(-.8,0,42.6)	*		EMS MODE-VHF RNG	*
	(0,269/213,0)	*		VHF RNG-RESET	*
	*****	*		MN BUS TIE(2)-OFF	*
	*****	*		PRO	*
*	SET MDC ET-RESET,START	*	F 37 BB		*
*	MONITOR	*		RENDEZVOUS NAV PROGRAM (P20)	
*	SPS THRUST LT-ON	*		AUTO MNVR TO SXT TRACK	
*	DV INDICATOR-DECREASING	*		(70 DEG)	
**00+01	ULLAGE-OFF	*		(0,224/160,0)	
*	06 40 (TFC,VG,DVM)	*		TPI TARGETING PROGRAM (P34) (P20)	
*	MONITOR SPS OPERATION	*		LOAD N55 R2=+20830	
*	PC INDICATOR=95-105 PSI	*		R3=+13000	
*	MONITOR SPS ENGINE CUTOFF	*		CALL MARKING ROUTINE(V87,V57)	
*	SPS THRUST LT-OFF	*		(V93 AFTER 3 SXT MARKS PROCESSED)	
F 16 40	(TFC,VG,DVM)	*	(104+53)	SUNSET	
*	DV THRUST(BOTH)-OFF	*		TERMINATE MARKS(18)	
*	MONITOR	*		MOVE TO CMD SEAT	
*	PC INDICATOR = 0	*		TPI TARGETING FINAL (P34) (P20)	
*	SPS INJ VLV IND(4)-CLOSE	*		COMP	
*	SPS HE VLV TB(BOTH)-BP	*		VERIFY ORDEAL(V83)	
*	SPS GMBLS RETURN TO SERVO	*		SPS THRUST PROGRAM,V56 (P40)	
*	NULL	*		AUTO MNVR TO BURN ATTITUDE	
*	GMBL MTRS-OFF(SEQUENTIALLY)	*		(95 DEG)	
*	TVC SERVO PWR(BOTH)-OFF	*		(0,165/353,0)	
*	FDAI SCALE=5/1	*			
*	RATE-LOW	*			
*	ROT CONT PWR DIRECT(BOTH)-OFF	*			
*	PRO	*			
F 16 85	(VG-BODY)	*			
*	THC-NULL VG COMPONENTS	*			
*	THC-LOCKED	*			

GET	EVENT	PROG	GET	EVENT	PROG
***8		*		ATT SET=GDC	*
	KEY 40E	*		DEPRESS GDC ALIGN PB	*
	F 50 18 (COMMENDED R,P,Y)	*		FDAI SELECT-1/2	*
	KEY V56E	*		ATT SET=TMU	*
	RHC-ARMED	*		***MTVC CHECKS**	*
	ALIGN ROLL TO 0 DEG	*		BMAG MODE(3)=ATT1/RATE2	*
	RHC-LOCKED	*		TV _C SERVO PWR 1=AC1/MNA	*
	PRO	*		TV _C SERVO PWR 2=AC2/MNB	*
	06 18 (COMMENDED R,P,Y)	*		TRANS CONTR PWR=ON	*
	MONITOR MNVR	*		RHC PWR NORM 2=AC	*
	F 50 18 (COMMENDED R,P,Y)	*		GMBL MTRS PITCH 1 STRT=ON	*
	MN BUS TIE(2)=ON(UP)	*		GMBL MTRS YAW 1 STRT=ON	*
	SPS HE VLV TB(BOTH)=BP	*		THC=CLOCKWISE	*
	SPS HE VLV(BOTH)=AUTO	*		RHC=ARMED	*
	NONESS BUS=MNA	*		RHC=VERIFY NO MTVC	*
	PRO	*		GMBL MTRS PITCH 2=STRT=ON	*
	06 18 (COMMENDED R,P,Y)	*		GMBL MTRS YAW 2=STRT=ON	*
	MONITOR ATT TRIM	*		SET SPS GIMBALS TW(2)=	*
	F 50 18 (COMMENDED R,P,Y)	*		PITCH = -.52	*
***5		*		YAW = +.59	*
	SPS THRUST SETUP	*		RHC=VERIFY MTVC	*
***5		*		THC=NEUTRAL	*
	GDC ALIGN	*		RHC PWR NORM 2=AC/DC	*
		*		RHC=LOCKED	*
		*		PRO	*
	ALT SET KNOB TO 60 NM	*		06 18 (COMMENDED R,P,Y)	*
	FDAI SELECT-1	*		MONITOR ATT TRIM	*
	ATT SET THUMBWHEELS TO N18	*		F 50 18 (COMMENDED R,P,Y)	*
	NULL ATT ERROR NEEDLES	*		KEY ENTER	*
	ON FDAI 1 WITH ATT	*		F 50 25 (00204, GMBL DRIVE TEST)	*
	SET THUMBWHEELS	*		RHC PWR DIRECT(BOTH)=MNA/MNB	*
		*		RATE=HIGH	*

GET	EVENT	PROG	GET	EVENT	PROG
*	AUTO RCS SEL A/C ROLL(4)=MNA	*	*	SET MDC ET=RESET,START	*
*	PRO	*	*	MONITOR	*
*	MONITOR GMBL DRIVE	*	*	SPS THRUST LT=ON	*
*	SEQ AND TRIM	*	*	DV INDICATOR=DECREASING	*
*	06 40 (TFI,VG,DVM)	*	*		*
*-02+00		* *00+01			
*	FDAI SCALE=5/5	*		ULLAGE=OFF	*
*	VERIFY SPS TH LT=OFF	*		06 40 (TFC,VG,DVM)	*
*	EMS MODE STBY	*		MONITOR SPS OPERATION	*
*	EMS FUNCTION=DV SET	*		PC INDICATOR=95-105 PSI	*
*	LOAD TPI VC	*		MONITOR SPS ENGINE CUTOFF	*
*	EMS FUNCTION=DV	*		SPS THRUST LT=OFF	*
*	THC=ARMED	*			*
*	RHC=ARMED	*		F 16 40 (TFC,VG,DVM)	*
*	DV THRUST(A)=NORMAL	*		DV THRUST(BOTH)=OFF	*
*-00+35	DSKY BLANKS	*		MONITOR	*
*-00+30	06 40 (TFI,VG,DVM)	*		PC INDICATOR = 0	*
*	EMS MODE=NORMAL	*		SPS INJ VLV IND(4)=CLOSE	*
*-00+15	PERFORM ULLAGE	*		SPS HE VLV TB(BOTH)=BP	*
*-00+05	F 99 40 (REQUEST FOR ENGINE ENABLE)	*		SPS GMBL RETURN TO SERVO	*
*	PRO	*		NULL	*
*	06 40 (TFI,VG,DVM)	*		GMBL MTRS=OFF (SEQUENTIALLY)	*
*****	(105+16+47)	*		TVC SERVO PWR(BOTH)=OFF	*
*	CSM TPI BURN(-21.1,0,-.8)	*		FDAI SCALE=5/1	*
*	(0,180/353,0)	*		RATE=LOW	*
*****		*		ROT CONT PWR DIRECT(BOTH)=OFF	*
		*		PRO	*
		*		F 16 85 (VG-BODY)	*
		*		THC=NULL VG COMPONENTS	*
		*		THC=LOCKED	*
		*		RHC=LOCKED	*
		*		EMS FUNCTION=VHF RNG	*

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GET	EVENT	PROG	GET	EVENT	PROG
	EMS MODE-VHF RNG	*		(105+40)	
	VHF RNG=RESET	*		SUNRISE	
	MN BUS TIE(2)=OFF	* 26		TERMINATE MARKS(9)	
	PRO	*		MCC TARGETING FINAL	(P35) (P20)
	F 37 BB	* 27		COMP	
+2	RENDEZVOUS NAV PROGRAM	(P20)	+28.5	RCS THRUST PROGRAM	(P41)
	AUTO MNVR TO SXT TRACK			BYPASS BURN ATTITUDE MNVR	
	(61 DEG)			RCS THRUST SETUP	(P41)
	(0,246/54,0)			MCC2 BURN(0,286/23,0)	
	MCC TARGETING PROGRAM	(P35) (P20)	+30	(105+46+47)	
	MOVE TO LEB DURING AUTO MNVR			*****	
+4	CALL MARKING ROUTINE(V93,V87,V57)			*****	
+11	TERMINATE MARKS(7)		+32	AUTO MNVR TO COAS TRACK(V89) (P00)	
+12	MCC TARGETING FINAL	(P35) (P20)		MOVE TO CMD SEAT	
	COMP			(37 DEG)	
+13.5	RCS THRUST PROGRAM	(P41)		(0,254/346,0)	
	BYPASS BURN ATTITUDE MNVR			CALL P47 AT R=1.25N.M.	
	RCS THRUST SETUP	(P41)		THRUST MONITOR PROGRAM	
+15	(105+31+47)			V83, PERFORM BRAKING AND	(P47)
	*****			LOS CONTROL	
	MCC1 BURN(0,265/37,0)			*****	
	*****			*****	
	LOS			* BRAKING GATES RETICLE ANGLE *	
+16	MCC TARGETING PROGRAM	(P35) (P20)		* 30 FPS AT 6000 FT 13 DEG.	
	POSSIBLE MNVR TO SXT TRACK			* 20 FPS AT 3000 FT 26 DEG.	
+17	CALL MARKING ROUTINE(V93,V87,V57)			* 10 FPS AT 2500 FT 54 DEG.	
				* 5 FPS AT 500 FT 1.60 DEG.	

GET EVENT

*51 (106*07*47)

TPF

MISSION F RESCUE TIMELINE APRIL 25, 1969

9.5.2 NAVIGATION SUMMARY
LM INSERTION = 0 FPS

P20 NAVIGATION SUMMARY WITH SUN ANGLES
IT - INITIATE TRACK
CT - CEASE TRACK
(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE (LOS TO SUN) DEG
102:46	CSM INSERTION BURN SUNSET	
103:03	IT (SXT/VHF) (V93;6/5;V93;14/15)	
103:08	CT (SXT)	
103:23	CT (VHF)	
103:32	CSM CSI BURN SUNRISE	
103:40	IT (SXT/VHF) (4/3;V67,00057,00034,00001;8/9)	158
103:45		172
103:50		174
103:52	CT (SXT/VHF)	168
104:04	CSM PLANE CHANGE	
104:09	IT (SXT/VHF) (4/3,V93,11/12)	123
104:14		109
104:19		97
104:24	CT (SXT/VHF)	83
104:37	CSM CDH BURN	
104:47	IT (SXT/VHF) (4/3,V93,14/15)	19
104:52		1
104:53	SUNSET	1
105:05	CT (SXT/VHF)	
105:17	CSM TPI BURN	
105:21	IT (SXT/VHF) (V93,7/7)	
105:28	CT (SXT/VHF)	
105:32	CSM MCC1 BURN	
105:33	IT (SXT/VHF) (V93,9/9)	
105:40	SUNRISE	93
105:42	CT (SXT/VHF)	94
106:02	CSM MCC2 BURN	
106:08	TPF	

9.5.3 CSM ATTITUDE SUMMARY
INERTIAL AND ORDEAL FDAI BALL GIMBAL ANGLE
PROFILE FOR LM INSERTION = 0 RESCUE CASE
(MISSION F) (YAW = 0.0)

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
102:46	180	32	272	
102:56	180	165	16	103
103:26	180	156	293	5
103:32	180	180	293	
103:34	180	158	266	27
104:04	180	173	205	
104:05	180	147	180	25
104:26	0	155	133	
104:28	0	245	231	98
104:37	0	269	231	
104:40	0	244	160	70
105:09	0	165	353	95
105:16	0	180	353	
105:19	0	246	54	61
105:32	0	265	37	
106:02	0	286	23	
106:04	0	254	346	37

9.6 LM PARTIAL INSERTION

9.6.3 SUMMARY TIMELINE

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GET	EVENT	PROG	GET	EVENT	PROG
	(LM INSERTION=120 FPS)		*****	*****	*****
			*-7	(103+25)	
(102+43+18)	*****	*		KEY 40E	*
	LM INSERTION BURN (-106.8,0+-54.5)	*	F 50 1A	(COMMANDED R,P,V)	*
	*****	*		KEY V56E	*
(102+47)	SUNSET	*		RHC-ARMED	*
	MOVE TO LER	*		ALIGN ROLL TO	*
	IMU REALIGN TO REFSMMAT	*		RHC-LOCKED	*
	(OPTION 3)	(P52)		PRO	*
(102+52)	MCC-H UPLINK (LM STATE	*	06 1A	(COMMANDED R,P,V)	*
	VECTOR)	(P00)		MONITOR MNVR	*
-29	(103+03)	COPY RESCUE,CSI ONE,CDH,TPI PADS	*	F 50 1A	(COMMANDED R,P,V)
				MN BUS TIE(2)=ON(UP)	*
				SPS HE VLV TB(BOTH)=BP	*
				SPS HE VLV(BOTH)=AUTO	*
				NONESS BUS-MNA	*
				PRO	*
(103+17)	LOS	*	06 1A	(COMMANDED R,P,V)	*
	MOVE TO CMD SEAT	*		MONITOR ATT TRIM	*
-12	(103+20)	EXTERNAL UV PROGRAM	(P30)	F 50 1A	(COMMANDED R,P,V)
-7	(103+25)	SPS THRUST PROGRAM,V56	(P40)	*****	*****
	AIUTO MNVR TO BURN ATTITUDE			SPS THRUST SETUP	(P40)
	(19 DEG)			*****	*****
	(180,165/282,0)		*-5	(103+27)	
				SPS THRUST SETUP	(P40)
				*****	*****
				GDC ALIGN	*
				*****	*
				ALT SET KNOB TO 60 NM	*
				FDAI SELECT=1	*

GET EVENT

ATT SET THUMBWHEELS TO #18 * *
NULL ATT ERROR NEEDLES * *
ON FDAI 1 WITH ATT * *
SET THUMBWHEELS * *
ATT SET-GDC * *
DEPRESS GDC ALIGN PB. * *
FDIAI SELECT-1/2 * *
ATT SET-TMU * *
* *
MTVC CHECKS * *-02+00

BMAG MODE(3)-ATT1/RATE? * *
TVC SERVO PWR 1-AC1/MNA * *
TVC SERVO PWR 2-AC2/MNR * *
TRANS CONTR PWR-ON * *
RHC PWR NORM 2-AC * *
GMBL MTRS PITCH 1 STRT-ON * *
GMBL MTRS YAW 1 STRT-ON * *
THC-CLOCKWISE * *
RHC-ARMED * *-00+35
RHC-VERIFY NO MTVC * *
GMBL MTRS PITCH 2-STR-ON * *-00+30
GMBL MTRS YAW 2-STR-ON * *
SET SPS GIMBALS TW(2)-
PITCH = -.52 * *-00+15
YAW = +.50 * *
RHC-VERIFY MTVC * *-00+05
THC-NEUTRAL * *
RHC PWR NORM 2-AC/DC * *
RHC-LOCKED * *
PRO * *
06 18 (COMMANDER R,P,Y) * *
MONITOR ATT TRIM * *
F 50 18 (COMMANDER R,P,Y) * *

GET EVENT

PROG

KEY FNTFR * *
F 50 25 (00204, GMBL DRIVE TEST) * *
RHC PWR DIRECT(BOTH)-MNA/MNB * *
RATE-HIGH * *
AUTO RCS SEL A/C ROLL(4)-MNA * *
PRO * *
MONITOR GMBL DRIVE * *
SEQ AND TRIM * *
06 40 (TFI, VG, DVM) * *
* *
FDIAI SCALE-5/5 * *
VERIFY SPS TH LT-OFF * *
EMS MODE STBY * *
EMS FUNCTION-DV SET * *
LOAD RESCUE VC * *
EMS FUNCTION-DV * *
THC-ARMED * *
RHC-ARMED * *
DV THRUST(A)-NORMAL * *
DSKY BLANKS * *
06 40 (TFI, VG, DVM) * *
EMS MODE-NORMAL * *
* *
PERFORM ULLAGE * *
* *
F 99 40 (REQUEST FOR ENGINE ENABLE) * *
PRO * *
06 40 (TFI, VG, DVM) * *

GET	EVENT	PROG	GET	EVENT	PROG
0	(103+31+59)	*	F 16 85	(VG=READY)	*
	*****	*	THC=NULL, VG COMPONENTS	*	
	CSM RESCUE BURN (-44.9,0,0),	*	THC=LOCKED	*	
	(180,180/282,0)	*	RHC=LOCKED	*	
	*****	*	EMS FUNCTION-VHF RNG	*	
	SET MUC ET-RESFT,START	*	EMS MODE-VHF RNG	*	
	MONITOR:	*	VHF RNG-RESET	*	
	SPS THRUST LT-ON	*	MN BUS TIE(2)-OFF	*	
	DV INDICATOR-DECREASING	*	PRO	*	
+00+01	ULLAGE-OFF	*	F 37 BB	*	
	06 40 (TFC,VG,DVM)	*	*****	*	
	MONITOR SPS OPERATION	*	(103+33)	*	
	PC INDICATOR-95-105 PSI	-55	SUNRISE		
	MONITOR SPS ENGINE CUTOFF	*	RENDEZVOUS NAV PROGRAM (P20)		
	SPS THRUST LT-OFF	*	AUTO MNVR TO SXT TRACK		
	F 16 40 (TFC,VG,DVM)	*	(49 DEG)		
	DV THRUST(BOTH)-OFF	*	(180,141/233,0)		
	MONITOR:	*	MOVE TO LEB		
	PC INDICATOR = 0	-52	(103+38)		
	SPS INJ VLV TND(4)-CLOSE	*	CSI TARGETING PROGRAM (P32)		
	SPS HE VLV TB(BOTH)-RP	*	LOAD N55 WITH		
	SPS GMBLIS RETURN TO SERVO	*	R1=+00002		
	NULL	*	R2=+20830		
	GMBL MTRS-OFF(SEQUENTIALLY)	-35	R3=+13000		
	TVC SERVO PWR(BOTH)-OFF	*	CALL MARKING ROUTINE(V93,V57 SXT		
	FDAI SCALE-5/1	*	ONLY)		
	RATE-LOW	*	AOS		
	ROT CONT PWR DIRECT(BOTH)-OFF*	*			
	PRO	*			

GET	EVENT	PROG	GET	EVENT	PROG
-15	(104+15)		-5	(104+25)	
	TERMINATE MARKS(20)			SPS THRUST SETUP	(P40)
	MOVE TO CMD SEAT			*****	*****
	CST TARGETING PROGRAM	(P32)	-5	(104+25)	*
	FTNAL COMP	*	*	##GDC ALIGN##	*
	OUT OF PLANE DATA(V90)	*			*
	OVERWRITE N81 WITH (-) CSM YDOT	*			*
-11	(104+19)		*		
	SPS THRUST PROGRAM,V56	(P40)	*	ALT SET KNOB TO 60 NM	*
	AUTO MNVR TO BURN ATTITUDE	*	*	FDAI SELECT-1	*
	(175 DEG)	*	*	ATT SET THUMBWHEELS TO N18	*
	(180,345/281,0)	*	*	NULL ATT ERROR NEEDLES	*
	ACQ, MSFN, R40=-48, GAMMA=180	*	*	ON FDAI 1 WITH ATT	*
	*****	*	*	SET THUMBWHEELS	*
-11	(104+19)		*	ATT SET-GDC	*
	KEY 40E	*	*	DEPRESS GDC ALIGN PB	*
	F 50 18 (COMMENDED R,P,Y)	*	*	FDAI SELECT-1/2	*
	KEY V56E	*	*	ATT SET-TMU	*
	RHC-ARMED	*		*****	*
	ALIGN ROLL TO	*		BMAG MODE(3)-ATT1/RATE2	*
	RHC-LOCKED	*		TVC SERVO PWR 1-AC1/MNA	*
	PRO	*		TVC SERVO PWR 2-AC2/MNB	*
	06 18 (COMMENDED R,P,Y)	*		TRANS CONTR PWR-ON	*
	MONITOR MNVR	*		RHC PWR NORM 2-AC	*
	F 50 18 (COMMENDED R,P,Y)	*		GMBL MTRS PITCH 1 STRT-ON	*
	MN.BUS TIE(2)-ON(UP)	*		GMBL MTRS YAW 1 STRT-ON	*
	SPS HE VLV T8(BOTH)-RP	*		THC-CLOCKWISE	*
	SPS HE VLV(BOTH)-AUTO	*		RHC-ARMED	*
	NONESS BUS-MNA	*		RHC-VERIFY NO MTC	*
	PRO	*		GMBL MTRS PITCH 2-STR-ON	*
	06 18 (COMMENDED R,P,Y)	*		GMBL MTRS YAW 2-STR-ON	*
	MONITOR ATT TRIM	*		*****	*
	F 50 18 (COMMENDED R,P,Y)	*			*
	*****	*			*

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GET	EVENT	PROG	GET	EVENT	PROG
	SET SPS GIMBALS, TW(2)-	* *-00+30			*
	PITCH = -.52	* *		06 40 (TFI, VG, DVM)	*
	YAW = +.59	* *		EMS MODE=NORMAL	*
	RHC=VERIFY MTVC	* *-00+15			*
	THC=NEUTRAL	* *		PERFORM ULLAGE	*
	RHC PWR NORM 2-AC/DC	* *-00+05			*
	RHC=LOCKED	* *		F 99 40 (REQUEST FOR ENGINE ENABLE)	*
	PRO	* *		PRO	*
06 1A	(COMMANDED R,P,Y)	* *		06 40 (TFI, VG, DVM)	*
	MONITOR ATT TRIM	* *****			*
F 50 1A	(COMMANDED R,P,Y)	* (104+29+59)			*
	KEY ENTER	*			*
F 50 25	(00204, GMBL DRIVE TEST)	*		CSM CSI ONE RURN(114,6,0,0)	*
	RHC PWR DIRECT(BOTH)=MNA/MNR	*		(180,0,281,0)	*
	RATE-HIGH	*			*
	AUTO RCS SEL A/C ROLL(4)=MNA	* *****		SET MDC ET=RESET, START	*
	PRO	* *		MONITOR	*
	MONITOR GMBL DRIVE	* *		SPS THRUST LT=ON	*
	SEQ AND TRIM	* *		DV INDICATOR=DECREASING	*
06 40	(TFI, VG, DVM)	* *+00+01			*
002+00				ULLAGE=OFF	*
	FDAI SCALE-5/5	* *	06 40	(TFC, VG, DVM)	*
	VERIFY SPS TH LT=OFF	* *		MONITOR SPS OPERATION	*
	EMS MODE STBY	* *		PC INDICATOR=95-105 PSI	*
	EMS FUNCTION=DV SET	* *		MONITOR SPS ENGINE CUTOFF	*
	LOAD CSI VC	* *		SPS THRUST LT=OFF	*
	EMS FUNCTION=DV	* *	F 16 40	(TFC, VG, DVM)	*
	THC=ARMED	* *		DV THRUST(BOTH)=OFF	*
	RHC=ARMED	* *		MONITOR	*
	DV THRUST(A)=NORMAL	* *		PC INDICATOR = 0	*
000+35	DSKY BLANKS	* *		SPS INJ VLV TND(4)=CLOSE	*
		* *		SPS HE VLV TB(BOTH)=BP	*

GET	EVENT	PROG	GET	EVENT	PROG
	SPS GMBLS RETURN TO SERVO #	-35	(104+57)		
	NULL	*		CST TARGETING PROGRAM	(P32)
	GMRL MTRS-OFF (SEQUENTIALLY)	*		LOAD N55 WITH	
	TVC SERVO PWR(BOTH)-OFF	*		R1=+00001	
	FDAI SCALE-5/1	*		R2=+20830	
	RATE-LOW	*		R3=+13000	
	ROT CONT PWR DIRECT(BOTH)-OFF*	-32	(105+00)		
	PRO	*		CALL MARKING ROUTINE(V93,V87,V57)	
F 16 85	(VG-BODY)	*		(V93 AFTER 5 MARKS PROCESSED)	
	THC-NULL VG COMPONENTS	*	(105+05)		
	THC-LOCKED	*		TERMINATE SXT MARKS V93	
	RHC-LOCKED	*		CONTINUE VHF	
	EMS FUNCTION-VHF RNG	*	(105+10)		
	EMS MODE-VHF RNG	*		OUT OF PLANE DATA(V90)	
	VHF RNG-RESET	*		OVERWRITE N81 WITH (-) CSM YDOT	
	MN BUS TIE(2)-OFF	*		CONTINUE VHF	
	PRO	*	(105+18)		
F 37 BR	*	*		LOS	
(104+34)	CMC IDLING PROGRAM	(P00)	(105+23)		
-50 (104+42)	SUNSET	-5	(105+27)	TERMINATE MARKS(23)	
	MOVE TO LER			MOVE TO CMD SEAT	
	IMU REALIGN TO REFSMMAT	(P52)		CSI TARGETING FINAL COMP	(P32)
	(OPTION 3)				
-45 (104+47)	MCC-H UPLINK(LM VECTOR) (IF LM		(105+29)	RCS THRUST PROGRAM,V56	(P41)
	ACTIVE)	(P00)		AUTO MNVR TO BURN ATTITUDE	
-40 (104+52)	RFNDEZVOUS NAV PROGRAM	(P20)	(105+30)	(180,165/276,0)	
	AUTO MNVR TO SXT TRACK			SUNRISE	
	(90 DEG)			RCS THRUST SETUP	(P41)
	(180,162/11,0)				

	GET	EVENT	PROG	-162-	GET	EVENT	PROG
	(105+31+55)	***** CSM CST TWO BURN(0, 0, 0) (180,171/276,0)		0	(106+03+00)	***** CSM PLANE CHGNE(0, 0, 0) (180,161/178,0)	
-28	(105+35)	OUT OF PLANE DATA(V90) MOVE TO LEB		-30	(106+04)	RENDÉZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (13 DEG) (180,154/165,0)	(P20)
-26	(105+37)	RENDÉZVOUS NAV PROGRAM AUTO MNVR TO SXT TRACK (27 DEG) (180,164/249,0)	(P20)		(106+06)	AOS	
-22	(105+41)	CALL MARKING ROUTINE(V87,V57) AFTER 3 SXT MARKS PROCESSED V67,LOAD WR(00057,00034,00001)		-27	(106+07)	CDH TARGETING PROGRAM MOVE TO LEB	(P33)
-12	(105+51)	TERMINATE MARKS(10) MOVE TO CMD SEAT OUT OF PLANE DATA V90		-25	(106+09)	CALL MARKING ROUTINE(V87,V57) (V93 AFTER 3 MARKS PROCESSED)	
-5	(105+58)	EXTERNAL DV TARGETTING OVERWRITE N81 WITH (-) CSM YDOT	(P30)	-10	(106+24)	TERMINATE MARKS(15) MOVE TO CMD SEAT	
-3	(106+00)	RCS THRUST PROGRAM,V56 BYPASS BURN ATTITUDE MNVR (180,155/178,0)	(P41)			CDH TARGETING FINAL COMP ROLL 180 DEG,ACQ MSFN (0,223/178,0) (RH0=-58,GMMA=173)	(P33)
-2	(106+01)	RCS THRUST SETUP	(P41)	-6	(106+27)	OUT OF PLANE DATA (V90) OVERWRITE N81 WITH (-) CSM YDOT SPS THRUST PROGRAM,V56 AUTO MNVR TO BURN ATTITUDE (3 DEG) (0,229/165,0)	(P40)

GET	EVENT	PROG
*****	*****	*****
0-6 (106+27)	KEY 40E	* *
	F 50 18 (COMMANDED R,P,Y)	* *
	KEY V5AE	* *
	RHC=ARMED	* *
	ALIGN ROLL 10	* *
	RHC=LOCKED	* *
	PRO	* *
06 18	(COMMANDED R,P,Y)	* *
	MONITOR MNVR	* *
F 50 18	(COMMANDED R,P,Y)	* *
	MN BUS TIE(2)=ON(UP)	* *
	SPS HE VLV TB(BOTH)=BP	* *
	SPS HE VLV(BOTH)=AUTO	* *
	NONESS BUS-MNA	* *
	PRO	* *
06 18	(COMMANDED R,P,Y)	* *
	MONITOR ATT TRIM	* *
F 50 18	(COMMANDED R,P,Y)	* *
*****	SPS THRUST SETUP (P40)	*
0-5 (106+29)	**GDC ALIGN**	* *
	ALT SET KNOB TO 60 NM	* *
	FDAI SELECT=1	* *
	ATT SET THUMBWHEELS TO N18	* *
		*
		*

GET	EVENT	PROG
*****	NULL ATT ERROR NEEDLES	*
	ON FDAI 1 WITH ATT	*
	SET THUMBWHEELS	*
	ATT SET-GDC	*
	DEPRESS GDC ALIGN PR	*
	FDAI SELECT=1/2	*
	ATT SET-TMU	*
		*
		*
	##MTVC CHECK\$##	*
	BMAG MODE(3)=ATT1/RATE2	*
	MTVC SERVO PWR 1=AC1/MNA	*
	MTVC SERVO PWR 2=AC2/MNB	*
	TRANS CONTR PWR ON	*
	RHC PWR NORM 2=AC	*
	GMBL MTRS PITCH 1 STRT=ON	*
	GMBL MTRS YAW 1 STRT=ON	*
	THC=CLOCKWISE	*
	RHC=ARMED	*
	RHC=VERIFY NO MTVC	*
	GMBL MTRS PITCH 2=STRT=ON	*
	GMBL MTRS YAW 2=STRT=ON	*
	SET SPS GIMBALS TW(2)=	*
	PITCH = -.52	*
	YAW = +.59	*
	RHC=VERIFY MTVC	*
	THC=NEUTRAL	*
	RHC PWR NORM 2=AC/DC	*
	RHC=LOCKED	*
	PRO	*
06 18	(COMMANDED R,P,Y)	*
	MONITOR ATT TRIM	*

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GET	EVENT	PROG	GET	EVENT	PROG
	F 50 1R (COMMANDER R,P,Y)	*	0	(106+33+49)	
	KEY ENTER	*		*****	
	F 50 25 (00204,GMHL DRIVE TEST)	*		CSM CUH BURN (-37.1,0,74.8),	
	RHC PWR DIRECT(BOTH)-MNA/MNR	*		(01244,165,0)	
	RATE-HIGH	*		*****	
	AUTO RCS SEL A/C ROLL (4)-MNA	*		*****	
	PRO	*		*****	
	MONITOR GMHL DRIVE	*		SET MDC ET-RESET,START	*
	SEQ AND TRIM	*		MONITOR	*
-02+00	06 40 (TFI,VG,DVM)	*		SPS THRUST LT-ON	*
	FDAI SCALE-5/5	*		DV INDICATOR-DECREASING	*
	VERIFY SPS TH LT-OFF	*		ULLAGE-OFF	*
	EMS MODE STBY	*		06 40 (TFC,VG,DVM)	*
	EMS FUNCTION-DV SET	*		MONITOR SPS OPERATION	*
	LOAD CDH VC	*		PC INDICATOR 95-105 PSI	*
	EMS FUNCTION-DV	*		MONITOR SPS ENGINE CUTOFF	*
	THC-ARMED	*		SPS THRUST LT-OFF	*
	RHC-ARMED	*		F 16 40 (TFC,VG,DVM)	*
	DV THRUST (A)-NORMAL	*		DV THRUST(BOTH)-OFF	*
-00+35	DSKY BLANKS	*		MONITOR	*
-00+30		*		PC INDICATOR 0	*
	06 40 (TFI,VG,DVM)	*		SPS INU VLV IND(4)-CLOSE	*
	EMS MODE-NORMAL	*		SPS HE VLV TB(BOTH)-OFF	*
-00+15		*		SPS GMBS RETURN TO SERVO	*
	PERFORM ULLAGE	*		NULL	*
-00+05		*		GMBS MTRS-OFF(SEQUENTIALLY)	*
	F 99 40 (REQUEST FOR ENGINE ENABLE)	*		TVC SERVO PWR(BOTH)-OFF	*
	PRO	*		FDAI SCALE-5/1	*
	06 40 (TFI,VG,DVM)	*		RATE-LOW	*
	*****	*		ROT CONT PWR DIRECT(BOTH)-OFF	*
		*		PRO	*

GET	EVENT	-165-	PROG	GET	EVENT	PROG
	F 16 R5 (VG-BODY)	* -7	(107+02)		SPS THRUST PROGRAM,V56	(P40)
	THC=NULL VG COMPONENTS	*			AUTO MNVR TO BURN ATTITUDE	
	THC-LOCKED	*			(56 DEG)	
	RHC-LOCKED	*			(0,174/10,0)	
	EMS FUNCTION-VHF RNG	*			*****	
	EMS MODE=VHF RNG	*			*****	
	VHF RNG-RESET	*	-7	(107+02)	*****	
	MN BUS TIE(2)-OFF	*	*		KEY 40E	
	PRO	*	*	F 50 18	(COMMANDED R,P,V)	
	F 37 RB	*	*		KEY V56E	
133	(106+36)	*****	*		RHC=ARMED	
	RENDEZVOUS NAV PROGRAM	(P20)	*		ALIGN ROLL TO	
	AUTO MNVR TO SXT TRACK		*		RHC-LOCKED	
	(27 DEG)		*		PRO	
	(0,229/138,0)		*	06 18	(COMMANDED R,P,V)	
131	(106+38)	MOVE TO LEB	*		MONITOR MNVR	
	TPT TARGETING PROGRAM	(P34)	*	F 50 18	(COMMANDED R,P,V)	
	LOAD N55 WITH		*		MN BUS TIE(2)-ON(UP)	
	R2+20830		*		SPS HE VLV TB(BOTH)-BP	
	R3+13000		*		SPS HE VLV(BOTH)-AUTO	
129	(106+40)	CALL MARKING ROUTINE(V87,V57)	*		NONESS BUS-MNA	
	(V93 AFTER 3 MARKS PROCESSED)		*		PRO	
	SUNSET	*	*	06 18	(COMMANDED R,P,V)	
12	(106+57)	TERMINATE MARKS(17)	*		MONITOR ATT TRIM	
	MOVE TO CMD SEAT	*	*	F 50 18	(COMMANDED R,P,V)	
	TPT TARGETING FINAL COMP	(P34)	*	*****	*****	
18	(107+01)	VERIFY ORDEAL(V83)	*	-4	(107+05)	
					SPS THRUST SETUP	(P40)

					GUC ALIGN	

MISSION F RESCUE TIMELINE APRIL 25, 1969

GET	EVENT	PROG	GET	EVENT	PROG
		-166-			
	ALT SET KNOB TO 60 NM	*		06 18 (COMMANDED R,P,Ŷ)	*
	FDAI SELECT-1	*		MONITOR ATT TRIM	*
	ATT SET THUMBWHEELS TO N18	*		F 50 18 (COMMANDED R,P,Ŷ)	*
	NULL ATT ERROR NEEDLES	*		KEY ENTER	*
	ON FDAI 1 WITH ATT	*		F 50 25 (00204,GMAL DRIVE TEST)	*
	SET THUMBWHEELS	*		RHC PWR DIRECT(BOTH)-MNA/MNB	*
	ATT SET-GUC	*		RATE-HIGH	*
	DEPRESS GUC ALIGN PB	*		AUTO RCS SEL A/ꝝ ROLL(4)-MNA	*
	FDAI SELECT-1/2	*		PRO	*
	ATT SET-TMU	*		MONITOR GMAL DRIVE	*
	ATT SET-TMU	*		SEQ AND TRIM	*
	ATT SET-TMU	*		06 40 (TFI,VG,DVM)	*
	MTVC CHECKS	*		FDAI SCALE-5/5	*
	BMAG MODE(3)-ATT1/RATE?	*		VERIFY SPS TH LT-OFF	*
	MTVC SERVO PWR 1-AC1/MNA	*		EMS MODE-SBY	*
	MTVC SERVO PWR 2-AC2/MNA	*		EMS FUNCTION-DV SET	*
	TRANS CONTR PWR-ON	*		LOAD TH MTVC	*
	RHC PWR NORM 2-AC	*		EMS FUNCTION-DV	*
	GMAL MTRS PITCH 1 STRT-ON	*		THC-ARMED	*
	GMAL MTRS YAW 1 STRT-ON	*		RHC-ARMED	*
	THC-CLOCKWISE	*		RY THRUST(A)-NORMAL	*
	RHC-ARMED	*		OSKY BLANKS	*
	RHC-VERIFY NO MTVC	*		06 40 (TFI,VG,DVM)	*
	GMAL MTRS PITCH 2-STRT-ON	*		EMS MODE-NORMAL	*
	GMAL MTRS YAW 2-STRT-ON	*		PERFORM ULLAGE	*
	SET SP5 GIMBALS TW(2)-	*		F 99 40 (REQUEST FOR ENGINE ENABLE)	*
	PITCH = -.52	*		PRO	*
	YAW = +.59	*		06 40 (TFI,VG,DVM)	*
	RHC-VERIFY MTVC	*		*****	*
	THC-NEUTRAL	*			*
	RHC PWR NORM 2-AC/DC	*			*
	RHC-LOCKED	*			*
	PRO	*			*

GET	EVENT	PROG	GET	EVENT	PROG
(107+08+54)	*****	*	F 16 85	(VG-BODY)	*
CSM TPT BURN(-24,0,3,9)	*****	*	THC-NULL VG COMPONENTS		*
(0,189/10,0)	*****	*	THC-LOCKED		*
*****	*****	*	RHC-LOCKED		*
SET MDC ET-RESET-START	*****	*	EMS FUNCTION-VHF RNG		*
MONITOR	*****	*	EMS MODE-VHF RNG		*
SPS THRUST LT-ON	*****	*	VHF RNG-RESET		*
DV INDICATOR-DECREASING	*****	*	MN BUS TIE(2)-OFF		*
+00+01	ULLAGE-OFF	*	PRO		*
06 40 (TFC, VG, DVM)	*****	*	F 37 BR		*
MONITOR SPS OPERATION	*****	*	*****		*
PC INDICATOR-95-105 PSI	*****	*	RENDEZVOUS NAV PROGRAM	(P20)	
MONITOR SPS ENGINE CUTOFF	*****	*	AUTO MNVR TO SXT TRACK		
SPS THRUST LT-OFF	*****	*	(37 DEG)		
F 16 40 (TFC, VG, DVM)	*****	*	(0,247/47,0)		
DV THRUST(BOTH)-OFF	*****	*	MOVE TO LEB		
MONITOR	*****	*	MCC TARGETING PROGRAM	(P35)	
PC INDICATOR = 0	*****	*	CALL MARKING ROUTINE(V93, V87, V57)		
SPS INJ VLV IND(4)-CLOSE	*****	*	(107+18)		
SPS HE VLV TB(BOTH)-HP	*****	*	LOS		
SPS GMBLS RETURN TO SERVO	*****	*	TERMINATE MARKS(8)		
NULL	*****	*	MCC TARGETING FINAL COMP	(P35)	
GMRL MTRS-OFF(SEQUENTIALLY)	*****	*	RCS THRUST PROGRAM	(P41)	
TVC SERVO PWR(BOTH)-OFF	*****	*	BYPASS BURN ATTITUDE MNVR		
FDAI SCALE-5/1	*****	*	(0,268/30,0)		
RATE-LOW	*****	*	RCS THRUST SETUP	(P41)	
ROT CONT PWR DIRECT(BOTH)-OFF	*****	*			
PRO	*****	*			

GET	EVENT	PROG	GET	EVENT	PROG
+15	(107+23+54)	-168-			
	***** CSM MCC1 BURN (0,264/30,0)			***** * BRAKING GATES SETTLE ANGLE * * 30 FPS AT 6000 FT .13 DEG. * * 20 FPS AT 3000 FT .26 DEG. * * 10 FPS AT 2500 FT .54 DEG. * * 5 FPS AT 500 FT 1.60 DEG. *	
+15				*****	
+15	MCC TARGETTING PROGRAM (P35)				
+17	CALL MARKING ROUTINE(V93,V87,V57)		+43	(107+51+54)	
+26	(107+31) SUNRISE			***** TPF	
+26	TERMINATE MARKS(9)			*****	
+27	MCC TARGETTING FINAL COMP (P35)				
+28.5	RCS THRUST PROGRAM (P41) BYPASS BURN AUTO MNVR (0,287/17,0)				
+30	RCS THRUST SET UP (P41)				
+30	(107+38+54)				
	***** CSM MCC2 BURN (0,293/17,0)				
+31	*****				
+31	AUTO MNVR TO COAS TRACK(V89) (P00) (28 DEG) (0,262/341,0)				
+31	MOVE TO CMD SEAT				
+37	(107+46)				
	THRUST MONITORING PROGRAM (P47) V83, PERFORM BRAKING AND LOS CONTROL				

9.6.2 RENDEZVOUS NAVIGATION SUMMARY WITH SUN ANGLES

CSM ACTIVE LM RESCUE-PARTIAL INSERTION=120 FPS

(MISSION F)

IT - INITIATE TRACK

CT - CEASE TRACK

(OUT-OF-PLANE SUN ANGLE IS 4 DEG)

GET	EVENT	SUN ANGLE (LOS TO SUN) DEG
102:43	LM PARTIAL INSERTION	
102:47	SUNSET	
103:32	CSM RESCUE	
103:33	SUNRISE	
103:55	IT (SXT) (V93,20/20)	129
104:00		116
104:05		102
104:10		87
104:15	CT (SXT)	74
104:30	CSM CSI ONE	
104:42	SUNSET	
105:00	IT (SXT/VHF) (V93,6/5,V93,17/18)	
105:05	CT (SXT)	
105:23	CT (VHF)	
105:29	SUNRISE	
105:32	CSM CSI TWO	
105:41	IT (SXT/VHF) (4/3,V67,00057,00034,00001;6/7)	171
105:46		174
105:51	CT (SXT/VHF)	159
106:03	CSM PLANE CHANGE	
106:09	IT (SXT/VHF)(4/3,V93,11/12)	99
106:14		84
106:19		68
106:24	CT (SXT/VHF)	59
106:34	CSM CDH	
106:40	IT (SXT/VHF) (4/3,V93,13/14)	9
106:44	SUNSET	4
106:57	CT (SXT/VHF)	
107:09	CSM TPI	
107:12	IT (SXT/VHF) (V93,8/8)	
107:20	CT (SXT/VHF)	
107:24	CSM MCC1	
107:26	IT (SXT/VHF) (V93,9/9)	
107:31	SUNRISE	63
107:35	(CT (SXT/VHF)	77
107:39	CSM MCC2	
107:52	TPF	

9.6.3 CSM ATTITUDE HISTORY

**
INERTIAL AND ORDEAL FDAI BALL GIMBAL ANGLE PROFILE FOR THE LM
PARTIAL INSERTION = 120 FPS RESCUE CASE (MISSION F) (YAW = 0.0)
**

TIME GET	ROLL	PITCH ORDEAL	PITCH INERTIAL	PITCH MNVR
102:46	180	32	272	
103:25	180	165	282	19
103:32	180	180	282	
103:35	180	141	233	49
104:19	180	345	281	175
104:30	180	0	281	
104:52	180	162	11	90
105:27	180	165	276	
105:32	180	171	276	
105:37	180	164	249	27
106:00	180	155	178	
106:03	180	161	178	
106:04	180	154	165	13
106:22	0	223	178	
106:27	0	229	165	
106:34	0	244	165	
106:36	0	229	138	27
107:02	0	174	10	56
107:09	0	189	10	
107:11	0	247	47	37
107:22	0	258	30	
107:24	0	264	30	
107:37	0	287	17	
107:39	0	293	17	
107:40	0	262	341	28
107:52	0	298	341	

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